

**DEPARTMENT OF CIVIL ENGINEERING
GEETHANJALI COLLEGE OF ENGINEERING &
TECHNOLOGY**



CHEERYAL (V), KEESARA (M), R.R. DIST. - 501 301

(Affiliated to JNTUH, Approved by AICTE, NEW DELHI, ACCREDITED BY NBA)

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HYDRAULICS AND HYDRAULIC MACHINERY
COURSE FILE

(Subject Code: A40111)

II Year B.TECH. (CIVIL ENGINEERING) II Semester

Prepared by **MOHD. ABDUL KHADEER, K.RAVINDER Asst.Professor**



GEETHANJALI COLLEGE OF ENGINEERING & TECHNOLOGY

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DEPARTMENT OF CIVIL ENGINEERING

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1. Introduction to the subject

This course is intended to introduce basic principles of fluid mechanics. It is further extended to cover the application of fluid mechanics by the inclusion of fluid machinery especially water turbine and water pumps. Now days the principles of fluid mechanics find wide applications in many situations directly or indirectly. The use of fluid machinery, turbines pumps in general and in power stations in getting as accelerated fill up. Thus there is a great relevance for this course for mechanical technicians. The Mechanical technicians have to deal with large variety of fluids like water, air, steam, ammonia and even plastics. The major emphasis is given for the study of water. However the principle dealt with in this course will be applicable to all incompressible fluids.

Pre-requisites

1. Statics and dynamics
2. Mathematics of the motion of particles and rigid bodies and the relation of force and motion of particles
3. Fundamental concepts and laws of mechanics including equilibrium and Newton's laws of motion
4. Differential calculus
5. Basic system of units

2. Syllabus

Sl.No	Unit No	Topic
1	1	Introduction of open channel flow: type of channels, velocity distribution, Energy momentum correction factors-chezy's, manning's and bazin formulae for uniform flow- most economical sections.
		Critical flow, specific energy, critical depth, computation of critical depth, critical, sub-critical and super critical flows.
		Non uniform flow- Dynamic equation for G.V.F, mild, critical, steep, horizontal and adverse slopes.
		Surface profiles, direct step method, Rapidly varied flow, hydraulic jump, Energy dissipation.
2	2	Dimensional analysis and similitude: Dimensional analysis- Rayleigh's method and Buckingham's pi theorem, study of hydraulic models.
		Geometric, kinematic and dynamic similarities, dimensionless numbers- model and prototype relations.

3	3	Hydrodynamic forces on jets: hydrodynamic force of jets on stationary and moving flat , inclined and curved vanes, jet striking centrally and at tip, velocity triangles at inlet and outlet.
		Expression for work done and efficiency, angular momentum principle, applications to radial floe turbines.
		Layout of a typical hydropower installation, heads and efficiencies.
4	4	Hydraulic turbines: classification of turbines, Pelton wheel, Francis turbine, Kaplan turbine working, working proportions.
		Velocity diagram, work done and efficiency, hydraulic design, draft tube-theory and function efficiency.
		Governing of turbines, surge tank, unit and specific turbines, unit speed, unit quantity, unit power.
		Specific speed performance characteristics, geometric similarity, cavitation.
5	5	Centrifugal pumps: pump installation details, classifications, work done, manometric head, minimum starting speed, losses and efficiencies.
		Specific speed, multistage pumps, pumps in parallel, performance of pumps, characteristics curves, NPSH- cavitation.
		Classification of hydropower plants, definition of terms, load factor, utilization factor, capacity factor, estimation of hydropower potential.

Text books:

1. Fluid Mechanics, Hydraulic and hydraulic machines by **Modi** and **Seth**, Standard book house.
2. Open channel flow by **K.Subramanya** , Tata Mc.Grawhill publishers.
3. Fluid mechanics & fluid machines by Narayana pillai, universities press.

Reference Text Books:-

1. Fluid Mechanics & fluid machines by Rajput , S.Chand &co.
2. Fluid Mechanics and Machinery, CSP Ojha, Oxford Higher Education

3. Fluid Mechanics by Frank.M. White (Tata Mc.Grawhill Pvt. Ltd.)
4. Fluid Mechanics by A.K. Mohanty, Prentice Hall of India Pvt. Ltd., New Delhi
5. A text of Fluid mechanics and hydraulic machines by Dr. R.K. Bansal - Laxmi Pub.(P) ltd., New Delhi.
6. Fluid Mechanics and Machinery by D. Ramdurgaia New Age Publications.

Websites:-

1. <http://jntuhupdates.net/jntuh-b-tech-2-2-semester-r13-syllabus-book/>
2. NPTEL Resources
3. www.ieeefmhm.org/

Journals:-

1. International Journal of fluid mechanics
2. International Journal of numerical methods in fluids.

3. Vision of the Department:

To develop a world class program with excellence in teaching, learning and research that would lead to growth, innovation and recognition

4. Mission of the Department:

The mission of the Civil Engineering Program is to benefit the society at large by providing technical education to interested and capable students. These technocrats should be able to apply basic and contemporary science, engineering and research skills to identify problems in the industry and academia and be able to develop practical solutions to them

5. Program Educational Objectives-PEOs:

The Civil Engineering Department is dedicated to graduating Civil engineers who:

- A. Practice Civil engineering in the general stems of fluid systems, civil systems and design, and materials and manufacturing in industry and government settings.
- B. Apply their engineering knowledge, critical thinking and problem solving skills in professional engineering practice or in non-engineering fields, such as law, medicine or business.
- C. Continue their intellectual development, through, for example, graduate education or professional development courses.
- D. Pursue advanced education, research and development, and other creative efforts in science and technology.
- E. Conduct them in a responsible, professional and ethical manner.
- F. Participate as leaders in activities that support service to and economic development of the region, state and nation.

6. Program Outcomes (PO)

Graduates of the Civil Engineering Programme will be able to:

1. Apply the knowledge of mathematics, science, engineering fundamentals, and Civil Engineering principles to the solution of complex problems in Civil Engineering.
2. Identify, formulate, research literature, and analyse complex Civil Engineering problems reaching substantiated conclusions using first principles of mathematics and engineering sciences.
3. Design solutions for complex Civil Engineering problems and design system components or processes that meet the specified needs with appropriate consideration for the public health and safety, and the cultural, societal, and environmental considerations.
4. Use research-based knowledge and research methods including design of experiments, analysis and interpretation of data, and synthesis of the information to provide valid conclusions related to Civil Engineering problems.
5. Create, select, and apply appropriate techniques, resources, and modern engineering tools such as CAD, FEM and GIS including prediction and modelling to complex Civil Engineering activities with an understanding of the limitations.
6. Apply reasoning informed by the contextual knowledge to assess societal, health, safety, legal and cultural issues and the consequent responsibilities relevant to the professional Civil Engineering practice.
7. Understand the impact of the professional Civil Engineering solutions in societal and environmental contexts, and demonstrate the knowledge of, and need for sustainable development.
8. Apply ethical principles and commit to professional ethics and responsibilities and norms of the Civil Engineering practice.
9. Function effectively as an individual, and as a member or leader in diverse teams, and in multidisciplinary settings.
10. Communicate effectively on complex Civil Engineering activities with the engineering community and with society at large, such as, being able to comprehend and write effective reports and design documentation, make effective presentations, and give and receive clear instructions.
11. Demonstrate knowledge and understanding of the engineering and management principles and apply these to one's own work, as a member and leader in a team, to manage Civil Engineering projects and in multidisciplinary environments.
12. Recognise the need for, and have the preparation and ability to engage in independent and life-long learning in the broadest context of technological change.

7. Course objectives

Students who successfully complete this course will have demonstrated ability to:

1. Define the nature of a fluid.
2. Show where fluid mechanics concepts are common with those of solid mechanics and indicate some fundamental areas of difference.
3. Introduce viscosity and show what are Newtonian and non-Newtonian fluids
4. Define the appropriate physical properties and show how these allow differentiation between solids and fluids as well as between liquids and gases
5. The purpose of this course is to learn the Fluid properties and fundamentals of Fluid statics and fluid flow
6. To introduce the concepts of flow measurements and flow through pipes
7. To introduce the concepts of momentum principles
8. To impart the knowledge on pumps and turbines
9. To impart the knowledge of impact of jets.

10. To introduce the flow measuring devices and velocity measuring devices.

8. Course Outcomes

1. Knowledge of basic principles of fluid mechanics
2. Know the definitions of fundamental concepts of fluid mechanics including: continuum, velocity field; viscosity, surface tension and pressure (absolute and gage); flow visualization using timelines, pathlines, streaklines, and streamlines; flow regimes: laminar, turbulent and transitional flows; compressibility and incompressibility; viscous and inviscid.
3. Apply the basic equation of fluid statics to determine forces on planar and curved surfaces that are submerged in a static fluid; to manometers; to the determination of buoyancy and stability; and to fluids in rigid-body motion.
4. Ability to analyze fluid flow problems with the application of the momentum and energy equations
5. Use of conservation laws in differential forms and apply them to determine velocities, pressures and acceleration in a moving fluid. Understand the kinematics of fluid particles.
6. Use Euler's and Bernoulli's equations and the conservation of mass to determine velocities, pressures, and accelerations for incompressible and inviscid fluids.
7. Understand the concepts of rotational vs. irrotational flows; stream functions, velocity potentials.
8. Understand the physical processes which govern the behavior of fluids at rest and in motion
9. Confidently pose and solve problems in engineering fluid mechanics

9. Instructional learning

A mixture of lectures, tutorial exercises, and case studies are used to deliver the various topics. Some of these topics are covered in a problem-based format to enhance learning objectives. Others will be covered through directed study in order to enhance the students' ability of "learning to learn." Some case studies are used to integrate these topics and thereby demonstrate to students how the various techniques are inter-related and how they can be applied to real problems in an industry.

10. Course mapping with PEO's and PO's

PEO/PO	Program Outcomes													
		1	2	3	4	5	6	7	8	9	10	11	12	
Program Educational Objectives (PEO)	A	X	X	X	X	X	X	X					X	
	B		X	X	X	X	X	X	X	X	X			
	C			X	X	X	X				X	X	X	
	D	X	X	X	X	X				X	X			
	E			X	X	X	X		X				X	

	F											X	X	

11. Class Timetable

DEPARTMENT OF CIVIL ENGINEERING

Ref: TLE/2014-2015/23.12.2014/SADM /CT -1004

PROGRAMME: B.TECH. (CIVIL ENGINEERING)

SEMESTER: II Year II- SEMESTER

NOTE: “*” Represents Tutorial Classes.

Time Table Coordinator

HOD

PRINCIPAL

Time	9.30-10.20	10.20-11.10	11.10-12.00	12.00-12.50	12.50-1.30	1.30-2.20	2.20-3.10	3.10-4.00
Period	1	2	3	4	LUNCH	5	6	7
Monday	SOM	HHM		BMC		P&S	S.A	
Tuesday	EVS	SOM		S.A		BMC	P&S	HHM
Wednesday	P&S	BMC	SA	HHM		SOM		LIBRARY
Thursday	EVS	SA	HHM	P&S		BMC		EVS
Friday	LAB			P&S		CRT		EVS
Saturday	EVS	LAB				MENTOR	SEMINAR	

12. Individual Time Table

Name of the faculty: Load = 10 Rev: w.e.f.:

Section- II A and II B

Name of the faculty: MOHD. ABDUL KHADEER Load = 16 ; w.e.f.: 29/06/15

Time	9.30-10.20	10.20-11.10	11.10-12.00	12.00-12.50	12.50-1.30	1.30-2.20	2.20-3.10	3.10-4.00
Period	1	2	3	4	LUNCH	5	6	7
Monday		HHM						
Tuesday								HHM
Wednesday				HHM				
Thursday			HHM					
Friday								
Saturday								

13. Unit wise Summary

Unit No	Total Periods	Topic	Reg/Additional	LCD/OH P/BB	Remark
1	9	Introduction of open channel flow: Type of channels, velocity distribution, Energy momentum correction factors-chezy's, manning's and bazin formulae for uniform flow-most economical sections.	Regular	BB	
		Critical flow, specific energy, critical depth, computation of critical depth, critical, sub-critical and super critical flows.	Regular	BB	
	6	Non uniform flow- Dynamic equation for G.V.F, mild, critical, steep, horizontal and adverse slopes.	Regular	BB	
		Surface profiles, direct step method, Rapidly varied flow, hydraulic jump, Energy dissipation.	Regular	BB	
				54	
2	6	Dimensional analysis and similitude: Dimensional analysis- Rayleigh's method and Buckingham's pi theorem,	Regular	BB	
		study of hydraulic models	Regular	BB	
	7	Geometric, kinematic and dynamic similarities, dimensionless numbers	Regular	BB	
		Model and prototype relations.	Regular	BB	
3	6	Hydrodynamic forces on jets: hydrodynamic force of jets on stationary and moving flat , inclined and curved vanes	Regular	BB	
		Jet striking centrally and at tip, velocity triangles at inlet and outlet.	Regular	BB	
		Expression for work done and efficiency, angular momentum principle	Regular	BB	

	6	Applications to radial flow turbines.	Regular	BB	
		Layout of a typical hydropower installation,	Regular	BB	
		Heads and efficiencies.	Regular	BB	
4	5	Hydraulic turbines: classification of turbines, Pelton wheel, Francis turbine, Kaplan turbine working, working proportions.	Regular	BB	
		Velocity diagram, work done and efficiency, hydraulic design, draft tube-theory and function efficiency.	Regular	BB	
	7	Governing of turbines, surge tank, unit and specific turbines, unit speed, unit quantity, unit power.	Regular	BB	
		Specific speed performance characteristics, geometric similarity, cavitation.	Regular	BB/OHP	
5	4	Centrifugal pumps: pump installation details, classifications, work done, manometric head, minimum starting speed, losses and efficiencies.	Regular	BB/OHP	
		Specific speed, multistage pumps, pumps in parallel, performance of pumps,	Regular	BB	
	5	characteristics curves, NPSH- cavitation.	Regular	BB	
		Classification of hydropower plants	Regular	BB	
	3	Definition of terms, load factor, utilization factor, capacity factor, estimation of hydropower potential.	Regular	BB	

13. Micro Plan with dates and closure report

Sl. No	Unit No.	Date (No. of Periods)	Topic to be covered in One Lecture	Reg/ Additional	Teaching aids used LCD/OHP /BB	Remarks
1	I	-12-2015 01	Introduction of open channel flow: Type of channels	Regular	OHP, BB	
2		-12-2015 01	velocity distribution, Energy	Regular	OHP, BB	

			momentum correction			
3		-12-2015 01	factors-chezy's, manning's and bazin formulae for uniform flow	Regular	OHP,BB	
4		-12-2015 -12-2015 02	most economical sections.	Regular	OHP,BB	
5		-12-2015 01	Critical flow, specific energy, critical depth, computation of critical depth	Regular	BB	
6		-12-2015 01	critical, sub-critical and super critical flows.	Regular	BB	
7		01	Non uniform flow- Dynamic equation for G.V.F	Regular	OHP,BB	
8		01	mild, critical, steep, horizontal and adverse slopes.	Regular	BB	
9		01	Surface profiles, direct step method	Regular	BB	
10		01	Rapidly varied flow, and related problems	Regular	BB	
11		02	hydraulic jump, Energy dissipation	Regular	BB	
12		01	Test	Regular	BB	
13	II	01	Introduction of Dimensional analysis	Regular	OHP,BB	
14		01	Rayleigh's method	Regular	BB	
15		01	Buckingham's pi theorem	Regular	BB	
16		01	study of hydraulic models	Regular	OHP,BB	
17		02	Geometric similarities	Regular	BB	
18		02	Kinematic similarities	Regular	BB	
19		01	dynamic similarities,	Regular	BB	
20		02	dimensionless numbers	Regular	OHP,BB	
21		01	Model and prototype relations.	Regular	OHP,BB	

22		01	Test	Regular	BB	
23	III	01	Introduction of hydrodynamic force of jets	Regular	BB	
24		02	Jets on stationary and moving flat , inclined and curved vanes	Regular	OHP,BB	
25		01	Jet striking centrally and at tip,	Regular	OHP,BB	
26		01	Velocity triangles at inlet and outlet.	Regular	OHP,BB	
27		01	Expression for work done and efficiency,	Regular	BB	
28		02	angular momentum principle	Regular	LCD,OH P,BB	
29		01	Applications to radial floe turbines.	Regular	BB	
30		01	Layout of a typical hydropower installation,	Regular	OHP,BB	
31		01	Heads and efficiencies.	Regular	BB	
32		01	Test	Regular	BB	
	IV	01	Classification of turbines, Pelton wheel,	Regular	OHP,BB	
33		01	Francis turbine and working proportions.	Regular	OHP,BB	
34		02	Kaplan turbine working, working proportions	Regular	BB	
35		01	Velocity diagram, work done and efficiency, hydraulic design,	Regular	OHP,BB	
36		01	Draft tube-theory and function efficiency.	Regular	OHP,BB	
37		02	Governing of turbines, surge tank, unit and specific turbines,	Regular	BB	
38		01	Specific speed of turbines	Regular	OHP,BB	
39		01	Specific speed performance characteristics,	Regular	BB	

40		01	Geometric similarity, cavitation.	Regular	BB	
41		01	Test	Regular	OHP,BB	
	V	01	Introduction of Centrifugal pumps, pump installation details, and classifications,	Regular	OHP,BB	
42		01	Work done, Manometric head,	Regular	OHP,BB	
43		01	Minimum starting speed, losses and efficiencies.	Regular	BB	
44		01	Specific speed, multistage pumps,	Regular	OHP,BB	
45		01	pumps in parallel, performance of pumps,	Regular	OHP,BB	
46		01	Characteristics curves.	Regular	OHP,BB	
47		02	NPSH- cavitation.	Regular	OHP,BB	
48		01	Classification of hydropower plants.	Regular	OHP,BB	
49		01	Definition of terms, load factor, utilization factor, capacity factor,	Regular	OHP,BB	
50		01	Estimation of hydropower potential.	Regular	OHP,BB	
51		01	Discussion of previous question papers	Regular	BB	

GUIDELINES:

Distribution of periods:

No. of classes required to cover JNTUH syllabus	: 64
No. of classes required to cover Additional topics	: Nil
No. of classes required to cover Assignment tests (for every 2 units 1 test)	: 4
No. of classes required to cover tutorials	: 2
No. of classes required to cover Mid tests	: 2
No of classes required to solve University Question papers	: 2

Total periods

64

14. Detailed Notes

Fluid Mechanics

Units and dimensions:

Angular Velocity..... rad/g

Angular Acceleration..... rad/g^2

Discharge m^3/g

Sp.mass(mass density)..... kg/m^3

Stress , elastic modulus..... N/m^2

Sp.weight(weight density)..... N.S/m^2

Dynamic viscosity

Kinematic viscosity m^2/g

Work, Energy, Torque.....J(N-m)
 Powerwatt(J/g)
 Surface Tension.....N/m
 Momentum, moment of momentum.....kg.m/g
 EntropyJ/kg.k
 Sp.heat, gas constant
 Thermal conductivity.....W/m.k
 Dynamic viscosity.....poise(p)=1/10=N.S/m²
 Kinematic viscosity –stoke(s)= 10⁻⁴ m²/s
 Pressure of fluid.....10⁵ pa

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$$\begin{aligned} \rightarrow \sin^2 \theta + \cos^2 \theta &= 1 \\ \rightarrow 1 + \cot^2 \theta &= \operatorname{cosec}^2 \theta \end{aligned}$$

$$\rightarrow \tan(A+B) = \frac{\tan A + \tan B}{1 - \tan A \tan B}$$

$\frac{\cos \theta}{\tan \theta}$	$\frac{\cos \theta}{\tan \theta}$
$\frac{\sin \theta}{\cos \theta}$	$\frac{\sin \theta}{\cos \theta}$
$\tan \theta$	$\tan \theta$

$$\rightarrow \sin C + \sin D = 2 \sin \frac{C+D}{2} \cos \frac{C-D}{2}$$

$$\rightarrow \sin C - \sin D = 2 \cos \frac{C+D}{2} \sin \frac{C-D}{2}$$

$$\rightarrow \cos C + \cos D = 2 \cos \frac{C+D}{2} \cos \frac{C-D}{2}$$

$$\rightarrow \cos C - \cos D = 2 \sin \frac{C+D}{2} \sin \frac{D-C}{2}$$

$$2 \sin A \sin B = \cos(A-B) - \cos(A+B)$$

$$2 \cos A \cos B = \cos(A+B) + \cos(A-B)$$

$$2 \sin A \cos B = \sin(A+B) + \sin(A-B)$$

$$2 \cos A \sin B = \sin(A+B) - \sin(A-B)$$

$$\sin 2A = 2 \sin A \cos A$$

$$\cos 2A = \cos^2 A - \sin^2 A$$

$$= 2 \cos^2 A - 1$$

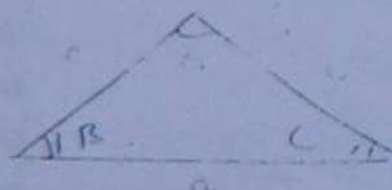
$$= 1 - 2 \sin^2 A$$

$$\tan 2A = \frac{2 \tan A}{1 - \tan^2 A}$$

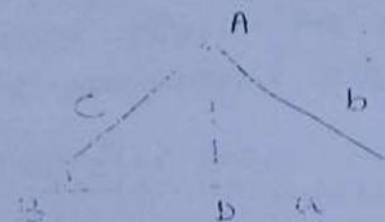
$$\sin 3A = 3 \sin A - 4 \sin^3 A$$

$$\cos 3A = 4 \cos^3 A - 3 \cos A$$

$$\tan 3A = \frac{3 \tan A - \tan^3 A}{1 - 3 \tan^2 A}$$



$$\frac{\sin A}{a} = \frac{\sin B}{b} = \frac{\sin C}{c}$$



$$c^2 = a^2 + b^2 - 2ab \cos C$$

$$a^2 + b^2 - c^2 = 2ab \cos C$$

The combination of these two laws is given by

$$\boxed{\begin{aligned} L_r^2 &= D_r^2 \\ \text{OR} \\ L_r^{2/3} &= H_0/P_0 \end{aligned}}$$

Above law satisfies both laws of viscosity Reynolds & Froude

27 In case of the rivers normally scale adopted for depth is different from scale for width & length therefore slope of bed is exaggerated and according to Manning's Law

$$V_0 = \frac{R_0^{2/3} S_0^{1/2}}{n_0}$$

where $S_0 = \frac{D_0}{L_0} \rightarrow$ Depth Ratio or Height Ratio
 \downarrow
 Length Ratio / width Ratio

For wide Rivers $A_0 = L_0 \times D_0$

(Perimeter) $P_0 \approx L_0$

$R_0 = \frac{A_0}{P_0} = D_0$

$$V_0 = \frac{D_0^{2/3}}{n_0} \left(\frac{D_0^{1/2}}{L_0^{1/2}} \right)$$

$$\boxed{V_0 = \frac{D_0^{7/6}}{n_0 L_0^{1/2}}} \quad \rightarrow$$

→ In case of the rivers in open channel, Froude Model Law is applicable.

$$V_0 = D_r^{1/2} \cdot g_0^{1/2} \quad \rightarrow$$

By eqn (1) & (11)

$$\boxed{n_0 = D_0^{4/6} L_0^{-1/2}}$$

[put $g_0 = 1$]

$$\boxed{n_0 = D_0^{2/3} L_0^{-1/2}}$$

→ For distorted model.

(LHSE)

37 Euler Number: \downarrow

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When the pressure force is very important apart from inertia force as compared to other forces than Euler No. is defined as

$$Eu = \sqrt{F_i / F_p}$$

$$= \sqrt{\frac{\rho L^2 v^2}{p \times L^2}} = \frac{v}{\sqrt{p/\rho}}$$

\Rightarrow \sqrt{Eu} is called Newton Number:

\Rightarrow Newton number square is called pressure coefficient.

$$\sqrt{Eu}^2 = \text{Newton Number} = \frac{F_p}{F_i} = \frac{p}{\rho v^2}$$

Euler Model Law: \downarrow

ex:

- a) Flow through pipes under pressure
- b) Flow over submerged bodies when pressure is important.
- c) Pressure rise due to sudden closure of valve.
- d) Discharge through weir & mouthpieces, under large head

According to this Law

$$(Eu)_m = (Eu)_p$$

NOTE

- 1) In some of the cases where viscosity force & gravity force both are important than Reynolds Law & Froude Law both should be applicable.

For ex: \downarrow

- a) Resistance to ship is generally caused by viscosity & eddies formed by wave hence both law should be satisfied.

Mach Law: - mainly in compressible fluid flowing with sound.

- ex:
- 1) flow of gases having high speed.
 - 2) water hammer problem
 - 3) aerodynamic testing
 - 4) Testing of torpedo's
 - 5) Motion of airplane with high speed
 - 6) Launching & Projectile of missile.

(345)

According to this Law

$$(Ma)_m = (Ma)_p$$

$$V_a/c_a = 1$$

4) Weber Model: ↓

When surface tension force apart from inertia force is important but other forces are less significant than Weber No. is defined as

$$We = \frac{F_i}{F_s}$$

$$We = \frac{\rho L^2 V^2}{\sigma L}$$

$$We = \frac{V}{\sqrt{\sigma/\rho L}}$$

Weber Model Law: ↓

Application

- a) capillary movement of water in soil
- b) flow of bloods in veins & arteries
- c) thin sheet flow
- d) Liquid atomization
- e) capillary tube flow.

According to this Law

$$(We)_p = (We)_m$$

$$\sqrt{\frac{\sigma}{\rho L}} = 1$$

$$= \rho_2 L_r^3 g_2 \times L_r^{3/2} g_2^{1/2}$$

$$= \rho_2 L_r^{7/2} g_2^{3/2}$$

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When compressibility force is significant apart from inertia force then mach number is important and it is defined as

$$M_a = \sqrt{F_i / F_c}$$

$$\therefore M_a = \sqrt{\frac{\rho L^2 v^2}{K \times L^2}} = \frac{v}{\sqrt{K/\rho}}$$

$$M_a = v/c$$

$\sqrt{K/\rho} \rightarrow$ velocity of sound (c)

Mach number is significant when velocity is comparable with sound velocity

- at
- (i) $M_a > 1 \rightarrow$ Flow is supersonic
 - (ii) $M_a = 1 \rightarrow$ " " sonic
 - (iii) $M_a >> 1 \rightarrow$ Flow is hypersonic
 - (iv) $M_a < 1 \rightarrow$ Flow is subsonic/ultra
 - (v) $M_a < 0.3 \Rightarrow$ Effect of compressibility is neglected.

$$M_a < 0.4$$

NOTE:

Square of mach number is called cauchy's

$$M_a^2 = \text{cauchy's No.}$$

$$\frac{v_a^2}{c^2} = "$$

$$= \frac{F_i}{F_c}$$

Flow over spillway of a dam

- 3) Flow of Liquid jet of orifice
- 4) Flow over weir & Notches.
- 5) Motion of ship in rough and turbulent

According to this Law

$$(Fr)_p = (Fr)_m$$

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$$\therefore \left(\frac{V}{\sqrt{gL}} \right)_p = \left(\frac{V}{\sqrt{gL}} \right)_m$$

$$\Rightarrow \boxed{\frac{V_n}{g_n^{1/2} L_n^{1/2}} = 1}$$

However $g_n = 1$

a) Time Ratio $T_n = ?$

$$L_n / T_n = V_n = g_n^{1/2} L_n^{1/2}$$

$$\therefore T_n = \frac{L_n^{1/2}}{g_n^{1/2}}$$

$$\therefore \boxed{T_n = \frac{L_n^{1/2}}{g_n^{1/2}}}$$

$$\boxed{T_r = L_r^{1/2}} \quad \text{at } g_n = 1$$

b) Acc'n Ratio:

$$a_n = g_n = 1$$

c) Force Ratio:

$$\begin{aligned} F_n &= \rho_n L_n^3 \times g_n \\ &= \rho_n L_n^3 \end{aligned}$$

Discharge Ratio

Discharge Ratio:

$$Q_n = \frac{L_n^3}{T_n} = \frac{L_n^3}{L_n^{1/2} g_n^{1/2}}$$

$$\boxed{Q_n = \frac{L_n^{5/2}}{g_n^{1/2}}}$$

$$\frac{L_n^2 \times \rho_n}{\rho_n} = \frac{L_n^2 \times \rho_n}{\rho_n}$$

> Force Ratio:

$$F_r = M_r \cdot q_r$$

$$= \rho_n \times L_n^3 \times \frac{u_n^2}{\rho_n^2 \times L_r^3}$$

$$F_r = \frac{u_n^2}{\rho_n}$$

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> Power Ratio:

$$P_r = F_r \times V_r$$

$$P_r = \frac{u_n^2}{\rho_n} \times \frac{u_n}{L_n \cdot \rho_n}$$

$$P_r = \frac{u_n^3}{L_n \rho_n^2}$$

> Froude Number:

When gravity force is important apart from inertia force but other forces are less significant than Froude Number is defined as

$$F_r = \sqrt{F_i / F_g} = \sqrt{\frac{\rho L^3 v^2}{\rho L^3 g}}$$

$$F_r = \frac{v}{\sqrt{L g}}$$

L: Length

Parameter

: A/D [For open channel]

Froude Model Law:

When gravity force is very important than this Law is applicable.

Ex: > Flow through open channels (waves & jumps)

Reynolds Model Law:

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In the flow conditions where viscosity forces are very predominant than other, then this Law is applied.

Example

- Flow in pipes under laminar conditions
- Flow of submarine & air plane but submarine fully emerged.
- Flow around submerged structure.
- Flow through low speed turbo machines.

→ For such conditions Re for Model be equal to Re for prototype.

$$\left(\frac{\rho V L}{\mu}\right)_m = \left(\frac{\rho V L}{\mu}\right)_p$$

$$\Rightarrow \boxed{\frac{\rho_2 V_2 L_2}{\mu_2} = 1} \rightarrow \text{Reynolds Law}$$

$$\mu/\rho = \nu,$$

$$\boxed{\frac{V_2 L_2}{\nu_2} = 1}$$

EX: 1)

$$T_2 = ?$$

$$\rho_2 \cdot \frac{L_2}{T_2} \times \frac{L_2}{\mu_2} = 1$$

$$\Rightarrow \boxed{T_2 = L_2^2 \frac{\rho_2}{\mu_2}}$$

2) Acc'n Ratio

$$a_2 = \frac{V_2}{T_2} = L_2 / T_2^2$$

$$\therefore a_2 = \frac{L_2 \mu_2^2}{\frac{1}{\rho_2} \cdot \rho_2^2}$$

$$\therefore \boxed{a_2 = \frac{\mu_2^2}{L_2 \cdot \rho_2}}$$

a) Gravity Force = $F_g = m \cdot g$

$$= \rho L^3 \cdot (L/T^2 \rightarrow \text{same for Model \& prototype})$$

$$= \rho L^3 \cdot g$$

b) Inertial Force \rightarrow Effect of mass and it always acts. \Rightarrow

$$F_i = M \times a$$

$$= \rho L^3 \times L/T^2$$

$$= \rho L^2 (L^2/T^2) = \rho L^2 v^2$$

c) viscosity Force:

$$F_v = \mu \times A$$

$$= \mu (V/L) \times L^2 = \mu V L$$

d) Pressure Force:

$$= p \times A$$

$$= p \times L^2$$

e) surface Tension Force:

$$F_s = \sigma \cdot L$$

f) Compressibility Force:

\rightarrow Bulk modulus

$$F_c = K \times A = K \times L^2$$

Reynolds Number: \downarrow

For situation, viscosity forces are very predominant with inertial forces but other forces are less significant than Reynolds number is defined as:

$$Re = \frac{F_i}{F_v}$$

$$= \frac{\rho L^2 v^2}{\mu V L}$$

Eff.

$L \rightarrow$ Length parameter

$\rho \rightarrow$ mass density

$$V_1 + c = \frac{2y_2}{2y_1} (y_1 + y_2)$$

$$3 + \frac{2}{y_2 - 2} = \frac{9.81 y_2}{2 \times 2} (2 + y_2)$$

$$\Rightarrow y_2 = ?$$

→ clearly: - The velocity of the surge relative to the initial flow velocity in a canal is called celerity.

$$C_b = V_2 - V_1 \quad \text{--- D/S}$$

$$C_s = V_2 + V_1 \quad \text{--- U/S}$$

$$C_s = \sqrt{\frac{g}{2} \frac{y_2}{y_1} [y_1 + y_2]}$$

MODELS

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Similarities: ↓

1) Geometric similarities: - It includes physical parameters such as Length, width, height, volume, area, etc.

eg

$$\text{Area ratio} = \frac{\text{Area of prototype}}{\text{Area of Model}}$$

$$A_r = \frac{L_r^2}{L_r} = \left(\frac{L_p}{L_m}\right)^2 = \left(\frac{\text{Length of prototype}}{\text{Length of Model}}\right)^2$$

$$V_r = \frac{L_r^3}{L_r}$$

$$[L_r = 100:1 \text{ or } 1:100]$$

2) Kinematic similarity: Those parameters which are effect of time such as velocity, acceleration, discharge etc.

$$a) V_r = \frac{V_p}{V_m} = \frac{L_r}{T_r}$$

$$b) \text{Accn. ratio} = a_r = \frac{V_r}{T_r} = \frac{L_r}{T_r^2}$$

$$c) \text{discharge ratio} = Q_r = \frac{L_r^3}{T_r}$$

3) Dynamic similarity: - It exists if the ratio of the forces at homologous model & prototype are similar.

In an open channel, if some external force or small surge is created and at ^{this} surge ~~this~~ has zero celerity then $V_1 \equiv \sqrt{gy} \Rightarrow V/\sqrt{gy} = Fr = 1$. Similarly if celerity is +ve means it causes upwards ^{flow} than $V_1 < \sqrt{gy}$. Therefore $Fr < 1$ Hence Flow is subcritical.

Similarly if celerity -ve wave will travel d/s.

Ex:1

A trapezoidal channel with base width of 6m & side slope with AH:V conveys water at the rate of 17 m³/sec at a depth of flow of 1.5m. At this flow situation, is subcritical or supercritical.

$$Fr = \frac{V}{\sqrt{gD}}, D = A/T = \frac{(b+ny)y}{b+2ny} \quad [n=2]$$

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Ex:2

A rectangular horizontal channel of 3m width and 2m depth conveys water at 18 m³/sec. If the flow rate is suddenly reduced to 2/3 of its original value. Compute the magnitude ^{of velocity} and speed of U/s surge wave. Assume that there is no friction in the channel.

$$V_1 = Q/A_1 = 18/(b \cdot y_1) = \frac{18}{3 \times 2} = 3 \text{ m/sec}$$

$$V_2 = Q_2/A_2 = \frac{2/3 \times 18}{3 \times y_2} = 4/y_2$$

At stable condition

$$b y_1 (V_1 + C) = b y_2 (V_2 + C) \quad \text{--- (1)}$$

$$P_1 - P_2 = \rho g [V_2 y_2 - V_1 y_1] \quad \text{--- (2)}$$

$$C = \frac{V_1 y_1 - V_2 y_2}{y_2 - y_1} \quad \text{--- (3)}$$

$$= \frac{3 \times 2 - 4/y_2 \times y_2}{y_2 - 2}$$

$$C = \frac{2}{y_2 - 2}$$

For prismatic Rectangular channel:

$$Q = by_1(v_1 + c) = by_2(v_2 + c) \Rightarrow y_1(v_1 + c) = y_2(v_2 + c)$$

$$\therefore c = \frac{v_1 y_1 - v_2 y_2}{y_2 - y_1} \quad \text{--- (1)}$$

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If there is no resistance force b/w (1) & (2) then
Total Force = 0.

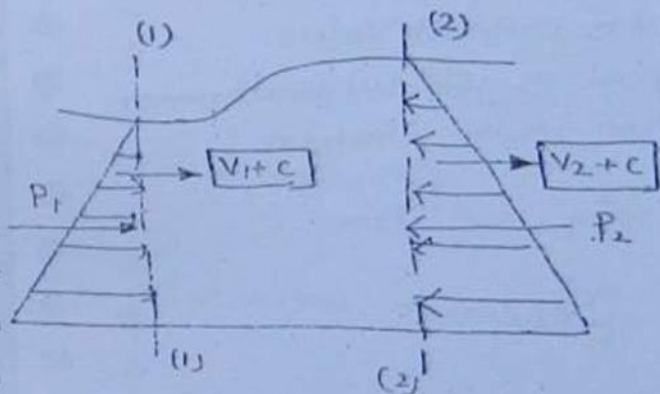
= static force + dynamic force

$$= (P_1 - P_2) + \rho A \{ [v_1 + c] - [v_2 + c] \} =$$

$$\Rightarrow P_1 - P_2 = \rho A [v_2 - v_1]$$

$$\therefore \omega b \cdot y_1 \times y_1/2 - \omega b \cdot y_2 \times y_2/2 = \rho A [v_2 - v_1]$$

$$\therefore \frac{\rho g b}{2} [y_1^2 - y_2^2] = \rho A [v_2 - v_1] \quad \text{--- (2)}$$



From eqn (1) & (2)

$$v_1 + c = \sqrt{\frac{g y_2}{2 y_1} (y_1 + y_2)}$$

$[v_1 + c] \rightarrow$ vel. of surge relative to coa
{ +ve surge }

For -ve surge: $c > v_1$

$$\therefore \text{Rel. vel.} = c - v_1 = \sqrt{\frac{g y_2}{2 y_1} (y_1 + y_2)}$$

Special case: \downarrow

If surge is very small and depth of fl
Large than $y_1 \approx y_2 \approx y$ then

$$v_1 + c = \sqrt{g y}$$

For Rectangular channel: \downarrow

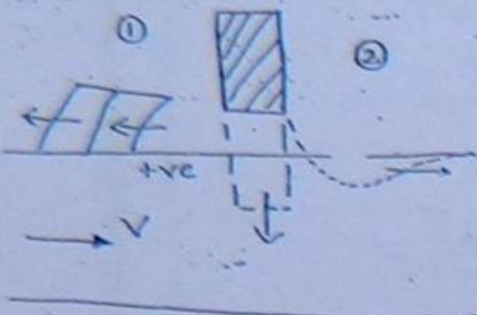
$$\frac{v_1}{\sqrt{g y}} = F_1$$

For critical Flow $v_1 = \sqrt{g y}$

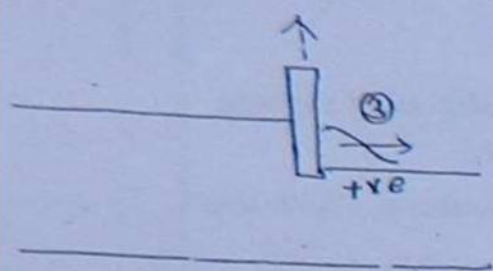
For subcritical Flow $v_1 < \sqrt{g y}$

For supercritical Flow $v_1 > \sqrt{g y}$

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→ If the flow in a channel is increased suddenly by means of opening gate a wave is formed which travels D/s.



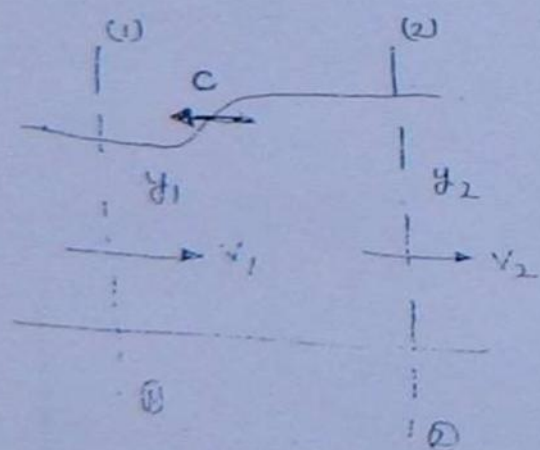
→ Similarly if a gate is suddenly closed and a flow is partially reduced, a wave formed travels such wave is called surge wave.

-ve surge wave:-

Also known as 'elevation surge wave' → If depth of water increases in the direction of motion of wave is called surge wave.

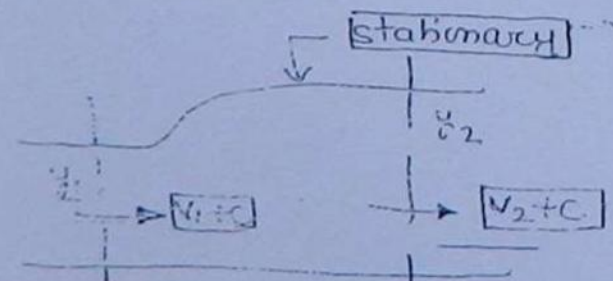
(1) & (3) are +ve surge wave & (2) is -ve surge wave

Analysis ↓



Let vel. of wave is c .
{celerity of wave}

For Analysis ↓



- (i) mild slope: - $y_0 > y_c \rightarrow$ subcritical flow at normal depth.
- (ii) steep slope: - $y_0 < y_c \rightarrow$ supercritical flow at normal depth.
- (13) (iii) critical slope: - $y_0 = y_c \rightarrow$ critical flow at normal depth.
- 213 (iv) horizontal slope: - $S_0 = 0 \rightarrow$ cannot sustain uniform flow
- 213 (v) adverse slope: - $S_0 < 0 \rightarrow$ " " " "

$\frac{dy}{dx} > 0$ if (a) $y > y_0$ and $y > y_c$ or
(b) $y < y_0$ and $y < y_c$

$\frac{dy}{dx} < 0$ if (i) $y_c > y > y_0$ or
(ii) $y_0 > y > y_c$

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- # (1) As $y \rightarrow y_0 \Rightarrow \frac{dy}{dx} \rightarrow 0$ i.e. The water surface approaches the normal depth line asymptotically.
- (2) As, $y \rightarrow y_c \Rightarrow \frac{dy}{dx} \rightarrow \infty$ i.e. water surface meets the critical depth vertically.
- (3) As, $y \rightarrow \infty, \frac{dy}{dx} \rightarrow S_0$ i.e. water surface meets a very large depth as a horizontal asymptote.

At critical depth the curves are indicated by dashed line to represent that the GVF eqn is strictly not applicable on that region.

- A control section is defined as a section on which a fixed relation exists b/w the discharge and depth of flow.
- critical depth is also a critical control point.
 - Subcritical flows have controls on the D/S end while supercritical flows have controls on the U/S end.

$$F_r = (P_2 - P_3) + \rho Q [V_2 - V_3]$$

$$= \rho g [A_2 \bar{x}_2 - A_3 \bar{x}_3] + \rho Q [V_2 - V_3]$$

$$= 1000 \times 9.81 \left[(1 \times 6) \times y_2 - (3 \times 6) \times \frac{1.5}{3} \right] +$$

$$1000 \times 65.77 [10.95 - 3.65]$$

$$F_r = 243.83 \text{ kN}$$

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Loss of energy = $E_1 - E_2$

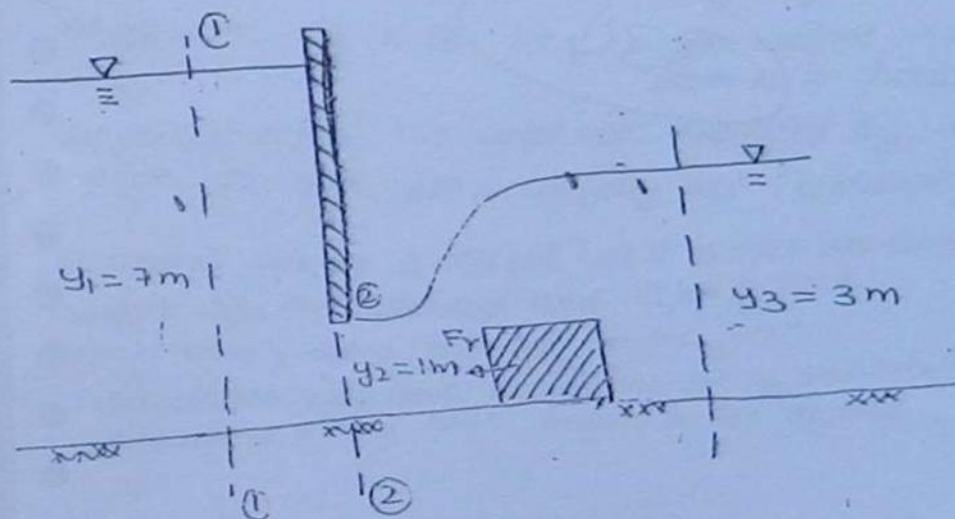
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$$\Delta E = \left(y_1 + \frac{v_1^2}{2g} \right) - \left(y_2 + \frac{v_2^2}{2g} \right)$$

Prob 2
ES/1992

A sluice across a channel is 6m wide, discharges a stream 1m deep. What is the flow rate when u/s of sluice is 7m. on the d/s side depth

a concrete block have been placed, to create the condition of hydraulic jump. Determine the force on the block if d/s depth is 3m



$$Q = (y_1 \times 6) v_1 = (y_2 \times 6) v_2 = (y_3 \times 6) v_3$$

$$\Rightarrow 7v_1 = v_2 = 3v_3 \quad \text{--- (1)}$$

There is NO LOSS of energy b/c (1)-(1) & (2)-(2)

So $E_1 = E_2$

$$\Rightarrow y_1 + \frac{v_1^2}{2g} = y_2 + \frac{v_2^2}{2g} \quad \text{--- (2)}$$

By eqn (1) & (2)

Power Loss in Jump = $WQ(\Delta E)$

Height of Jump = $y_2 - y_1$

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Length of Jump = 5 to 7 times $(y_2 - y_1)$

Fri

Type of Jump

1) 1-4.7

undular Jump

2) 1.7 to 2.5

weak Jump

3) 2.5 to 4.5 ^{golf}

oscillating Jump

4) 4.5 to 9.0 ^{golf}

steady Jump

5) > 9.0

standing Jump

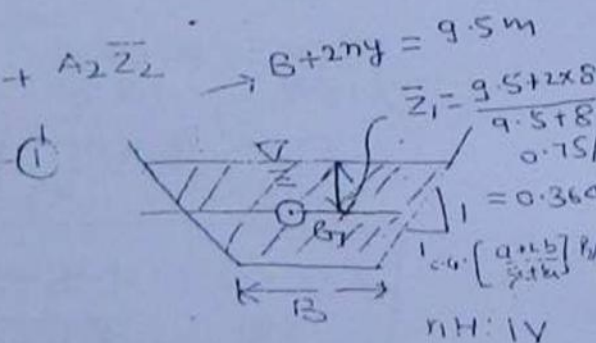
Prob 1

A Trapezoidal section having bottom width of 8 m & side slope is 1:1, carries a discharge of 30 m³/sec. Find the depth of conjugate to initial depth of 0.75 m before the jump. Also determine the loss of energy in the jump.

St Force

$F_1 = F_2$

$\frac{Q^2}{A_1 g} + A_1 \bar{Z}_1 = \frac{Q^2}{A_2 g} + A_2 \bar{Z}_2$



$A_2 = (B + y_2) y_2$

$\bar{Z}_2 = \frac{(8 + 2y_2) + 2 \times 8}{8 + 2y_2 + 8} \times y_2 / 3$

Solving for y_2 from eqn (1)

$y_2 = 4.167 \text{ m}$

$\Rightarrow n = 1$
 $A_1 = (B + ny_1) y_1$
 $= (8 + 0.75) \times 0.75$
 $= 6.56 \text{ m}^2$
 $\bar{Z}_1 = 0.36 \text{ m}$

Using eqn (iii) we can find y_2

Let's Find y_2

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$$y_2^2 \cdot y_1 + y_1^2 \cdot y_2 - \frac{2q^2}{g} = 0$$

$$\Rightarrow x^2 a + b \cdot x + c = 0$$

$$\Rightarrow y_2 = \frac{-y_1^2 \pm \sqrt{y_1^4 + 4y_1(2q^2/g)}}{2 \times y_1}$$

$$y_2 = +y_1/2 \left[-1 \pm \sqrt{1 + \frac{8q^2}{gy_1^3}} \right]$$

$$\frac{y_2}{y_1} = Y_2 \left[-1 \pm \sqrt{1 + \frac{8V_1^2}{gY_1}} \right]$$

$$\therefore \boxed{\frac{y_2}{y_1} = Y_2 \left[-1 + \sqrt{1 + 8 Fr_1^2} \right]} \quad \text{only}$$

$$\text{OR} \quad [Fr_1 > 1]$$

> only for Rect. Section

$$\boxed{y/y_2 = Y_2 \left[-1 + \sqrt{1 + 8 Fr_2^2} \right]}$$

Loss of Energy in Jump: ↓

$$\Delta E = E_1 - E_2$$

$$= \left(y_1 + \frac{V_1^2}{2g} \right) - \left(y_2 + \frac{V_2^2}{2g} \right)$$

$$\boxed{\Delta E = \frac{[y_2 - y_1]^3}{4 y_1 y_2}} \quad \text{only}$$

OR

$$\boxed{\Delta E = \frac{[V_1 - V_2]^3}{2g(V_1 + V_2)}}$$

~~At minimum Sp. force there is only one critical depth.~~
 For other values of F there are two depths of flow y_1 & y_2 , called conjugate depth.

$$y_1 < y_c < y_2 \quad \text{galt}$$

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For critical Flow, sp. force is Minimum
 for Rect. channel

$$F = \frac{q^2 B}{g} \times \frac{1}{y} + \frac{B y^2}{2}$$

For $F(\min)$, $\frac{dF}{dy} = 0$

$$\Rightarrow \frac{q^2 B}{g} \left(-\frac{1}{y^2}\right) + \frac{2 B y}{2} = 0$$

$$\Rightarrow y_c = \left(\frac{q^2}{g}\right)^{1/3} \quad \text{--- (i)}$$

NOTE:

For Non-Rectangular channel

$$\left[\frac{Q^2}{g} = \frac{A^3}{T} \right] \quad \text{--- (ii) at critical Flow.}$$

$$F_1 = F_2$$

$$\Rightarrow \frac{Q^2}{A_1 g} + A_1 \bar{Z}_1 = \frac{Q^2}{A_2 g} + A_2 \bar{Z}_2$$

$$[Q = q \cdot B \text{ For Rect. channel}]$$

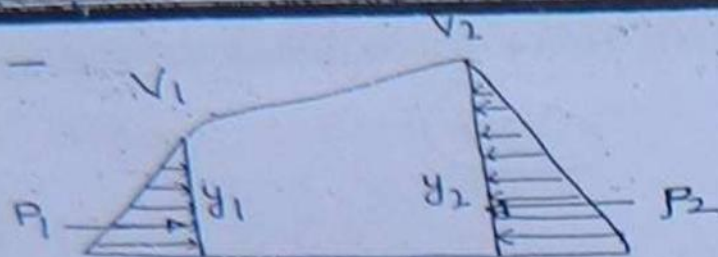
$$\frac{q^2 B^2}{B y_1 g} + B y_1 \left(\frac{y_1}{2}\right) = \frac{q^2 B^2}{y_2 B g} + B y_2 \left(\frac{y_2}{2}\right)$$

or on solving

$$2 \frac{q^2}{g} = y_1 y_2 (y_1 + y_2) \quad \text{--- (iii)}$$

$$\Rightarrow y_c^3 = \frac{y_1 y_2 (y_1 + y_2)}{2} \quad \text{galt --- (iv)}$$

For alternate depth $\rightarrow F_1 = F_2 \rightarrow y_c^3 = \frac{2 y_1^2 y_2^2}{(y_1 + y_2)} \quad \checkmark$



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If Resultant Force is in equilibrium then $\Sigma F = 0$

$$\Rightarrow \underbrace{(P_1 - P_2)}_{\text{Static Force}} + \underbrace{\rho Q [V_1 - V_2]}_{\text{Dynamic Force}} = 0$$

$$P_1 = \omega A_1 \bar{x}_1 \quad \left[\bar{x}_1 = \text{Position of C.G. From Top surface} \right]$$

$$P_2 = \omega A_2 \bar{x}_2$$

$$\therefore \omega [A_1 \bar{x}_1 - A_2 \bar{x}_2] + \rho Q [V_1 - V_2] = 0$$

$$\Rightarrow \omega [A_1 \bar{x}_1 - A_2 \bar{x}_2] = \rho Q \left[\frac{Q}{A_2} - \frac{Q}{A_1} \right]$$

$$\Rightarrow \cancel{\omega} g [A_1 \bar{x}_1 - A_2 \bar{x}_2] = \cancel{\rho} Q \left[\frac{Q}{A_2} - \frac{Q}{A_1} \right]$$

$$\Rightarrow g [A_1 \bar{x}_1 - A_2 \bar{x}_2] = \frac{Q^2}{A_2} - \frac{Q^2}{A_1}$$

$$\Rightarrow A_1 \bar{x}_1 + \frac{Q^2}{A_1 g} = A_2 \bar{x}_2 + \frac{Q^2}{A_2 g}$$

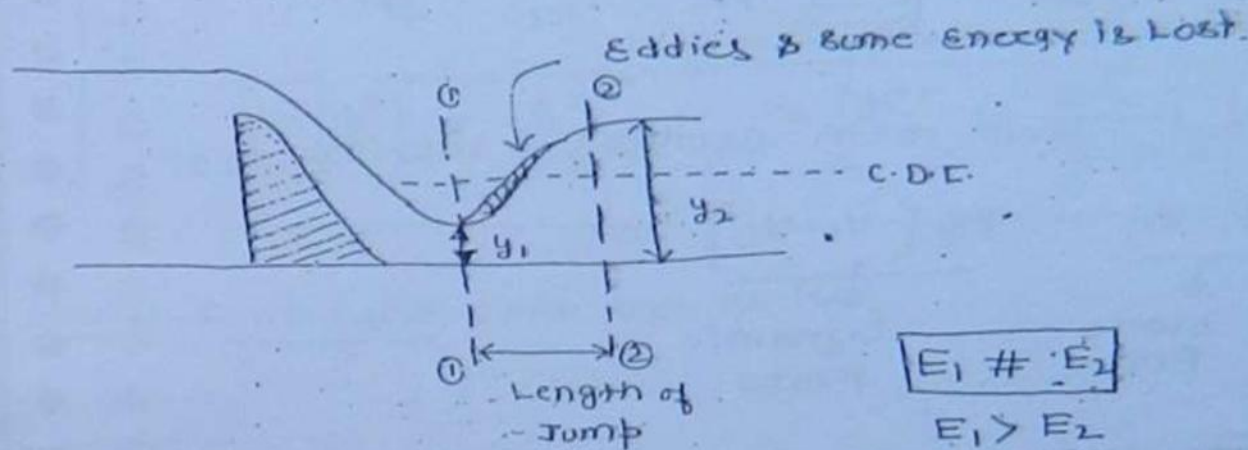
$$\Rightarrow \boxed{A\bar{x} + \frac{Q^2}{Ag} = \text{constant}} \rightarrow \text{specific Force } F = \text{const.}$$

Generally For Rect. channel $\bar{x} = \bar{z} = y/2$

[Either from top or bottom]

Rapid Varied Flow (RVF)

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$$E_1 \neq E_2$$

$$E_1 > E_2$$

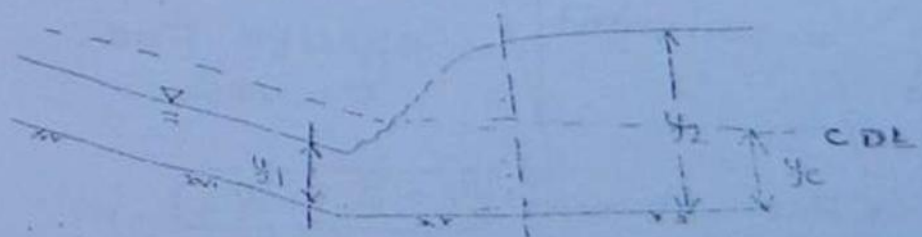
$$E_1 - E_2 = \Delta E \quad (\text{Loss of Energy})$$

An essential & necessary condition for hydraulic jump to occur Flow must change supercritical to subcritical & this change is over a small Length hence if flowing fluid having Froude No. greater than 1, jump may be created.

Hydraulic jump is defined as sudden & turbulent Passes of water supercritical to subcritical & is also called shooting, rapid, Tranquil, unstable type.

→ There is considerable dissipation of energy during the formation of jump.

sp. Force concept is always applied with hydraulic jump i.e. Flow changing from supercritical to subcritical.



$$E_1 \neq E_2$$

$$y_1 < y_2 < y_c$$

- But $F_1 = F_2$ (sp. force)

y_1 & y_2 are subsequent depth or conjugate depths

$$y_2 = \frac{(v_1^2) V_1}{(g)} \quad \frac{(3.82^2) 1.5}{(9.81)}$$

$$\Delta Z_{\max} = E_1 - 3/2 \times 1.81$$

$$\begin{aligned} E_1 &= y_1 + \frac{v_1^2}{2g} \\ &= 2.0 + \frac{3.82^2}{2 \times 9.81} \\ &= 2.74 \text{ m} \end{aligned}$$

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$$\begin{aligned} \Delta Z_{\max} &= 2.74 - 1.5 \times 1.81 \\ &= 0.026 \text{ m} \end{aligned}$$

b) If Now $y_1' = 2.4 \text{ m}$ (V.L. would change)
 Discharge would remain same, y_1 would change

$$y_1' + \frac{v_1'^2}{2g} = \Delta Z^1 + E_{\min} \quad [\Delta Z^1 > \Delta Z_{\max}]$$

$$2.4 + \frac{v_1'^2}{2g} = \Delta Z^1 + 3/2 \times 1.812$$

$$\Rightarrow 2.4 + \frac{(3.18)^2}{2 \times 9.81} = \Delta Z^1 + 1.812 \times 1.5 \quad \left[\begin{aligned} v_1' &= Q/A_1' \\ &= \frac{26.74}{3.5 \times 2.4} \\ &= 3.18 \text{ m/sec} \end{aligned} \right]$$

$$\Delta Z^1 = 0.198 \text{ m}$$

Probs

Water flows at a depth of 1.6 m and velocity of 1.1 m/sec in a open channel of Rect. section of width 4m. At a certain section width is reduced to 3.5 m and the bed is raised by 0.35 m through a smooth flat hump. Calculate water surface elevation at the contracted section as well as the vls section. Neglect losses.

At section (2) depth = 1.158 m [$y_2 > y_c$]
(y_2)

At section (1) elevation is = $y_1 = 1.6 \text{ m}$
raised

$$\Delta y = y_1 - y_2 - y_3$$

$$= 1.524 - 0.312 - 0.914$$

$$= 0.294 \text{ m}$$

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Prob 4

A rect. channel 3.5 m wide laid at a slope of 0.0036, uniform flow occurs at a depth of 2.0 m. Find how high a hump can be raised on the channel bed without causing a change in up depth. If the up depth is to be raised to 2.4 m, what should be the height of hump. Assume Manning's $N = 0.015$.

$$A_1 = 3.5 \times 2.0 = 7 \text{ m}^2$$

$$P_1 = 3.5 + 2 \times 2 = 7.5 \text{ m}$$

$$R_1 = A_1 / P_1 = 0.938$$

$$V_1 = \frac{1}{N} R_1^{2/3} S^{1/2}$$

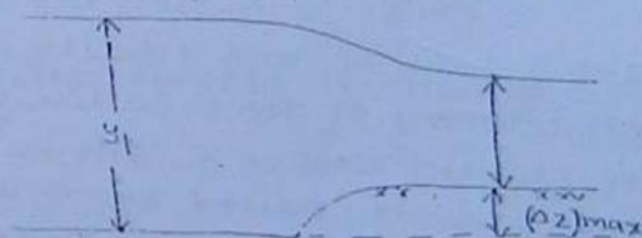
$$= 3.82 \text{ m/sec}$$

check for Froude Number

$$F_1 = \frac{V_1}{\sqrt{g y_1}} = \frac{3.82}{\sqrt{9.81 \times 2}} = 0.86 < 1$$

→ subcritical flow

TEL -----



9) If up depth is not to be changed than at equilibrium

$$E_1 = (\Delta z)_{\max} + E_{\min}$$

$$(\Delta z)_{\max} = E_1 - E_{\min}$$

$$Q = V_1 \times A_1 = 3.82 \times 7$$

$$Q/B_1 = \frac{1}{2} = 3.82 \times 7 / 9.5 = 7.64 \text{ m}^3/\text{sec/m}$$

$$\Rightarrow y_2 = 1.8 - 0.3 - 0.15 = 1.35 \text{ m}$$

By using eqn (ii) -

$$\frac{Q^2}{2g A_2} - \frac{Q^2}{2g A_1} = 0.15$$

$$\Rightarrow \frac{Q^2}{2g} [y_{A_2} - y_{A_1}] = 0.15$$

$$\Rightarrow Q = 6.418 \text{ m}^3/\text{sec}$$

$$\begin{cases} A_1 = 3.6 \times 1.8 \\ A_2 = 2.4 \times 1.35 \end{cases}$$

3.24

2068
S/2005

A wide rect channel carries a flow of $2.76 \text{ m}^3/\text{sec}$.
The depth of flow is 1.524 m .

a) calculate the min. rise of flow at a section required to produce critical flow conditions

b) what is corresponding fall in the water level

Given

$$Q = 2.76 \text{ m}^3/\text{sec}/\text{m}$$

$$y_1 = 1.524 \text{ m}$$

$$\Delta Z_{\text{max}} = ? \quad (\text{For critical flow condition})$$

At critical Flow condition at ^{Section} ②

$$E_1 = \Delta Z_{\text{max}} + E_{\text{min}}$$

$$\begin{aligned} V_1 &= Q/y_1 \\ &= \frac{2.76}{1.524} \\ &= 1.81 \text{ m/sec} \end{aligned}$$

$$\begin{aligned} E_1 &= y_1 + \frac{V_1^2}{2g} \\ &= 1.524 + \frac{1.81^2}{2 \times 9.81} \\ &= 1.69 \text{ m} \end{aligned}$$

$$E_{\text{min}} = \frac{3}{2} (y_c)$$

$$\begin{aligned} y_c &= \left(\frac{Q^2}{g} \right)^{1/3} \\ &= 0.949 \text{ m} \end{aligned}$$

$$\begin{aligned} \Delta Z_{\text{max}} &= E_1 - E_{\text{min}} \\ &= 1.69 - 1.5 \times 0.949 \\ &= 0.342 \text{ m} \end{aligned}$$

$$\left(\frac{F_2}{F_1} \right)^3 = \frac{2 + F_2^2}{2 + F_1^2}$$

$$\left[F_1 = \frac{V_1}{\sqrt{g y_1}}, F_2 = \frac{V_2}{\sqrt{g y_2}} \right]$$

223

$$y_1^3 = \frac{Q^2}{B^2 g F_1^2}$$

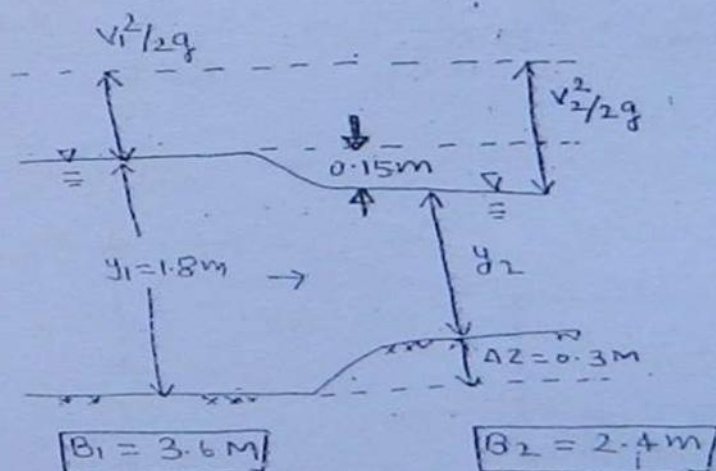
$$y_2^3 = \frac{Q^2}{B^2 g F_2^2}$$

$$y_1/y_2 = \left(F_2/F_1 \right)^{2/3}$$

$$\left(F_2/F_1 \right)^{2/3} = \frac{2 + F_2^2}{2 + F_1^2}$$

Prob 2

A 3.6 m wide rect. wide channel carries a water depth of 1.8 m. In order to measure the discharge, channel width is reduced to 2.4 m and hump of 0.15 m is provided in the bottom. calculate the drop of the water surface in the contracted section. Assume no losses.



$$E_1 = \Delta Z + E_2$$

$$y_1 + \frac{V_1^2}{2g} = \Delta Z + y_2 + \frac{V_2^2}{2g} \quad \text{--- (1)}$$

$$\frac{V_2^2}{2g} - \frac{V_1^2}{2g} = -0.15 \text{ m} \quad \text{--- (2)}$$

From (2) in Eqn (1)

Qb 1

and a flow through a rect. channel for a certain discharge the Froud No. corresponding to two alternate depths are F_1 & F_2 than prove

$$\left(\frac{F_2}{F_1}\right)^{2/3} = \frac{2 + F_2^2}{2 + F_1^2}$$

(322)

$$F_2 = \frac{V}{\sqrt{gY}} = \frac{V}{\sqrt{gD}}$$

$$E_1 = E_2$$

$$y_1 + \frac{V_1^2}{2g} = y_2 + \frac{V_2^2}{2g}$$

$$\Rightarrow y_1 \left[1 + \frac{V_1^2}{2gy_1} \right] = y_2 \left[1 + \frac{V_2^2}{2gy_2} \right]$$

$$\therefore y_1/y_2 = \frac{1 + F_2^2/2}{1 + F_1^2/2} = \frac{2 + F_2^2}{2 + F_1^2}$$

$$\begin{aligned} \text{Now } E_1 &= \frac{V_1}{\sqrt{gY}} \\ &= \frac{Q}{By_1 \sqrt{gY_1}} \\ &= \frac{Q}{B \sqrt{gY_1^3}} \end{aligned}$$

$$\text{Similarly } F_2 = \frac{Q}{B \sqrt{gY_2^3}}$$

$$\therefore y_1^3 = \frac{Q^2}{3^2 g F_1^2}$$

$$y_2^3 = \frac{Q^2}{3^2 g F_2^2}$$

$$A = \frac{(2.8 + 0.5y)}{2} \times y$$

$$A = (1.3 + 0.25y) \times y$$

$$\therefore B = A/y - 0.75y$$

$$\therefore P = A/y - 0.75y + 2.8y$$

$$\therefore dP/dy = -A/y^2 - 0.75 + 2.8 = 0$$

$$\Rightarrow y^2 = \frac{A}{2.05} \Rightarrow \boxed{A = 2.05y^2}$$

$$\therefore \boxed{B = 1.3y}$$

321

For
Given Area
Perimeter
should be minimum.

By applying Manning's Equation

$$Q = V_N \cdot A \cdot R^{2/3} S^{1/2} \Rightarrow \text{Get } (y) = ?$$

channel with a hump: ↓

a) subcritical Flow: ↓

when $c < \Delta Z \leq \Delta Z_{max}$

the U/S water level remains stationary at y_1 while the depth of flow at section 2

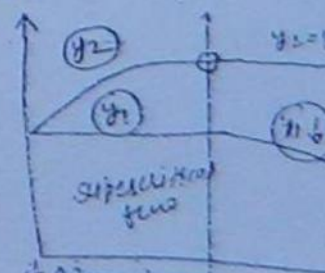
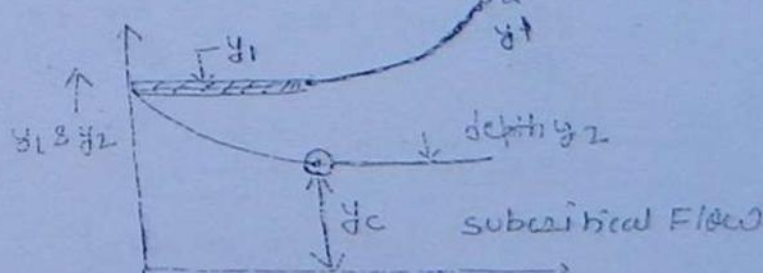
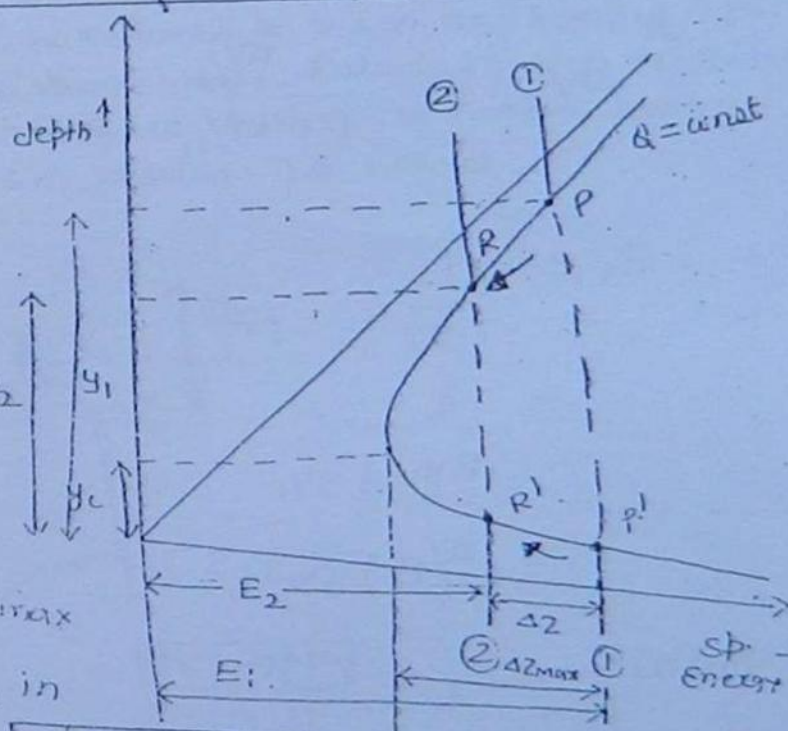
decreases with ΔZ reaching at minimum

value of y_c at $\Delta Z = \Delta Z_{max}$

with further increase in

the value of ΔZ i.e. $\Delta Z > \Delta Z_{max}$

y_1 will change to y'_1 while y_2 will continue to remain at y_c



15/11/1985

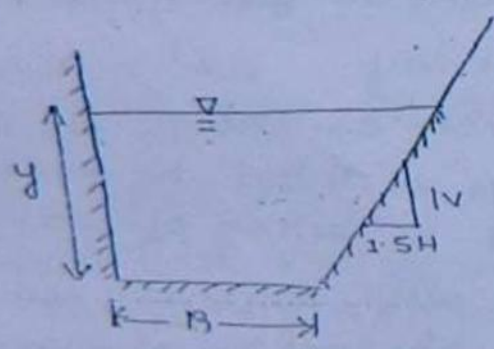
A trapezoidal channel of base width 6m and side slope of 2H:1V carries a flow of 60 m³/s at a depth of 2.5 m. There is a smooth transition to a rect. section 6m wide accomplished by a gradual lowering of channel bed by 0.6 m

- 1) Find the depth of water in the rect. section & change in water surface level.
- 2) In case the drop of water surface level is to be restricted to 0.30 m what is the amount of bed must be lowered

320

Prob 2
Uniform
Flow
Problem 2

A lined channel ($N = 0.014$) is of trapezoidal section with one side vertical and other side on slope (1.5H:1V) If the channel is to be deliver 9 m³/sec when laid on a slope of 0.0002. calculate the dimensions of efficient section which requires minimum lining. Also calculate ^{corresponding} Mean velocity.



$\left[\begin{array}{l} nH:1V \\ n=1.5 \end{array} \right.$

For Minimum Lining, P should be Minimum

$$\begin{aligned}
 P &= y + B + y \sqrt{n^2 + 1} \\
 &= B + y [1 + \sqrt{3.25}]
 \end{aligned}$$

a =

$$A = \frac{B + (B + ny)}{2} \times y$$

(319)

$$(y_2 < y_1)$$

$$E_1 = \Delta Z + E_2$$

OR

$$E_1' = \Delta Z + E_2'$$

→ If flow $Fr > 1 \rightarrow \{y_1' < y_c\}$

$$(y_2' > y_1')$$

→ The max height of hump ($\Delta Z = \text{Max}$) will be obtain when point (2) in the sp. energy curve or when depth of section (2) coincides with critical depth at that section. ($y_2 = y_c$)
Thus flow over raised section will be critical.

Then

$$E_1 = (\Delta Z)_{\text{max}} + E_{\text{min}}$$

$$E_2 = E_{\text{min}}$$

$$(\Delta Z)_{\text{max}} = E_1 - E_{\text{min}} \quad \text{self}$$

NOTE:

if the height of hump is further ~~increased~~ beyond ΔZ , Let's say $\Delta Z' > (\Delta Z)_{\text{max}}$ than flow at given sp. energy E_1 & at given depth y_1 will not be possible which will result in piling of water. Hence depth y_1 will increase, causing an increase in sp. energy of approach flow.

water level at section (1) will continue to rise until at section (2) flow becomes critical.

The increased sp. energy E_1'' at (1) is obtained as

$$E_1'' = E_{\text{min}} + \Delta Z'$$

$$= \frac{3}{2} y_c + \Delta Z'$$

→ For Rect. channel

~~Case I: - when at section (1) depth is y_1 and~~

flow is subcritical. ($Fr < 1$)

(318)

If flow is transform such that V_2 decreases than Flow will tend to become critical and depth will tend to become critical hence Lowering of water surface and type-a curve will be formed

Case II: -

If flow is supercritical at section (1) - (2) than at section (2), flow tend to become critical hence flow will rise \wedge will be observed. In actual curve

If flow is through smooth transition and there is no change in bed level than $E_2 = E_1$

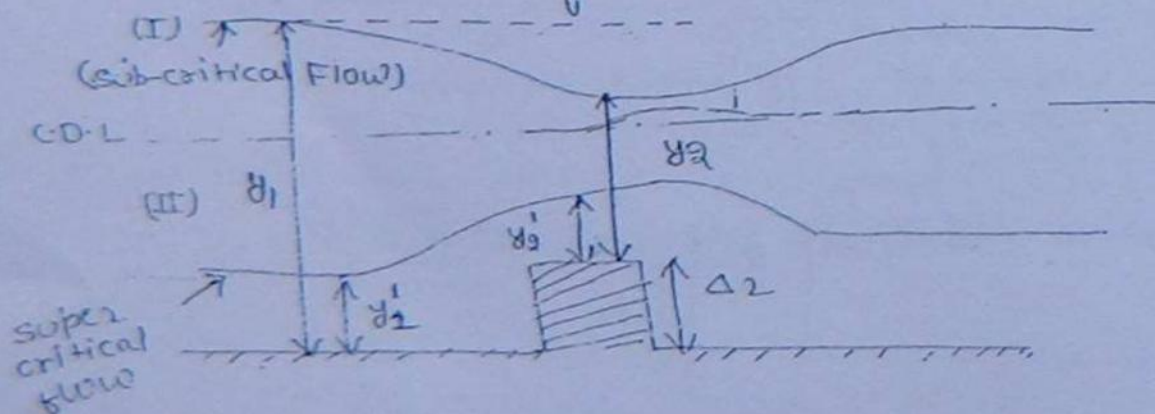
NOTE: - If B_2 is further contracted even after E_2 has reached E_{min} (corresponding to critical stage) than piling of water in V/s will occur and flow will not be possible at original depth y_2 .

Imp

- flow over Local rise in the channel slope: \downarrow

Def Flow over Local rise is called 'hump flow'.

Let's consider a rectangular channel having its bottom raised by an amount Δ .



Applications of sp. Energy :

- 1) Analysis of flow through the channels when one section is transformed into another section. Such channels are called transition channels.
- 2) Flow over raised channel bottom slope.
Ex: Broad crested weir
- 3) Flow through sluice gate opening.

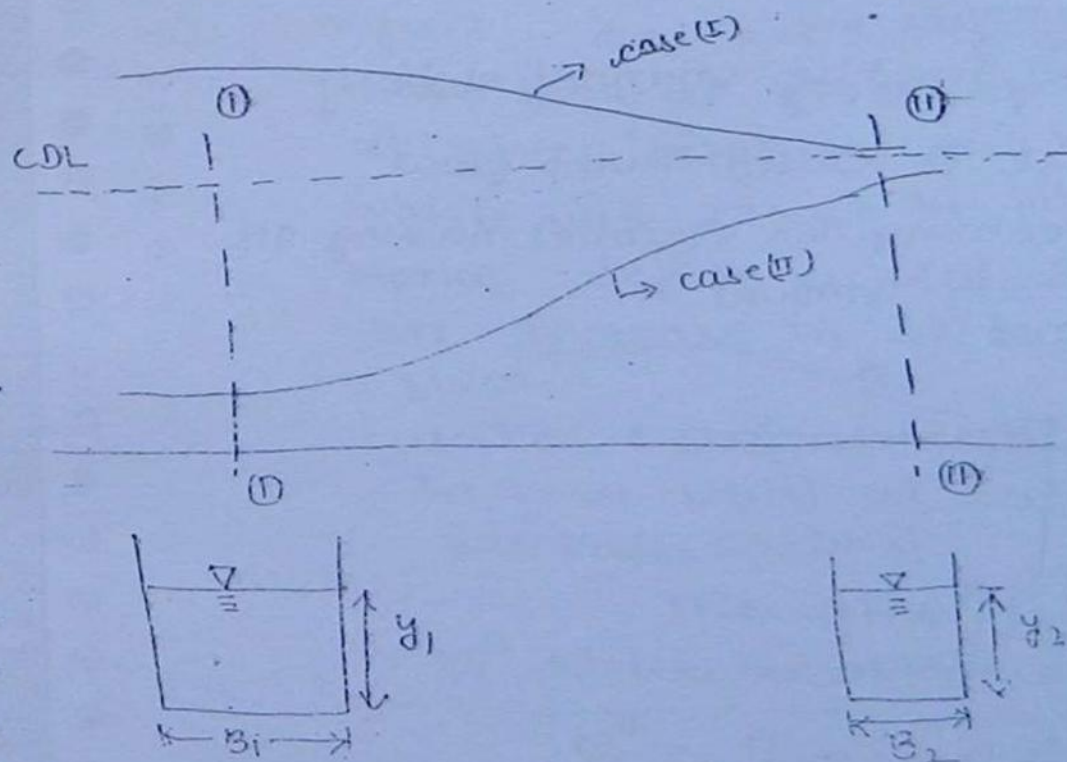
(317)

Flow through Rectangular channel Transition :

when the width of the channel is reduced, it can be done by

- a) Sudden contraction
- b) Gradual contraction

An sudden contraction Loss of Energy is much more than Gradual contraction therefore all practical problems Gradual contraction is preferred.



$$= 1.6 + \frac{1.875^2}{2 \times 9.81}$$

$$= 1.779$$

$$V_1 = \frac{1.6 \times 10}{1.6 \times 10} = 1.0$$

(316)

At mid depth

$$y = \frac{1.6 + 2.0}{2} = 1.8 \text{ m}$$

$$Q = Y_N \cdot A R^{2/3} S_e^{1/2}$$

$$30 = \frac{1}{[0.015]} \times [10 \times 1.8] \left[\frac{10 \times 1.8}{10 + 3.6} \right]^{2/3} \times S_e^{1/2}$$

$$\Rightarrow S_e = 0.00043$$

$$\therefore \Delta X = \frac{E_2 - E_1}{S_0 - S_e} = \frac{2.11 - 1.779}{0.0001 - 0.00043} = (-) 1003$$

$$\therefore \Delta X = 1.003 \text{ km} \rightarrow \text{u/s to section (1)}$$

ob2

choking ↓

In the case of a channel with a bump and also in the case of a width contraction, it is observed that the u/s water-surface elevation is not affected by the conditions at section (2) till a critical stage is first achieved.

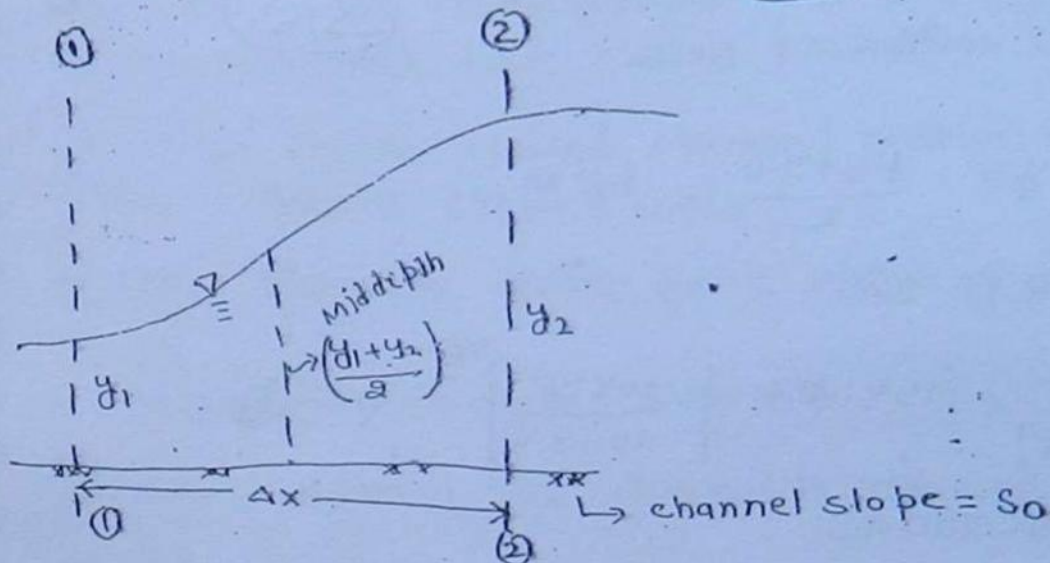
So in the case of a bump for all $\Delta Z \leq \Delta Z_{max}$, the u/s water depth is constant and for all $\Delta Z > \Delta Z_{max}$ the u/s depth is different from y_1 . Similarly in the case of width contraction for $B_2 > B_{2min}$ the u/s depth y_1 is constant while for all $B_2 < B_{2min}$, the u/s depth undergoes a change.

This creates a critical condition at section (2) and is known as choking.

Thus for each case with $\Delta Z > \Delta Z_m$ or $B_2 < B_{2m}$ was known as choked conditions.

Approximate Method to determine distance b/w two depths in G.V.F. \downarrow

(315)



$$E_1 = y_1 + \frac{V_1^2}{2g}$$

$E_2 =$ Sp. Energy at section 2

$$= y_2 + \frac{V_2^2}{2g}$$

$$\Delta x = \frac{E_2 - E_1}{S_o - S_e}$$

slope of Energy line at mid section

At mid section $y = \frac{y_1 + y_2}{2}$

S_e is given by

$$R = \frac{V}{S_e} \quad A R^{2/3} S_e^{1/2}$$

$A = B \cdot y \rightarrow$ depth at mid section

$p = B + 2y, \quad R = A/p$

Ref. Ex:

$$E_2 = y_2 + \frac{V_2^2}{2g}$$

$$= 2.0 + \frac{1.5^2}{2 \times 9.81}$$

$$= 2.11 \text{ m}$$

$$\left[V_2 = \frac{Q}{A_2} = \frac{30}{10 \times 2.0} = 1.5 \right]$$

Normal depth of flow

$$A = 10 y_n$$

314

$$Q = y_N A R^{2/3} S_o^{1/2}$$

$$30 = \frac{1}{0.015} (10 y_n) \left[\frac{10 y_n}{10 + 2 y_n} \right]^{2/3} (0.0001)^{1/2}$$

$$\therefore y_n^{5/2} = 1.209 (y_n + 5)$$

$$\Rightarrow y_n = 2.97 \text{ m (By trial & bit method)}$$

2) Find critical depth

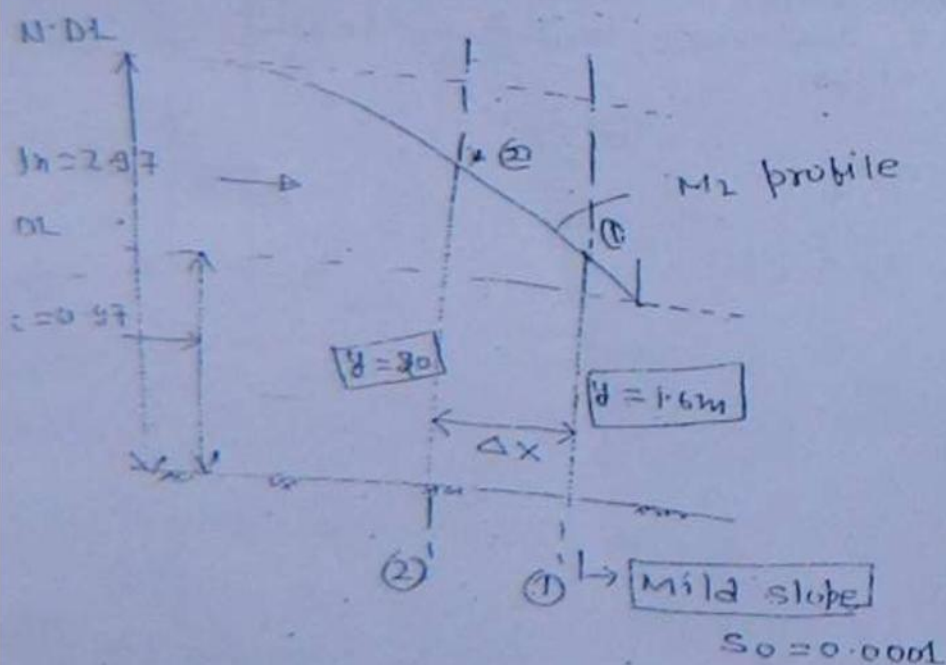
$$q = Q/B = 30/10 = 3 \text{ m}^3/\text{sec/m}$$

$$\therefore y_c^3 = (q^2/g) \rightarrow \text{For Rectangular}$$

$$\therefore y_c = \left(\frac{3^2}{9.81} \right)^{1/3} = 0.97 \text{ m}$$

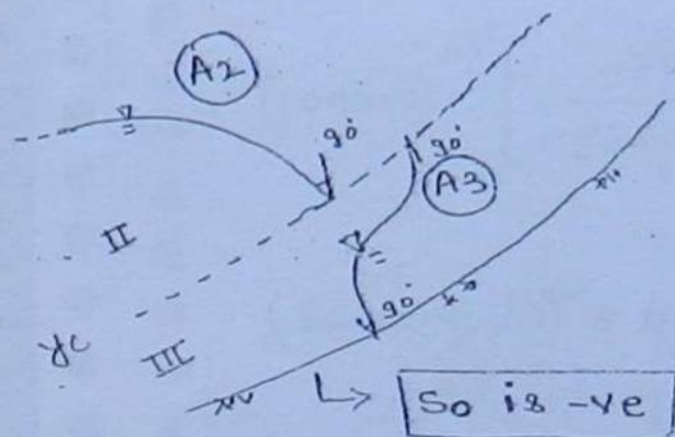
since $y_n > y_c$ the given slope is mild slope

since actual depth of flow are b/w y_c & y_n
therefor surface profile will be M_2 .



Adverse Slope:

313



A_2 & A_3

* y_n will be imaginary so Normal depth line does not exist.

≠ A_1 zone will not exist and only A_2 & A_3 will exist.

Prob 1

A Rectangular channel 10m wide, carry a discharge $30 \text{ m}^3/\text{sec}$. It is laid at a slope of 0.0001 . At a section in this channel the depth of flow is 1.6 m . Find whether upstream or downstream from this section the depth of flow is 2 m . Also determine the surface profile type. Assume $N = 0.015$. Also determine the distance b/w two depths along the channel slope.

Find Normal depth of flow ↓

$$Q = 30 \text{ m}^3/\text{sec}$$

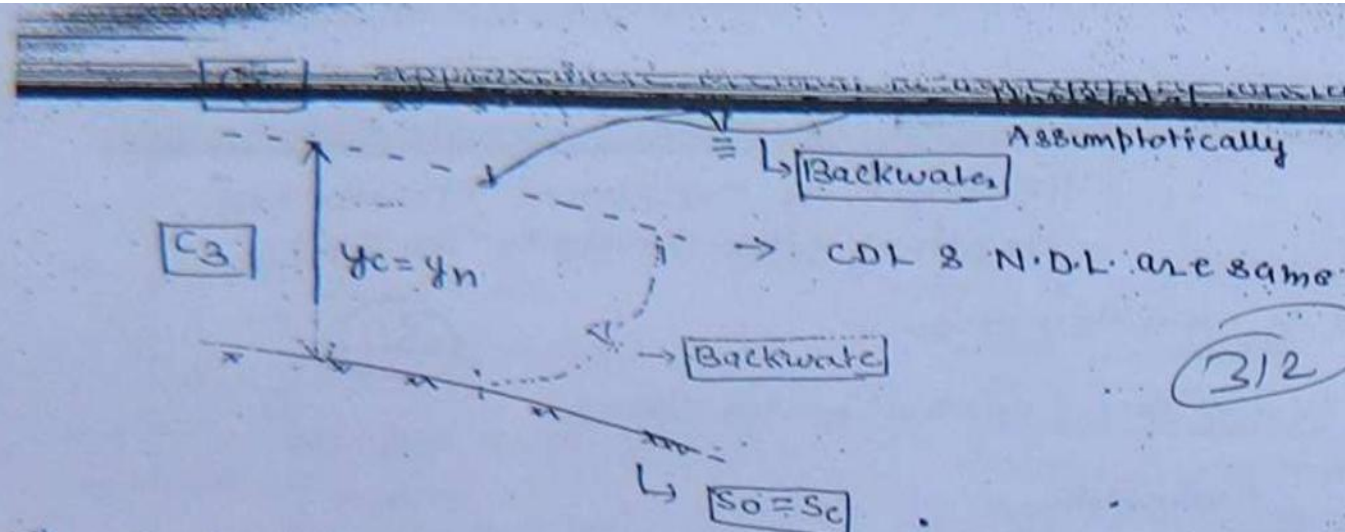
$$b = 10 \text{ m}$$

Section 1

$$y = 1.6 \text{ m}$$

At section 2

$$y = 2 \text{ m}$$

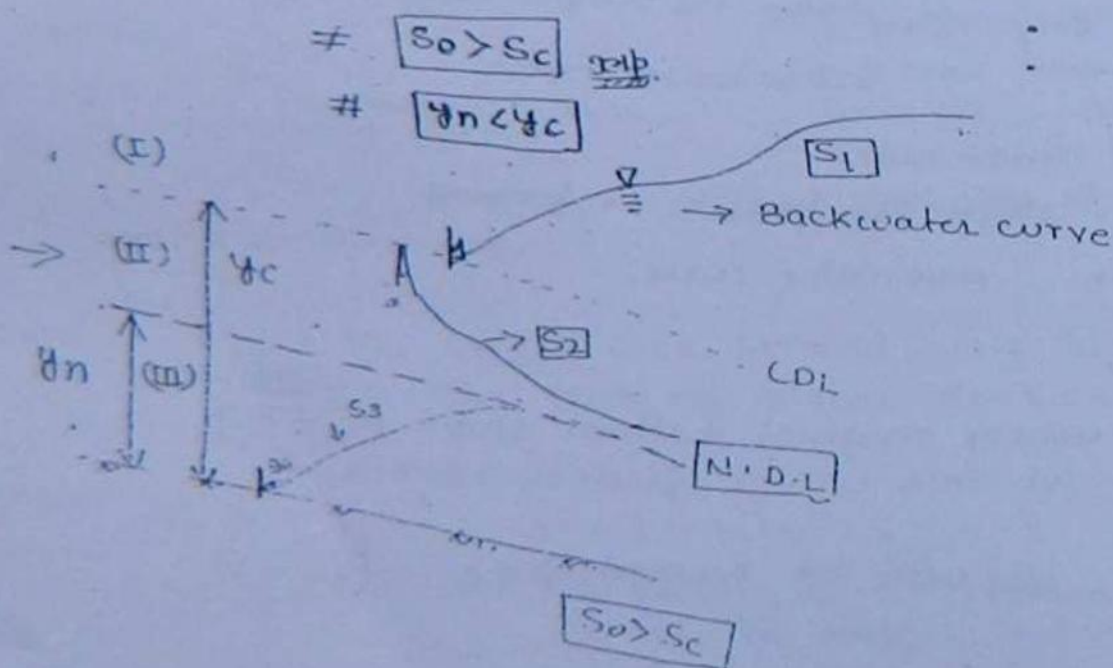


c_2 profile does not exist.

if $y > y_n \Rightarrow c_1$ profile
 $(= y_c) \Rightarrow$ Backwater curve

if $y < y_n \Rightarrow c_3$ profile
 $(= y_c) \Rightarrow$ drawdown curve.
 Backwater

steep slope: \downarrow



S_1 profile \Rightarrow when $y > y_c$

S_2 profile \Rightarrow when $y_n \leq y \leq y_c$

S_3 profile \Rightarrow when $0 < y < y_n$

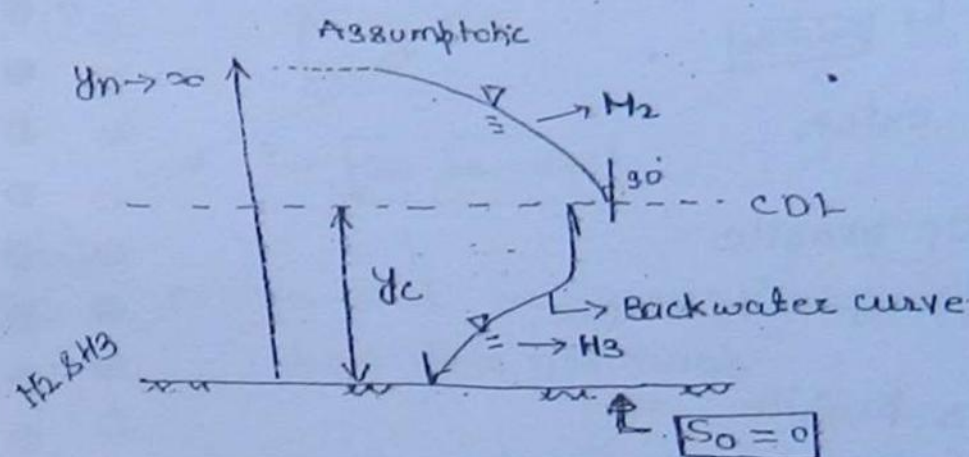
2) Horizontal slope: \downarrow

If slope is horizontal then primary depth of flow tends to infinity.

$$d \propto y_N \propto R^{2/3} S^{1/2} \rightarrow 0$$

(311)

$$\Rightarrow \boxed{y_N \rightarrow \infty} \text{ (For horizontal slope)}$$



H_1 profiles are not formed and H_1 zone does not exist

If $\boxed{y_c \leq y \leq y_N} \rightarrow$ then H_2 profile is formed
 \rightarrow drawdown curve.

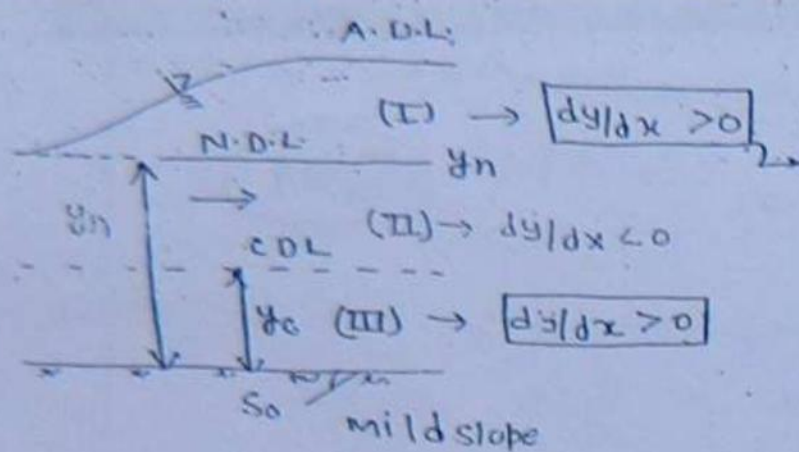
If $\boxed{y < y_c} \rightarrow$ then H_3 profile is formed.
 \rightarrow Backwater curve.

3) critical slope: \downarrow

1) when channel bottom slope $\boxed{S_0 = S_c}$ at this stage flow is critical

(9193)

2) y_N will be equal to y_c . $\boxed{y_N = y_c}$



In mild slope

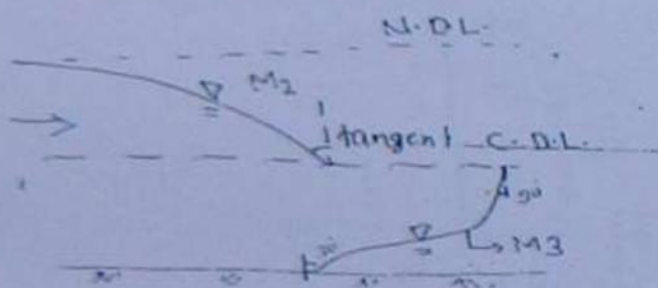
$$S_o < S_c$$

3/0

curve is backwater on the direction of flow. i.e. M1 profile meets asymptotically N.D.L. and rises on the direction of flow.

When

$y_c < y < y_n$, then M2 profile is formed



M2 profile meets asymptotically to N.D.L. and meets normal to C.D.L. The curve is draw-down in the direction of flow.

NOTE

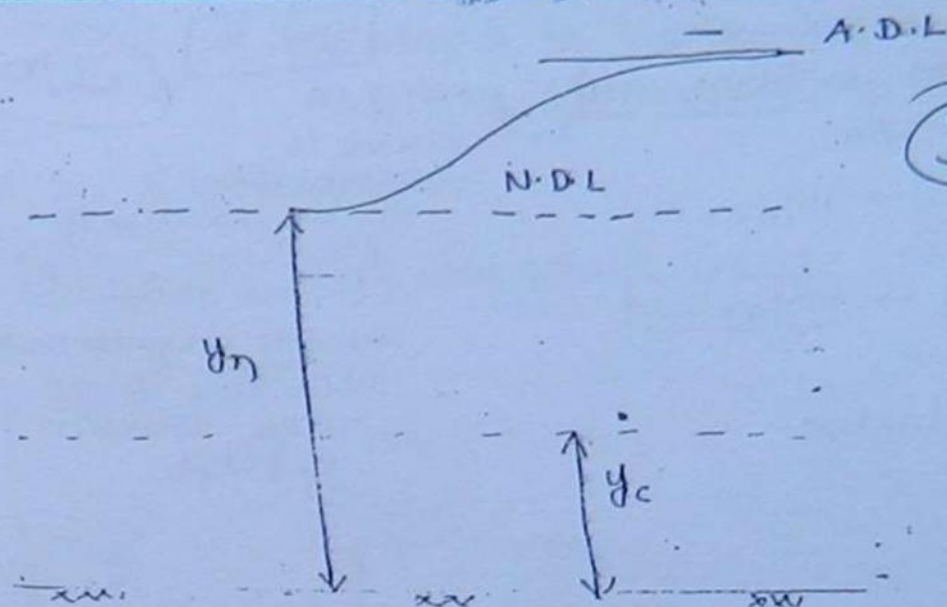
M1 & M2 profile are formed when flow are of stable type having Froude No. Less than 1

When

$0 < y < y_c \rightarrow$ M3 profile is formed

\rightarrow M3 profile is backwater curve which is normal to channel bottom slope & C.D.L.

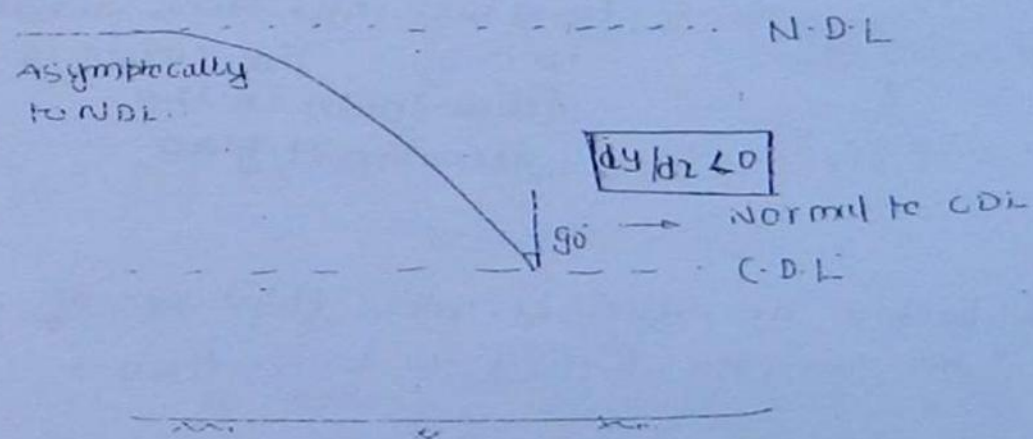
\rightarrow M3 curve is formed when flow is supercritical



309

9f.

$y_c \leq y \leq y_n \rightarrow M_2$ profile will be formed
 \rightarrow Means asymptotically to normal depth and Normally to critical depth in direction of flow. The curve is drawn



* * * M_1 & M_2 curve profile are formed when $Fr < 1$

1. Determine normal depth of flow by y_n .

(308)

$$Q = y_n \cdot A R^{2/3} S^{1/2}$$

$$\text{OR } Q = C \cdot A \cdot R^{1/2} S^{1/2}$$

2. Determine critical depth of flow y_c .

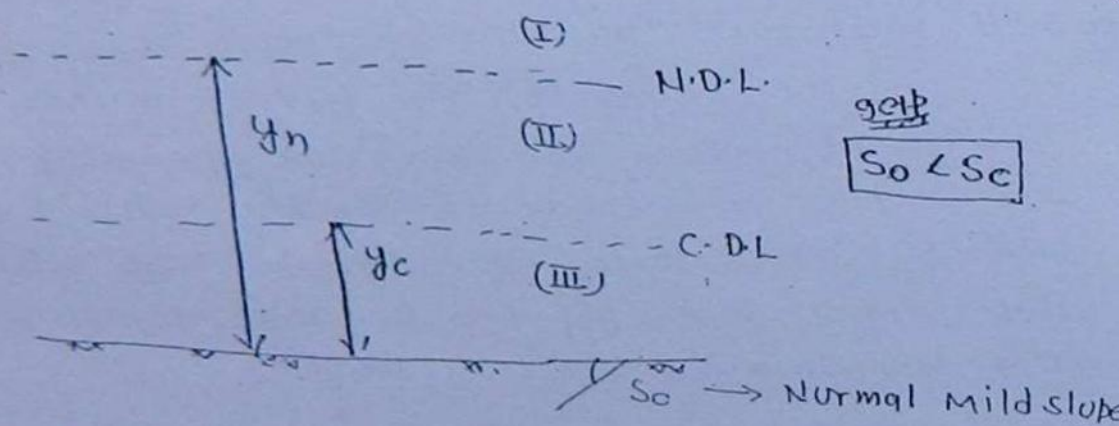
$$\boxed{Q^2/g = A^3/T} \rightarrow \text{In General}$$

$$\boxed{Q^2/g = y_c^3} \rightarrow \text{For Rectangular channel}$$

3. Actual depth of flow y .

(a) Mild slope: \downarrow

when $\boxed{y_n > y_c}$



If actual depth of flow y is such that

$$\boxed{y_c < y < y_n} \quad \boxed{y > y_n > y_c}$$

\rightarrow Then surface profile is M1 type

M1 profile means normal depth line is asymptotically and tend to become horizontal in d/s towards actual depth line. The curve water is backwater and rising hence

CRITICAL DEPTH

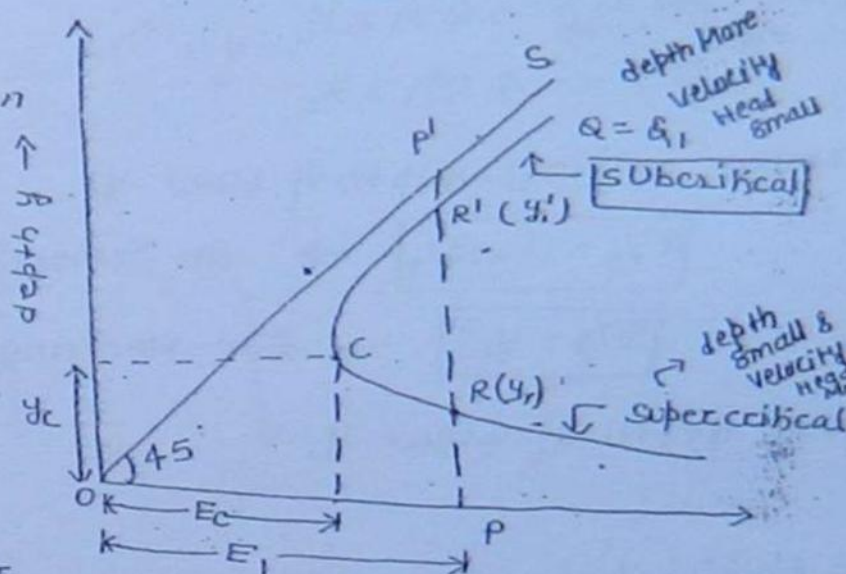
of constant discharge situation: ↓

$$E = y + \frac{Q^2}{2gA^2}$$

* For a channel of known geometry

$$E = f(y, Q)$$

Keeping $Q = \text{constant}$
 $Q = Q_1$ the variation of E with y is represented by a cubic parabola.



That at any particular discharge Q_1 can be passed, sp. energy $E = y + \frac{V^2}{2g} \rightarrow$ in a given channel at two depths & still maintain the same sp. energy E .

The depth of flow can be either $PR = y_1$ or $PR' = y'_1$. These two have same sp. energy. The intercept PR' or PR represents the velocity head depth ($PR = y_1$) is smaller and has a large velocity head while other ($PR' = y'_1$) has a larger depth and consequently a smaller velocity head.

→ For a given Q , as the sp. energy is increased the difference b/w the two alternate depths increases.
 → If E is decreased, the difference ($y'_1 - y_1$) will decrease and at a certain value $E = E_c$.

→ At the lower limb CR of the sp. energy curve the depth $y_1 < y_c$ as such $V_1 > V_c$ and $F_1 > 1$
 ↳ supercritical flow Region

→ In the upper limb CR', $y'_1 > y_c$ as such $V'_1 < V_c$ and $F'_1 < 1$ ⇒ subcritical flow Region.

$$\frac{v^2}{gy} \neq 1$$

306

iv) If $\frac{v^2}{gy} = 1$, $Fr = 1$

$$\frac{dv}{dx} = \infty \rightarrow$$

Since $m.g.v.F [dy/dx]$ is small depth of flow change over a larger length hence this condition beyond the assumption of G.V.F.

For a given discharge normal depth of flow can be calculated as follows:

Normal depth of flow:

[depth of flow at which a given discharge flows as uniform flow in a given channel]

For given values of Mannings N and chezy, C and for given value of Q and channel bottom slope S_0 there will exist one depth of flow (y_n) at which the uniform flow will be maintained. Such a depth of flow is called Normal depth of flow.

- choking: - (i) u/s, water surface elevation is not affected by the conditions section (2) till a critical stage is not achieved.

(2) in case of hump, for all $[d_2 \leq d_{2max}]$ - u/s water depth is constant
For all $[d_2 > d_{2max}] \rightarrow y_1$ increases in subcritical flow
 $\rightarrow y_2$ decreases in supercritical flow

(3) in case of width contraction:-

$[B_2 > B_{2m}] \rightarrow$ u/s depth y_1 is constant while for $[B_2 < B_{2m}] \rightarrow$ u/s depth y_1 goes to change

- onset of \rightarrow critical condition at (2) is prerequisite to choking.
- All cases $[d_2 > d_{2max}, B_2 < B_{2m}] \Rightarrow$ known as choked condition
- In subcritical flow, water surface will drop due to decrease in sp. energy
or supercritical flow, depth of flow increases due to reduction in sp. energy.

According to Manning's Eqn: 7

$$\frac{dy}{dx} = S_0 \left[\frac{1 - (y_n/y)^{10/3}}{1 - (dy/dy)^{10/3}} \right]$$

(303)

$y \rightarrow$ Actual depth of flow

$y_n \rightarrow$ It is that depth of flow at which flow is uniform

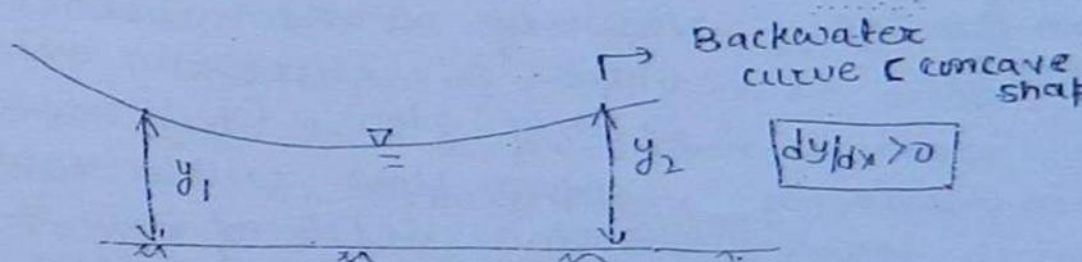
$dy/dx \rightarrow$ Rate of change of depth of flow w.r.t channel bottom

2) $\frac{dy}{dx} > 0 \rightarrow$ depth of flow is increasing in direction of flow
 \rightarrow Backwater flow curve

$\frac{dy}{dx} = 0 \rightarrow$ depth of flow is constant

$\frac{dy}{dx} < 0 \rightarrow$ depth of flow is decreasing
 \rightarrow drawdown curve.

1)



$$\frac{dy}{dx} = \frac{S_0 - S_f}{1 - \frac{V^2}{gY}} ; \quad \frac{dy}{dx} > 0 \text{ if}$$

\rightarrow channel slope > energy slope

\rightarrow Flow is subcritical.

$$\left[\begin{array}{l} S_0 > S_f \\ i > \frac{V^2}{gY} \end{array} \right]$$

2)

$$S_0 < S_f$$

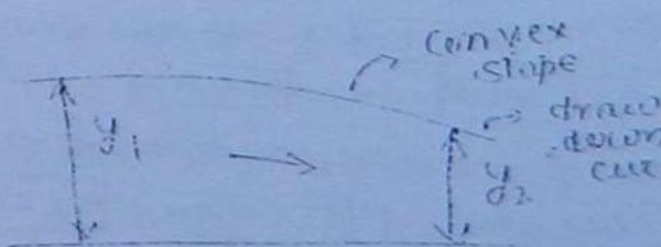
$$1 < \frac{V^2}{gY}$$

$$\frac{dy}{dx} < 0$$

\rightarrow supercritical flow

drawdown curve

\rightarrow convex shape.



$$d/dx \left[\frac{Q^2}{2gB^2y^2} \right] = \frac{Q^2}{2gB^2} \times -2/y^3 \cdot dy/dx$$

$$[A = By]$$

$$= \frac{Q^2}{gB^2y^2} \times y \cdot dy/dx$$

$$= -\frac{V^2}{gy} \cdot dy/dx$$

$$\therefore dE/dx = dz/dx + dy/dx - \frac{V^2}{gy} \cdot dy/dx$$

$$dy/dx \left[1 - \frac{V^2}{gy} \right] = dE/dx - dz/dx$$

$$\therefore dy/dx = \frac{-S_f - (-S_0)}{1 - V^2/gy}$$

$$\boxed{dy/dx = \frac{S_0 - S_f}{1 - V^2/gy}}$$

dynamic eqn for G.V.F
Rectangular section.

$$\boxed{dy/dx = \frac{S_0 - S_f}{1 - Fr^2}}$$

For Rectangular
channel

For All sections:

$$\boxed{dy/dx = \frac{S_0 - S_f}{1 - \frac{Q^2 T}{g A^3}}}$$

$$\left[Fr = \frac{V}{\sqrt{gy}} \right]$$

Energy slope

$$\boxed{S_f = h_f/L}$$

$$\boxed{S_0 - S_f = \frac{dE}{dx}}$$

↑
differential
energy eqn of G.V.F

According to chezy's eqn

$$\boxed{dy/dx = S_0 \left[\frac{1 - (y_n/y)^3}{1 - (y_c/y)^3} \right]}$$

V.Imp
(obj)

$y_c \rightarrow$ critical depth of flow
 $y_n \rightarrow$ Normal " "

Dynamic Equations Gradually Varied Flow:

Assumptions: ↓

- 1) Chezy's Formula & Manning Formula is used (with S_o as energy slope)
- ✓ Bottom slope of the channel is very small
- ✓ channel is prismatic
- ✓ Energy correction factor is 1
- ✓ Pressure distribution is only hydrostatic
- ✓ Discharge is constant, flow is steady.
- ✓ Roughness coeff. of channel is independent of depth of flow and taken constant throughout length of channel.



channel slope $\boxed{\frac{dz}{dx} = -S_o}$

If total energy is E

$$\frac{dE}{dx} = \text{Energy slope} - S_f$$

$$E = z + y + \frac{v^2}{2g}$$

$$\frac{dE}{dx} = \frac{dz}{dx} + \frac{dy}{dx} + \frac{d}{dx} \left[\frac{v^2}{2g} \right]$$

$$\therefore \frac{d}{dx} \left(\frac{v^2}{2g} \right) = \frac{d}{dx} \left[\frac{Q^2}{2gA^3} \right]$$

$$\frac{V}{\sqrt{gD}} = 1$$

∴ For Rect section $D = A/T = \frac{By}{B} = y$

$$V_c = \sqrt{gy_c}$$

For Triangular section

$$D = A/T = y/2$$

$$V_c = \sqrt{gy_{c/2}}$$

For Parabolic section

$$D = A/T = \frac{2}{3}y$$

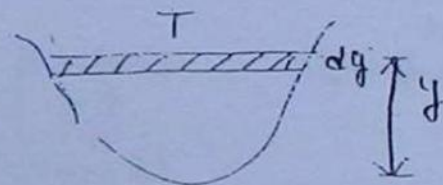
$$V_c = \sqrt{\frac{2gy_c}{3}}$$

critical depth for Non-uniform channel: ↓

$$dA = T dy$$

$$\therefore dA/dy = T$$

$$E = y + \frac{Q^2}{2gA^2}$$



For ~~critical~~ critical Flow $dE/dy = 0 = 1 + \frac{Q^2}{2g} \left(-\frac{2}{A^3} \right) \frac{dA}{dy}$

$$\therefore \frac{Q^2}{g} = \frac{A^3}{(dA/dy)} = A^3/T$$

$$\frac{Q^2}{g} = A^3/T$$

→ For critical Flow
Applicable for Rectangular or
Non-uniform section.

gpt

$$y_c = \frac{2}{3} E_{min} \quad \text{For Rectangular}$$

$$= \frac{3}{4} E_{min} \quad \text{For Parabolic}$$

$$= \frac{4}{5} E_{min} \quad \text{For Triangular}$$

→ At critical flow it may be proved that discharge is Maxm. For a Rectangular section.

$$E = y + \frac{q^2}{2g} \frac{1}{y^3}$$

(301)

$$\therefore \text{For } E_{min}, \quad dE/dy = 0$$

$$\Rightarrow 1 + \frac{q^2}{2g} \left(-\frac{3}{y^4}\right) = 0$$

$$\Rightarrow y^3 = \left(\frac{q^2}{g}\right)$$

$$\therefore \text{At } E_{min}, \quad y = y_c$$

$$\therefore \boxed{y_c^3 = \left(\frac{q^2}{g}\right)} \rightarrow \text{Valid For Rectangular section.}$$

$$\text{If } y = y_c, \quad E = E_{min}$$

$$\therefore \text{at } y = y_c, \quad E_{min} = y_c + \frac{y_c^3}{2} \times \frac{1}{y_c^3}$$

$$\therefore \boxed{E_{min} = \frac{3}{2} y_c}$$

At critical flow, for Rectangular section Kinetic head is half of potential head.

* For parabolic section ↓

$$E_{min} = \frac{4}{3} y_c$$

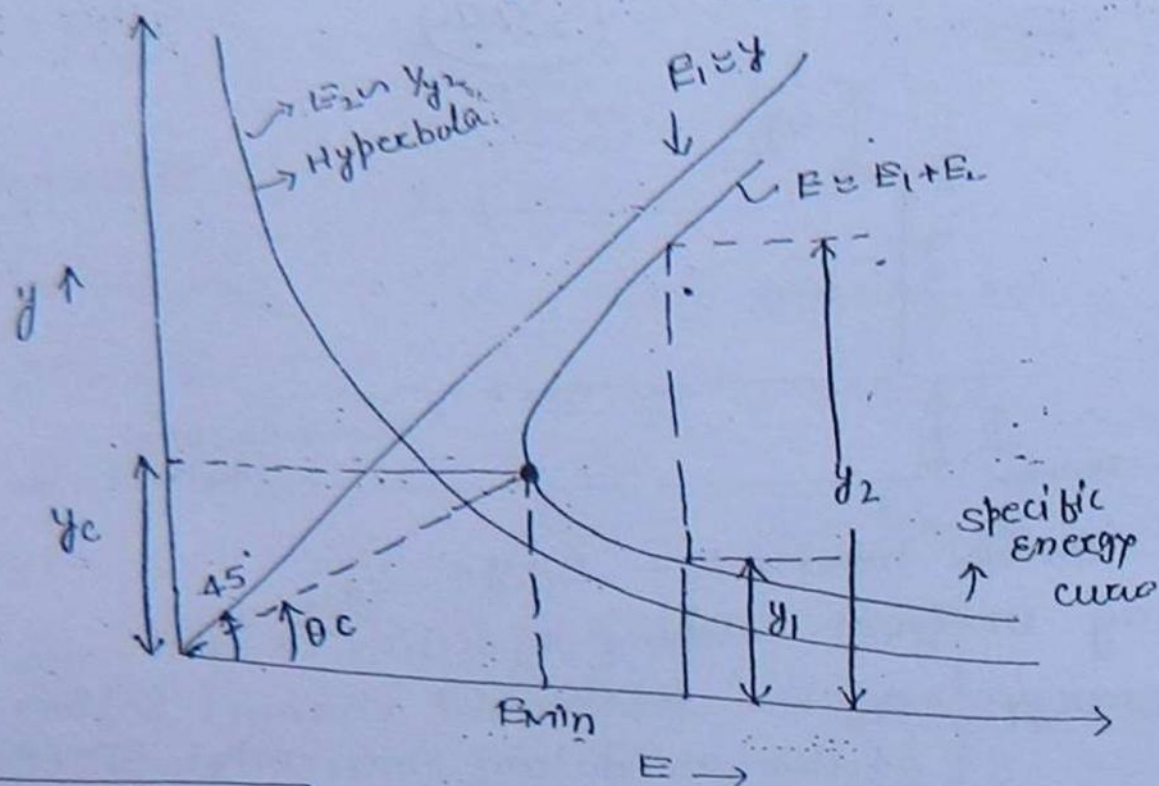
$$= y_c + \frac{y_c}{3} \rightarrow \text{Kinetic head}$$

* For Triangular section ↓

$$\boxed{E_{min} = \frac{5}{4} y_c = \left(y_c + \frac{y_c}{4}\right)} \rightarrow \text{Kinetic head.}$$

$$E_2 = E_K = \frac{v^2}{2g} + y_2$$

300



$$\tan \theta = \frac{y_c}{E_{min}} = \frac{8/9 E_{min}}{E_{min}}$$

$$\Rightarrow \theta_c = 33.7^\circ \quad \text{For Rectangular} \\ = 38.65^\circ \quad \text{For Triangular} \\ = 36.8^\circ \quad \text{For Parabolic}$$

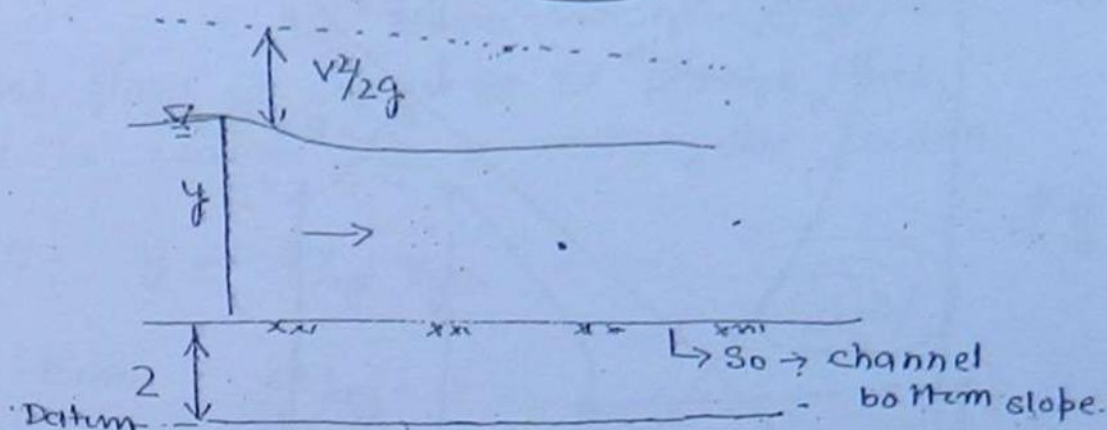
For G.V.F. Minimum specific energy occurs at which there is only one depth of flow called critical depth of flow (y_c). It means at critical flow specific energy is minimum.

For other sp. energy there will be two depth of flow y_1 & y_2 known as alternate depths. For Rectangular section y_c is critical depth $2/3$ of E_{min} .

NON-UNIFORM FLOW (For channel)

G.V.F. ↓

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$$\text{Total energy head} = E = z + y + \frac{V^2}{2g}$$

Assuming uniform velocity of section $a=1$

Specific Energy: If the ~~channel~~ channel bottom is taken as datum then total energy per unit wt. is called specific energy.

For G.V.F. specific energy is constant

$$E = y + \frac{V^2}{2g}$$

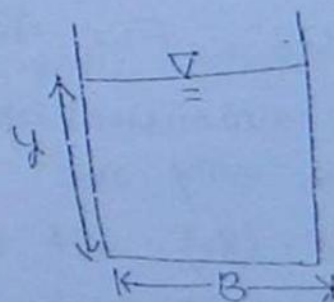
Case: 1st: For Rectangular section

Specific Energy

$$E = E_p + E_k$$

$$E = y + \frac{V^2}{2g}$$

$$= y + \frac{Q^2}{2gA^2}$$



Let discharge per unit width $q = Q/B$

$$E = y + \frac{q^2}{2gy^2} = y + \left(\frac{q^2}{2g}\right) \times \frac{1}{y^2}$$

mp

$$R_{\max} = 0.60R$$

$$= 0.80D$$

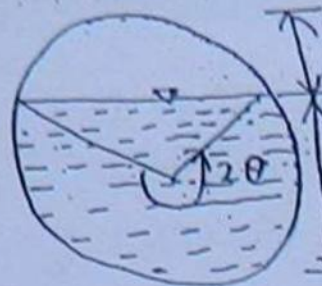
298

For V_{\max} :

$$2\theta = 257^{\circ} 30'$$

$$y = 0.81D$$

$$R_{\max} = 0.3D$$



Condition For Max^m discharge:

A) using chezy's eqn:

$$Q = CA R Y_2 S Y_2$$

For $Q_{\max} \Rightarrow (AR)^{1/2}_{\max}$

$$\therefore \frac{d}{d\theta} [AR Y_2] = 0$$

$$\Rightarrow \frac{d}{d\theta} \left[\frac{A^{3/2}}{P Y_2} \right] = 0$$

$$\Rightarrow \begin{cases} 2\theta = 308^{\circ} \\ y = 0.95D \\ R = 0.29D \end{cases}$$

B) using Manning eqn:

$$Q = Y_N \cdot A R^{2/3} S Y_2$$

$$Q_{\max} = (AR^{2/3})_{\max}$$

$$\therefore \begin{cases} 2\theta = 302^{\circ} 20' \\ y = 0.938D \\ R = 0.29D \end{cases}$$

COMMENT: →

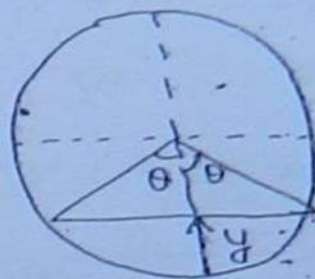
From technical consideration Manning result are more realistic because Mannings N based on surface grain roughness which can be computed directly whereas chezy's C is given arbitrary.

$$R = \frac{y}{2\sqrt{2}}$$

CIRCULAR SECTION: ↓

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- 1) For Maxm. velocity condition
- 2) For Maxm. discharge condition



$$V \propto (R = A/P)^{1/2} \quad \left| \quad A \propto R^2 \theta \right.$$

$\propto R^{2/3}$ ↳ Manning's eqn

$\propto A \cdot R^{1/2}$ Chezy's eqn.

For V_{max} - R should be Maxm.

$$A = R^2 \left[\theta - \frac{\sin 2\theta}{2} \right]$$

$$P = 2\theta R$$

$$\frac{dR}{d\theta} = 0 \Rightarrow \frac{d}{d\theta} (A/P) = 0$$

$$\downarrow$$

$$= \frac{P \cdot \left(\frac{dA}{d\theta} \right) - A \cdot \frac{dP}{d\theta}}{P^2} = 0$$

$$\left[\begin{array}{l} \frac{dA}{d\theta} = R^2 [1 - \cos 2\theta] \\ \frac{dP}{d\theta} = 2R \end{array} \right.$$

$$\Rightarrow \frac{P R^2 [1 - \cos 2\theta] - A (2R)}{P^2} = 0$$

$$\Rightarrow 2R\theta [1 - \cos 2\theta] R^2 - R^2 \left[\theta - \frac{\sin 2\theta}{2} \right] \times 2R = 0$$

$$\Rightarrow \boxed{2\theta = \sin 2\theta}$$

$$\tan \theta = \theta \quad \rightarrow \text{Trial and Error}$$

$$\theta = 4.5 \text{ rad}$$

$$\boxed{2\theta = 4.5 \text{ rad} = 257^\circ 30'}$$

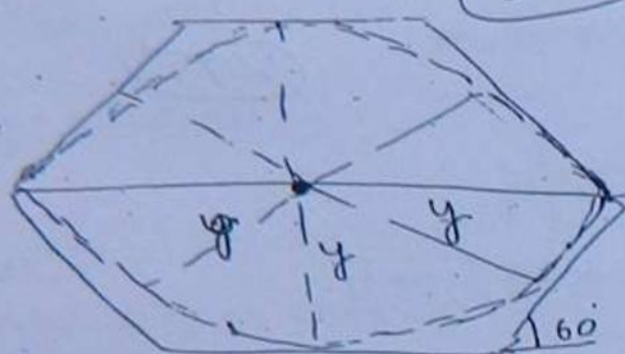
$$= y/AC = \frac{y}{y\sqrt{n^2+1}} = \frac{OB}{\left(\frac{B+2ny}{2}\right)}$$

$$\therefore \boxed{OB = y}$$

condition:

For an economical trapezoidal section side slope will be at $nH:1V$

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$$\boxed{\theta = 60^\circ}$$

channel is most economic

$$\boxed{1H:3V}$$

Trapezoidal section is a part of Hexagon whose centre is at middle of top width.

Triangular section: \downarrow

$$A = ny^2$$

$$P = 2y\sqrt{1+n^2}$$

$$R = A/P$$

\therefore For maxm B. Pmin

$$\therefore \frac{dP}{dn} = 0$$

$$P = 2 \times \frac{\sqrt{A}}{\sqrt{n}} \sqrt{n^2+1}$$

$$P = 2\sqrt{A}\sqrt{1+\frac{1}{n}} \Rightarrow P^2 = 4A(n+1/n)$$

$$2P\left(\frac{dP}{dn}\right) = 4A(1-\frac{1}{n^2})$$

$$\boxed{\frac{dP}{dn} = 2\sqrt{A}}$$

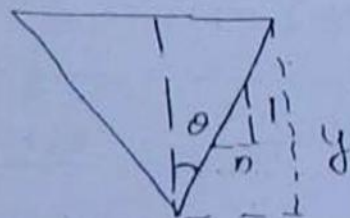
$$\frac{dP}{dn} = 0, n=1 \Rightarrow \boxed{\theta = 45^\circ}$$

\therefore Triangular section to be most economical

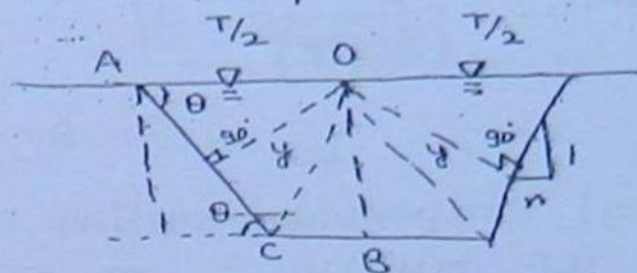
$$\theta = 45^\circ$$

and

$$\boxed{R = \frac{y}{2\sqrt{2}}}$$



case I: \rightarrow side slopes are constant



$$A = (B + ny)y$$

$$P = B + 2y \sqrt{1 + n^2}$$

$$T = B + 2ny$$

For Maxm Q at given area

P should be Minimum

$$\therefore \frac{dP}{dy} = 0$$

$$P = B + 2y \sqrt{n^2 + 1}$$

$$= \frac{A}{y} - ny + 2y \sqrt{n^2 + 1}$$

$$\therefore \frac{dP}{dy} = -\frac{A}{y^2} - n + 2 \sqrt{n^2 + 1}$$

$$= -\left(\frac{(B + ny)y}{y^2}\right) - n + 2 \sqrt{n^2 + 1} = 0$$

$$= -\left(\frac{B + ny}{y}\right) - n + 2 \sqrt{n^2 + 1} = 0$$

$$y \sqrt{n^2 + 1} = \frac{B + 2ny}{2}$$

$$T/2 = \frac{B + 2ny}{2}$$

$$\text{Side} = y \sqrt{n^2 + 1}$$

\therefore For Most Economical channel

$$\frac{1}{2} \text{ top width} = \text{one of sloping side length}$$

$$R = \frac{A}{P} = \frac{(B + ny)y}{B + 2y \sqrt{n^2 + 1}}$$

$$= \frac{(B + ny)y}{B + 2 \left(\frac{B + 2ny}{2}\right)}$$

$$\Rightarrow R = y$$

Best economical section. ↓

A section of a channel is said to be economical when its cost of construction is Least or for a given discharge and given area.

For a given sectional Area, dimension of section designed in such a way that discharge carrying capacity is maximum.

Rectangular section: ↓

$$A = By = \text{constant}$$

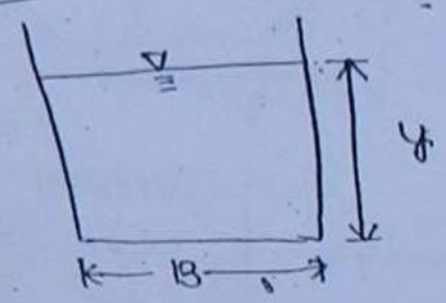
For Maxm Q

we know that

$$Q \propto V \propto R^{2/3} \rightarrow \text{Chezy's}$$

$$\propto R^{5/3} \rightarrow \text{Manning}$$

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$Q_{\text{max}} \rightarrow R$ should be Maxm

For a given area $R = A/P$

For $R_{\text{max}} \rightarrow P$ should be ~~Maximum~~ Minimum

wetted perimeter $P = B + 2y$

$$P = \frac{A}{y} + 2y$$

For $P_{\text{min}} \quad \frac{dP}{dy} = 0$

$$-\frac{A}{y^2} + 2 = 0$$

$$\Rightarrow \boxed{y = \frac{B}{2}} \quad \text{Ans}$$

$$\boxed{\text{depth of flow} = \frac{1}{2} \text{ width}} \quad \text{Ans}$$

Hydraulic Radius $R = \frac{A}{P} = \frac{By}{B+2y} = \frac{B \cdot B/2}{B+2 \cdot B/2}$

$$\boxed{R = \frac{B}{4} = \frac{y}{2}} \quad \text{Ans}$$

Prob. Find the discharge for the channel section shown below whose bed slope is 0.001 and Manning's $N = 0.018$

$$S = 0.001$$

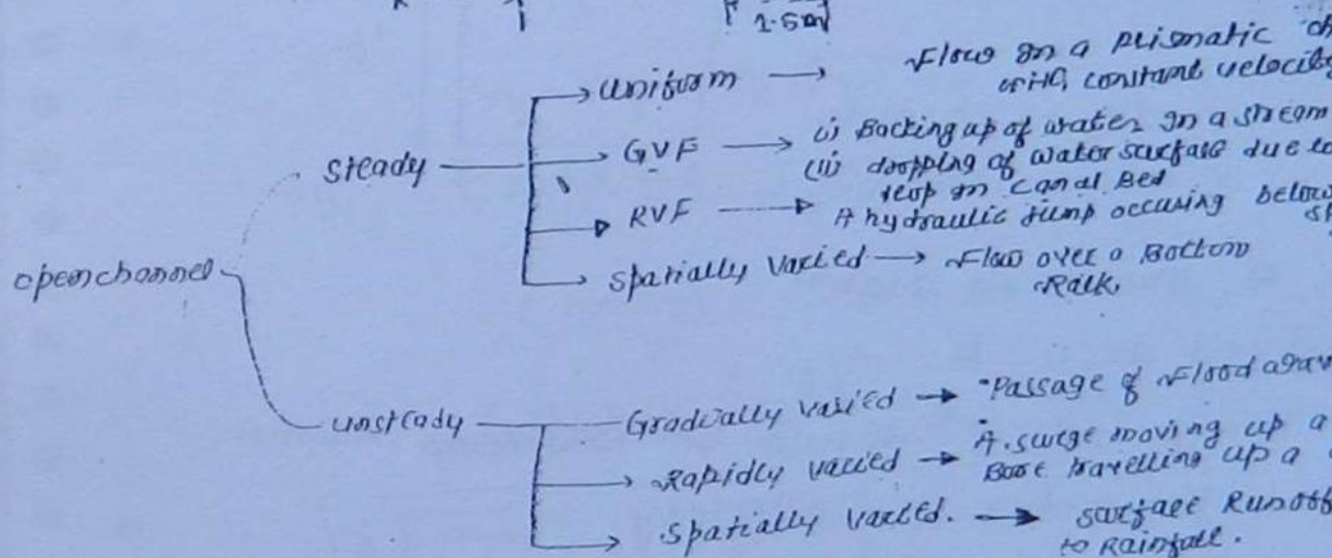
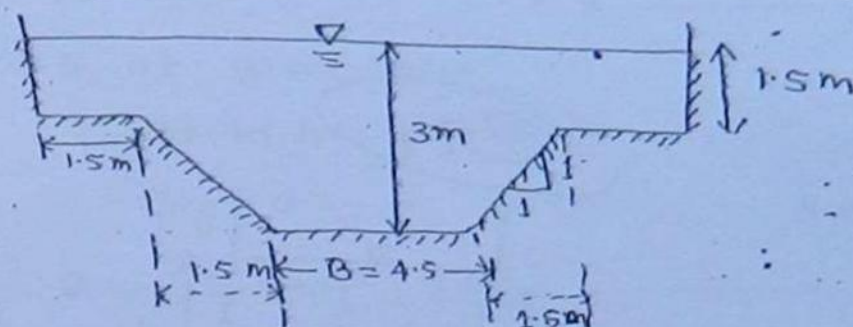
$$N = 0.018$$

$$0.012 \text{ to } 0.025$$

FOR Smooth Lined channels.

FOR Very Rough Earthen channels.

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- * In G.V.F, Frictional resistance plays an important role.
- * In G.V.F, R.V.F No flow is externally added or taken out of the system.
- spatially varied flow - either some flow is added or subtracted from the system.

→ Specific Force is sum of the Pressure Force + Momentum Force per unit wt. of the fluid at a section.

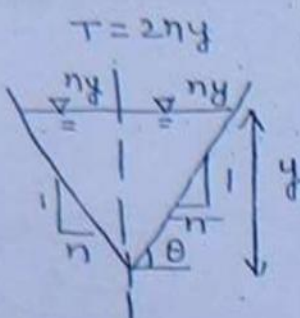
Sp. Force is constant in a horizontal, frictionless channel.

critical flow condition is governed by the channel geometry and discharge. other channel properties such as the bed slope and roughness do not influence the critical flow condition for any given discharge.

$$E_c = 5/4 y_c$$

$$f_c = \left(\frac{2.5^2}{9m^2} \right)^{1/5}$$

$$F = \sqrt{2} - \frac{V}{\sqrt{gy}}$$



sideslopes

$$\boxed{nh:1V}$$

$$\tan \theta = n$$

$$A = ny^2$$

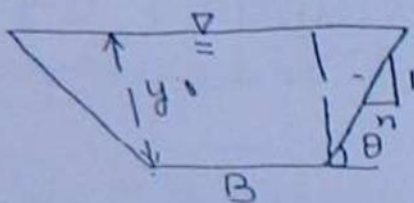
$$T = 2ny$$

$$R = A/P = \frac{ny}{2\sqrt{1+n^2}}$$

$$P = 2y\sqrt{1+n^2}$$

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d) Trapezoidal section: ↓



$$T = B + 2ny$$

$$A = \left(\frac{B+T}{2} \right) y = \left(\frac{B+B+2ny}{2} \right) y$$

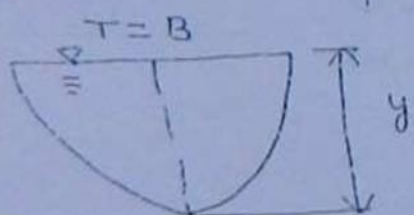
$$\boxed{A = (B+ny)y} \checkmark$$

wetted perimeter $\boxed{P = B + 2y\sqrt{1+n^2}}$

e) Parabolic channel: ↓

hydraulic depth

$$\boxed{D = A/T = 2/3 y}$$



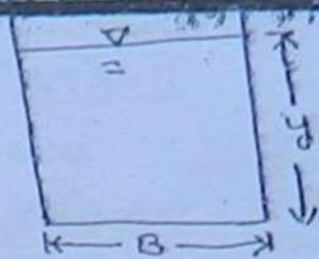
section factor

$$Z = 2/9 \sqrt{6} B y^{3/2}$$

$$A = 2/3 T y = 2/3 B y$$

wetted perimeter $P = B + 8/3 \frac{y^2}{B}$
 $= 3B^2 + 8y^2$

A) Rectangular section:



(29)

$$A = By$$

$$\text{Wetted Perimeter} = B + 2y$$

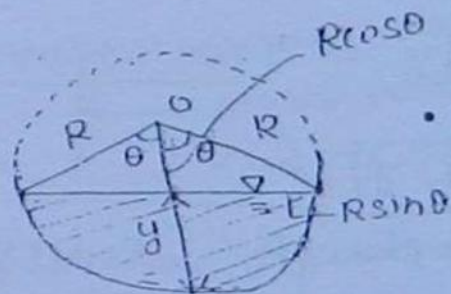
$$R = A/P = \frac{By}{B + 2y}$$

$$D = Hy \cdot \text{depth} = A/H = y$$

$$\text{Section Factor} = \sqrt{A^3/I}$$

$$= \left[\frac{(By)^3}{B} \right]^{1/2} = B \cdot y^{3/2}$$

B) Circular section: ↓



• Area of Flow

$$= \theta R^2 - 2 \int_0^{\theta} R^2 \cos \theta d\theta$$

$$= \theta R^2 - 2 R^2 \sin \theta$$

$$= R^2 \left[\theta - 2 \sin \theta \right]$$

$$= R^2 \left[\theta - \frac{1}{2} \sin 2\theta \right]$$

$$A = R^2 \left[\theta - \frac{1}{2} \sin 2\theta \right]$$

$$\bullet \text{ Top width } T = 2R \sin \theta \checkmark$$

$$\bullet \text{ Wetted Perimeter } P = 2\theta R$$

$$C = \frac{23 + \frac{0.00155}{S} + Y_n}{\left[1 + \left(23 + \frac{0.00155}{S} \right) \right] \frac{n}{\sqrt{R}}}$$

Bazin's eqn: ↓

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$$C = \frac{157.6}{1.81 + K/\sqrt{R}}$$

• K = Bazin's coeff

$$C = \frac{m/sec}{\sqrt{m}} = L^{1/2} T^{-1/2}$$

$$C = \frac{148}{1 + M/R}$$

Average or Mean velocity is find out

Manning's Equation: ↓

$$V = Y_N R^{2/3} S^{1/2}$$

N = Manning's
Rugosity coeff

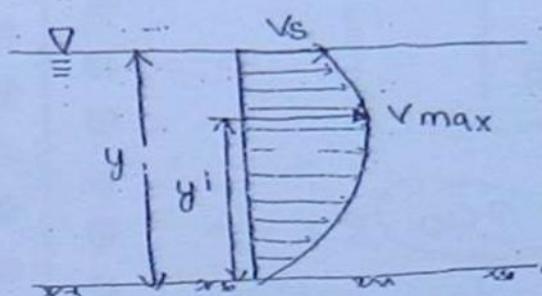
$$N = L^{1/3} T^{-1}$$

$$C = Y_N R^{1/6}$$

v. Imp.

$$C = \sqrt{8g/f} = Y_N R^{1/6}$$

Velocity distribution in open channel:



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y_1 = distance of V_{max} $\approx 0.8y$ to $0.95y$ ^{gib}
from bottom

$V_s \approx 0.91 V_{max}$ ^{gib} \rightarrow (May be b/w 0.85 to $0.95 V_{max}$)
 \rightarrow velocity at surface

$$V_{mean} = Q/A$$

$V_{mean} \approx$ velocity at $0.6y$ from bottom to (Free surface)

$$\approx \frac{V_{at 0.2y} + V_{at 0.8y}}{2}$$

UNIFORM FLOW:

Methods to determine velocity & discharge

(a) chezy's eqn: \downarrow

$$V = C \sqrt{RS}$$

$$Q = A C \sqrt{RS}$$

chezy's constant can be calculated by the Kutter's or Bazin's eqn and C depends upon the surface roughness

S = slope of the channel bottom
 A = Area of flow
 R = Hydraulic Radius [Effective length parameter]

C = chezy's constant
 $C = 47.55 \sqrt{S}$ ^{gib}

$$C = \frac{23 + \frac{0.00155}{S} + V_n}{\left[1 + \left(23 + \frac{0.00155}{S}\right) \frac{n}{V_n}\right]}$$

n = Kutter's roughness coefficient
= Roughness of channel surface

Subcritical Flow:

if $FR < 1$

Subcritical / Tranquil / Streaming / stable flow
(velocity Low, depth of flow high)

critical Flow: \downarrow

if $FR = 1 \Rightarrow$ critical Flow

supercritical Flow: \downarrow

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if $FR > 1 \Rightarrow$ supercritical Flow / shock

Rapid / unstable flow

(\uparrow velocity, depth of flow Low)

$$FR = \frac{V}{\sqrt{gD}}$$

$V \rightarrow$ Mean velocity = Q/A

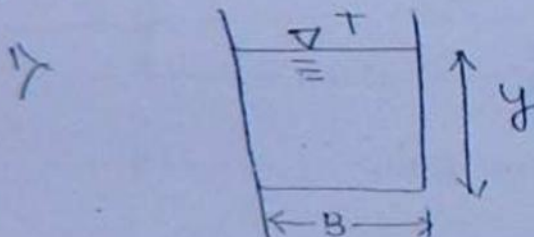
$D \rightarrow$ Hydraulic depth

$$D = A/T$$

$T =$ top width

$A =$ Area of flow

For eg: \downarrow



\therefore Hyd. depth = A/T

= $By/B = y$

= depth of flow { in case of Rect. channel }

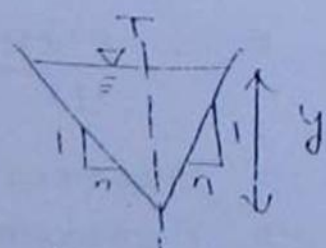
2)

$T = 2ny$

$A = \frac{1}{2} \times y \times 2ny = ny^2$

\therefore Hydraulic depth = A/T

= $y/2$



[side slope = n]

IDEAL CHANNEL FLOW

O.C.F. $\left\{ \begin{array}{l} \text{Uniform} \rightarrow V = \text{const}, \bar{d} = \text{const}, A = \text{const} \left[\begin{array}{l} \text{chan} \\ \text{is} \\ \text{pris} \end{array} \right] \\ \text{Non-uniform} \left\{ \begin{array}{l} \text{GVF} \\ \text{RVF} \end{array} \right. \end{array} \right. \quad Q = \text{const}$

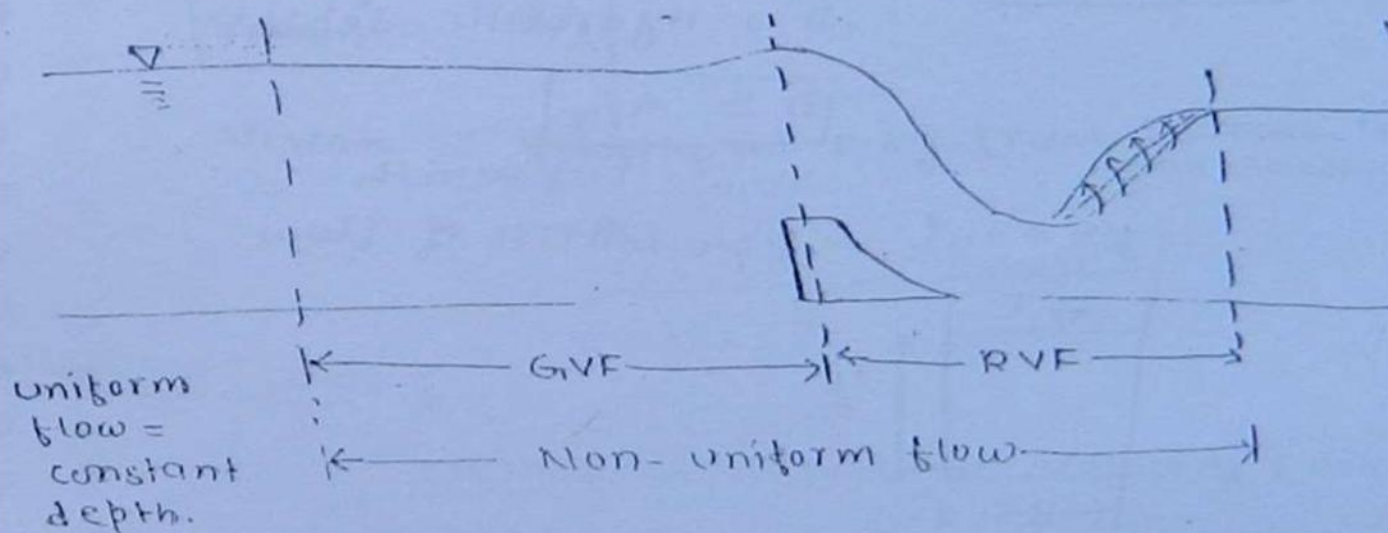
(287)

GVF \rightarrow depth of flow changes over a long distance

\rightarrow No Loss of energy if friction Loss are Negligible.

RVF \rightarrow depth of flow changes suddenly & dissipation

Energy ~~dissipation~~ takes place at point of Jump formation.



* For Laminar Flow in open channel

$Re \leq 500 \rightarrow$ Laminar

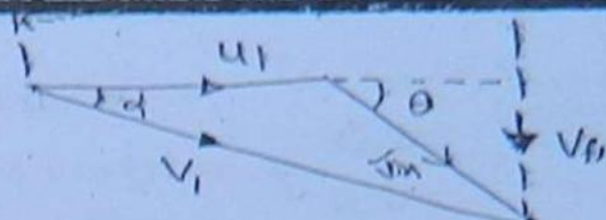
$Re > 2000 \rightarrow$ Turbulent

$$Re = \frac{RVR}{\mu} = \frac{VR}{\nu}$$

$R \rightarrow$ Hydraulic Radius $= A/p \rightarrow$ wetted perimeter

\rightarrow Hydraulic Mean

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(285)

$$V_{f1} = V_{f2} = V_f$$

$$V_{w2} = 0$$

$$\boxed{\frac{V_{w1}}{V_{f1}} = \cot \alpha}$$

$$\therefore V_{w1} = V_{f1} \cot \alpha$$

$$\boxed{V_2 = V_{f1}}$$

$$\therefore V_1^2 = V_{f1}^2 + V_{w1}^2$$

$$\therefore V_1 = \sqrt{1 + \cot^2 \alpha} \cdot V_{f1}$$

$$\therefore \boxed{V_1 = V_{f1} \sqrt{1 + \cot^2 \alpha} = V_{f1} \operatorname{cosec} \alpha} \quad \text{--- (i)}$$

$$\therefore P = 1 - \frac{V_{f1}^2 - V_{f1}^2 \operatorname{cosec}^2 \alpha}{2 V_{f1} \cot \alpha \times V_{f1} (\cot \alpha - \cot \theta)}$$

$$P = 1 + \frac{V_{f1}^2 (1 - \operatorname{cosec}^2 \alpha)}{2 V_{f1}^2 \cot \alpha (\cot \alpha - \cot \theta)}$$

$$= 1 - \frac{\cot^2 \alpha}{2 \cot \alpha (\cot \alpha - \cot \theta)}$$

$$= 1 - \frac{\cot \alpha}{2 (\cot \alpha - \cot \theta)}$$

$$\boxed{P = \frac{1}{2} \left[1 - \frac{\cot \theta}{(\cot \alpha - \cot \theta)} \right]}$$

velocity of flow from inlet to exit remains constant
 at the turbine discharges radially so that the degree
 of reaction (f) can be expressed as

$$f = \frac{1}{2} \left[1 - \frac{\cot \theta}{\cot \alpha - \cot \theta} \right]$$

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where θ is the runner vane angle at inlet
 α is the guide blade angle &

$f \rightarrow$ degree of rxn. defined as ratio of
 pressure head dropped to the

hydraulic work done in the runner

Assume that losses in the runner are negligible

$$f = \frac{\text{Pressure head drop b/w inlet \& outlet of Runner}}{\text{work done by the water on the runner / sec. / unit of water}}$$

$$= \frac{P_1/\omega - P_2/\omega}{\left(\frac{V_{w1}u_1 - V_{w2}u_2}{g} \right)}$$

Apply the B.Eqn b/w inlet & exit of runner

$$P_1/\omega + Z_1 + \frac{V_1^2}{2g} = P_2/\omega + \frac{V_2^2}{2g} + Z_2 + \frac{V_{w1}u_1}{g}$$

$$(P_1/\omega - P_2/\omega) = \left(\frac{V_2^2}{2g} - \frac{V_1^2}{2g} \right) + \frac{V_{w1}u_1}{g} \quad \left[\begin{array}{l} \text{No losses} \\ \text{are} \\ \text{considered} \end{array} \right]$$

$$f = \frac{\left[\frac{V_2^2}{2g} - \frac{V_1^2}{2g} \right] + \frac{V_{w1}u_1}{g}}{\left(\frac{V_{w1}u_1}{g} \right)}$$

$$f = 1 + \left(\frac{V_2^2 - V_1^2}{2 V_{w1}u_1} \right)$$

Prob 1
CS/2007
ES/2001

A C.P. has an impeller of 0.5 m of outer dia. when running at 600 r.p.m. discharges 8000 L against a head of 8.5 m.

(253)

Prob 2

The cylinder bore dia. of a single acting reciprocating pump is 150 mm and stroke is 300 mm. The pump runs at 50 r.p.m. and water is lifted to a height of 25 m. The length of delivery pipe is $L_d = 22$ m and $d_d = 100$ mm.

Find the theoretical discharge & power required to running the pump, if actual discharge = 4.2 l/s. Find the slip. Also determine accel head at the beginning and middle of stroke.

$$A = \pi/4 D^2$$

$$D = 150 \text{ mm}$$

$$L = 2r = 300 \text{ mm}$$

$$N = 50$$

$$h_s = 25$$

$$L_d = 22$$

$$Q_{th} = \frac{A L N}{60} = \frac{\pi/4 (0.15)^2 \times 0.3 \times 50}{60} \text{ (m}^3/\text{s)}$$

$$= 4.42 \times 10^{-3} \text{ m}^3/\text{sec}$$

$$= 4.42 \text{ l/sec}$$

$$Q_a = 4.2 \text{ l/sec}$$

$$\therefore C_d = 4.2 / 4.42 = 0.95$$

$$\therefore \% \text{ slip} = \frac{4.42 - 4.2}{4.42} \times 100 = 4.95\%$$

Accel head = pressure head due to accel

Accel head in suction pipe \downarrow

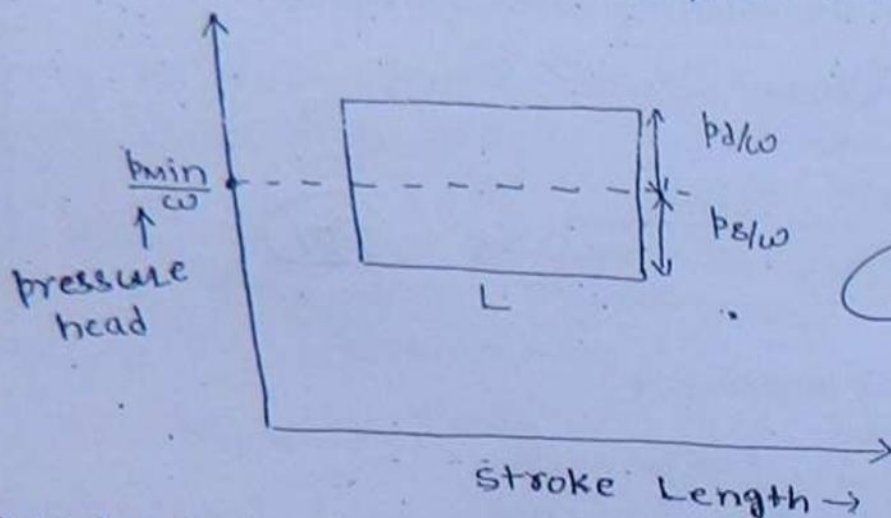
$$h_{as} = (L_s/g) \times A/a_s \times \omega^2 \cos \theta$$

$$\left[\omega = \frac{2\pi N}{60} \Rightarrow r = \frac{L}{2} = 150 \right]$$

Accel head in delivery pipe \downarrow

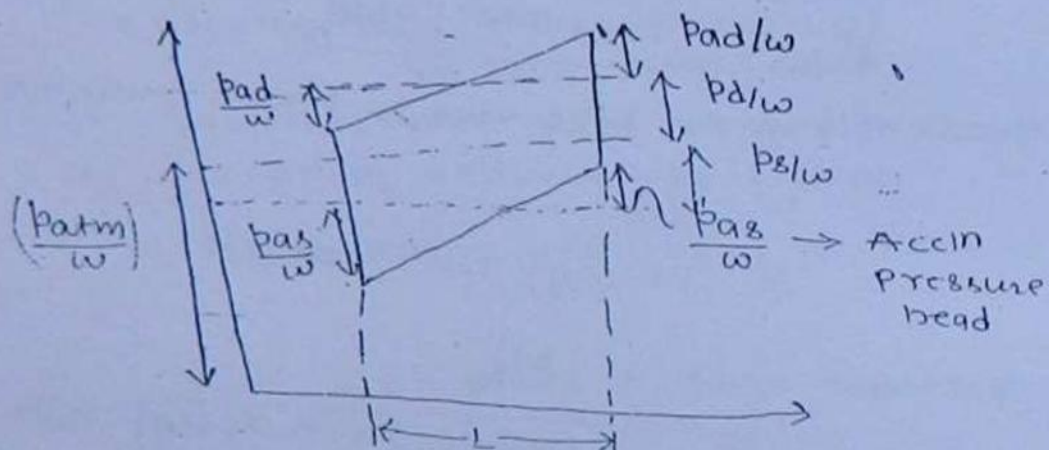
$$h_{ad} = \frac{L_d}{g} \times \frac{A}{a_d} \times \omega^2 \cos \theta \rightarrow [\theta = 180^\circ \& 270^\circ]$$

distance travelled by the piston
in complete revolution of crank.



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Ideal Indicator diagram when P_f is neglected & effect of accn. also neglected.



significance: →

1) Area of Indicator diagram is directly proportion to discharge and $Q \propto \text{Power}$. Hence if a Indicator diagram of a pump is given then discharge output and power consumed can be compared of two pumps.

10. (A/a) $\sin \theta$ \rightarrow velocity of water

$$\frac{dv}{dt} = \text{acceleration of water in pipe}$$

$$= (A/a) \times r\omega \cos \theta \left(\frac{d\theta}{dt} \right)$$

$$\therefore a = \text{Accn. in pipe}$$

$$a = (A/a) r\omega^2 \cos \theta \quad \text{301p}$$

(281)

a_{max} at $\theta = 0$ & π
required

• Force in suction water / pipe water

$$= \text{Mass} \times \text{accn}$$

$$= (\rho a l) \times (A/a) r\omega^2 \cos \theta$$

$$F = \rho l A r\omega^2 \cos \theta$$

• Pressure head in pipe due to piston movement

$$= F/a$$

$$= \rho l (A/a) r\omega^2 \cos \theta$$

$$\therefore \text{pressure head} = h/\rho g$$

$$h = (l/g) \times (A/a) \times r\omega^2 \cos \theta \quad \text{301p}$$

• Friction head on suction / delivery pipe

$$h_{fs} = \frac{f l_s v^2}{2g d_s}$$

suction
[d \rightarrow dia of pipe]

$$= \frac{f l_s \left[\left(\frac{A}{a} \right) r\omega \sin \theta \right]^2}{2g d_s}$$

$$(h_f)_{\text{max}} = \frac{f l_s \left[\left(\frac{A}{a} \right) r\omega \right]^2}{2g d_s}$$

Power Required:

$$P \propto Q$$

∴ Power required get doubled

It may be noted that though the operating cost is doubled but increased in installation cost is marginal

Slip: → (i)

$$= \frac{Q_{\text{theoretical}} - Q_{\text{actual}}}{Q_{\text{theoretical}}} \times 100$$

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$$= (1 - C_d) \times 100$$

[C_d : coefficient of discharge]

-ve slip: →

when $Q_{\text{actual}} > Q_{\text{theoretical}}$

If some of the water coming from suction pipe directly enters into delivery pipe without entering into piston chamber -
-ve slip occurs when

- (i) piston is moving at very high speed
- (ii) Length of delivery pipe is small

Effect of Accn of piston on the velocity of suction and delivery pipe: ↓

$$x = r(1 - \cos \theta)$$

∴ velocity of piston

$$V = \frac{dx}{dt} = r \sin \theta \left(\frac{d\theta}{dt} \right) = r \omega \sin \theta$$

$$\boxed{\frac{dx}{dt} = r \omega \sin \theta} \quad \text{grip}$$

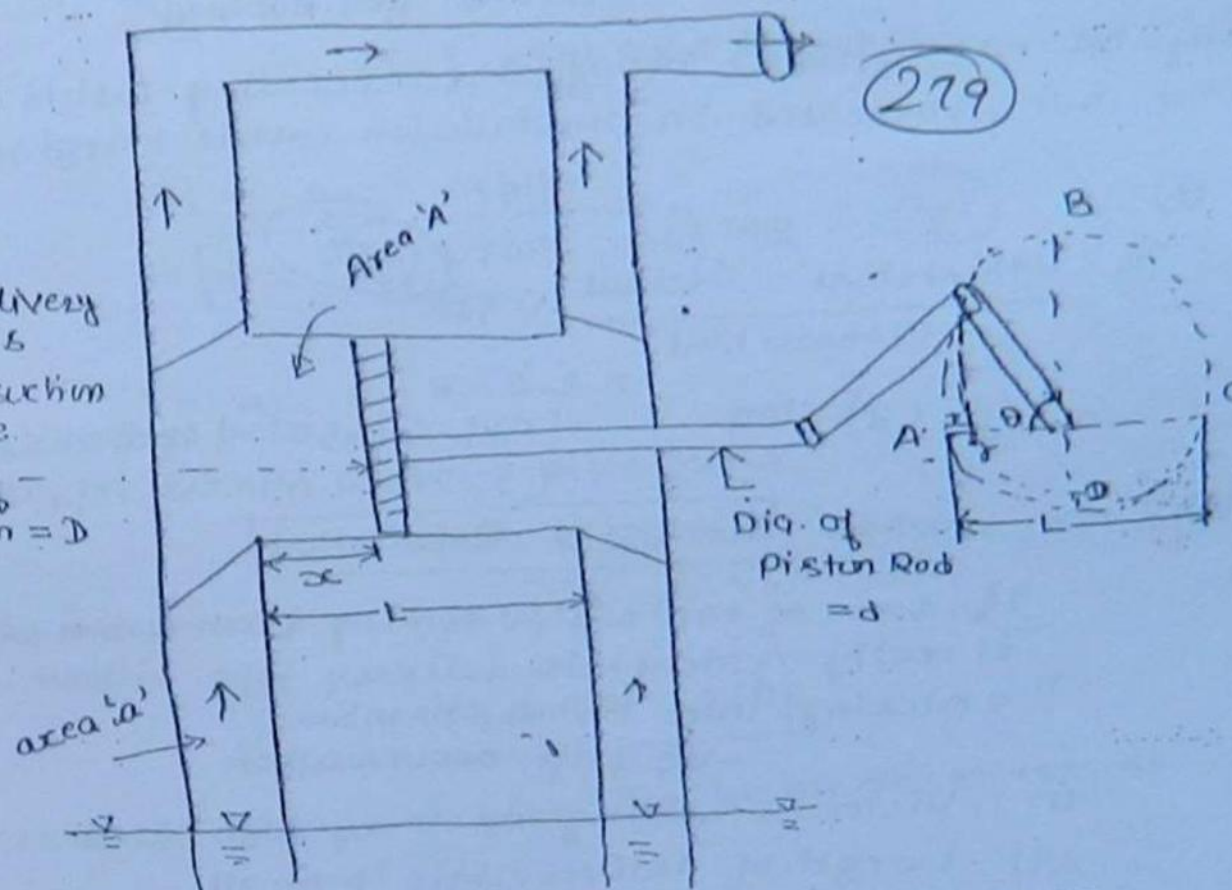
V_{max} occurs at $\theta = \pi/2$

Let v be the velocity of water in suction and delivery pipe
a be the area of that pipe

$$a v = A V \quad \text{grip}$$

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Two delivery
pipes
Two suction
pipe
Dia of
piston = D



Area on the right of the piston = $\pi/4 D^2$
Area " " Left = $\pi/4 (D^2 - d^2)$

Length $L = 2A$

So, Total volume of water delivered in one revolution of crank

$$V = \left[\pi/4 D^2 + \pi/4 (D^2 - d^2) \right] \times L$$

Volume of water delivered in one minute

$$= \left[\pi/4 D^2 + \pi/4 (D^2 - d^2) \right] L \times N$$

Discharge / sec = $\frac{\pi/4 [D^2 + (D^2 - d^2)] L N}{60}$

∴ If area, $d \ll D$ then

Q = $\frac{2 \times \pi/4 D^2 \times L N}{60} = \frac{2ALN}{60}$

$$t = 16 \text{ hr}$$

$$\therefore Q = \frac{18000}{16 \times 60 \times 60} = 0.3125 \text{ m}^3/\text{sec}$$

$$H_s = 21 \text{ m}$$

$$h_{fs} = h_{fs} + h_{fd}$$

$$h_{fs} = \frac{4fL_s Q^2}{12.1 D^5}, \quad h_{fd} = \frac{4fL_d Q^2}{12.1 D^5}$$

$$\therefore h_f = \frac{4fL_s Q^2}{12.1 D^5} + \frac{4fL_d Q^2}{12.1 D^5}$$

$$= \frac{4 \times 0.01 \times (40+150) \times (0.3125)^2}{12.1 \times (0.5)^2} = 1.96 \text{ m}$$

$$\therefore H_m = H_s + h_f$$

$$= 21 + 1.96 = 22.96 \text{ m}$$

$$\therefore \text{Manometric power} = \omega Q H_m \rightarrow \text{KW}$$

$$= \frac{\rho Q H_m}{75} \text{ (H.P.)}$$

$$= \frac{1000 \times 0.3125 \times 22.96}{75}$$

$$= 95.67 \text{ H.P.}$$

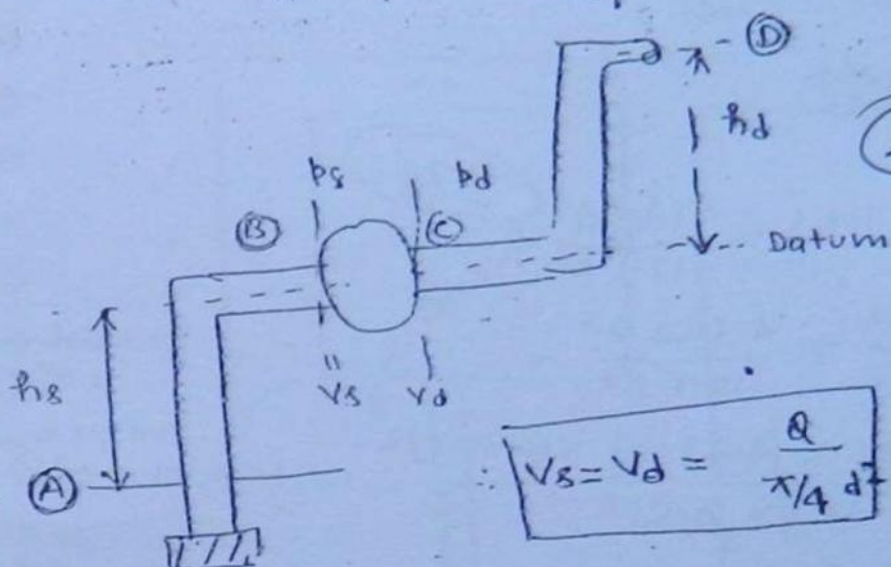
$$\eta_o = \frac{\text{Mano. Power (H.P.)}}{\text{S.H.P.}}$$

$$\text{S.H.P.} = \frac{95.67}{0.80} = 119.58 \text{ (B.H.P.)}$$

$$\text{No. of pump required} = \frac{\text{Total B.H.P.}}{\text{B.H.P. of one pump}}$$

$$= \frac{119.58}{30} \approx 4$$

hence $V_f = V_1 = Q/A_1$



$$V_s = V_d = \frac{Q}{\pi/4 d^2} = \frac{0.237}{\pi/4 (0.33)^2} = 2.46 \text{ m/sec.}$$

Apply B. Eqn b/w (A) and (B)

$$\left(\frac{p_{atm}}{\omega} \right) = \frac{p_s}{\omega} + \frac{V_s^2}{2g} + [h_s + h_{fs}]$$

Loss in suction pipe

Intend of water $\rightarrow 10.3 = \frac{p_s}{\omega} + \frac{2.46^2}{2 \times 9.81} + 3 + 2$

$$\frac{p_s}{\omega} = 5$$

$\geq 2.5 \text{ m of vapour pressure at } 20^\circ\text{C}$

Apply B. Eqn b/w (C) and (D)

$$\frac{p_d}{\omega} + \frac{V_d^2}{2g} + 0 = \frac{p_{atm}}{\omega} + \frac{V_d^2}{2g} + (h_d + h_{fd})$$

Loss in delivery pipe

$$p_d/\omega = 10.3 + 3.7 + 6 = 53.3 \text{ m}$$

Prob 4

In a pumping station 18000 m³ water is to be lifted per day from a intake well to a sedimentation tank under a static head of 21 m. Length of suction & delivery pipes are 40 m & 150 m respectively. dia. of pipes is constant = 150 mm. There are two shifts of working each of 8 h. If the efficiency of pump & motor combined is 80% and friction coeff is 0.01. Recommend the unit of pumps each having B.H.P. of (30) pipe burst

a) Discharge through pump: ↓

Let N be the s.p.m. of crank
1 ~~pipe~~ ^{one} revolⁿ of crank

Vol. of water discharged = AL

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If in crank, there are N revⁿ.

than discharge in one minute is = ALN

∴ discharge in one sec.

$$\text{gelp } Q = \frac{ALN}{60} \quad N \rightarrow \text{R.P.M}$$

$$A = \frac{\pi D^2}{4}$$

b) Power Required: ↓

$$P = \omega Q H_{\text{required}}$$

$$H_{\text{required}} = (h_s + h_d) + \underbrace{h_{fs} + h_{fd}}_{\substack{\rightarrow \text{if Neglected} \\ \text{then } H = h_s + h_d}}$$

$$\therefore P = \omega Q (h_s + h_d)$$

COMMENTS: ↓

1) The discharge through the pump is not continuous in nature therefore power required is fluctuating

2) Reciprocating pumps are suitable for high suction head & low delivery head & low discharge. More

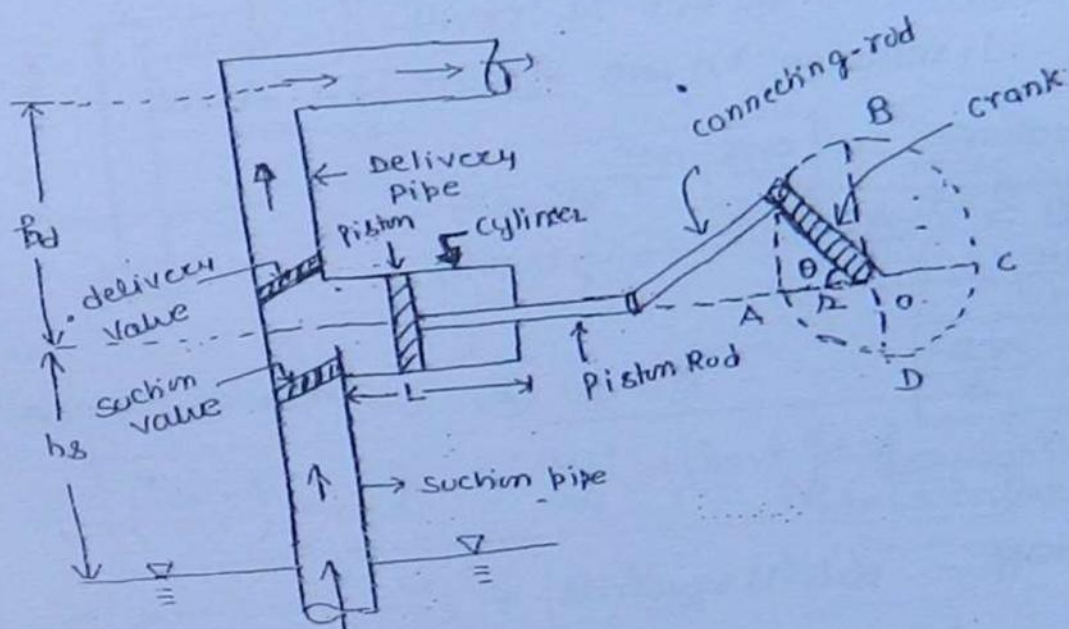
3) In order to make the continuous supply of water double acting pump may be used.

27 Reciprocating pump

These work on the principle of creating suction head.

Main parts: ↓

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Working: ↓

The crank is rotated by an external source of power. When the crank is at A, suction starts and when $\theta = 0^\circ - 180^\circ$ suction completes. The piston is moving in the cylinder forward & backward.

$A \rightarrow$ Area of cylinder = Area of piston

L = Length of cylinder chamber

$$L = 2r$$

when $\theta = 0^\circ$ to 90° , accelⁿ takes place and
when $\theta = 90^\circ$ to 180° , deaccelⁿ & at $\theta = 90^\circ$
the velocity of piston is maximum.

For delivery pipe \rightarrow for $\theta = 180^\circ$ to $270^\circ \Rightarrow$ Accelⁿ of delivery
for $\theta = 270^\circ$ to $360^\circ \Rightarrow$ Deaccelⁿ of delivery
at $\theta = 270^\circ \Rightarrow$ velocity is Max

It may be noted that while suction stroke only suction valve is open & delivery valve is closed.

when running at 600 r.p.m. discharged at the rate of 8000 Lit/min against a head of 8.5 m. The water enters the impeller without whirl & shock. The inner dia. is 0.25 m & the vanes are set back at outlet at an angle of 45° & area of flow which is constant from inlet to outlet of the impeller of 0.06 m^2 . determine

- Manometric efficiency of pump
- The Vane Angle at inlet $\rightarrow 39^\circ$
- Minimum speed at which the pump commences to work.

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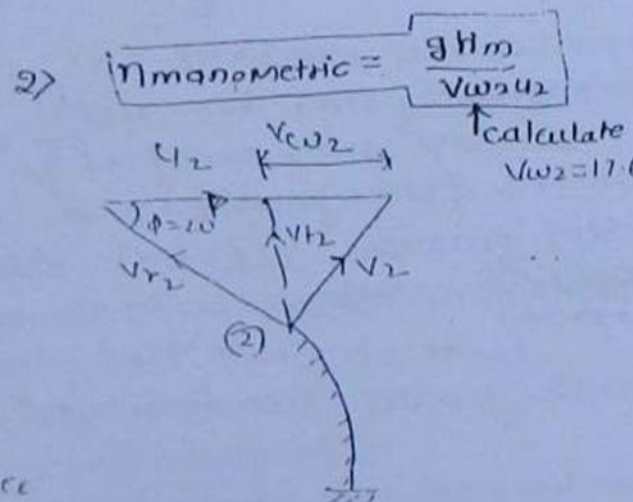
A C.P. lifts water under a static lift of 40 m of which 3 m is suction lift. The suction & delivery pipes are of 30 cm dia. both. The friction loss in the suction pipe is 2 m & in delivery pipe is 6 m. The impeller is 0.5 m dia. & 3 cm wide at outlet & runs at a speed of 1200 r.p.m. The exit blade angle is 20° and $\eta_{\text{mano}} = 85\%$. Find

- The discharge
- Pressure at the suction and at delivery point

$$1) \quad h_m = h_s + h_f + h_d + h_{fd} \\ = 40 + 2 + 6 \\ = 48 \text{ m}$$

using exit velocity diagram

$$2) \quad \tan 20^\circ = \frac{V_{f2}}{u_2 - V_{w2}} \\ \Rightarrow V_{f2} = 5.02 \text{ m/sec}$$



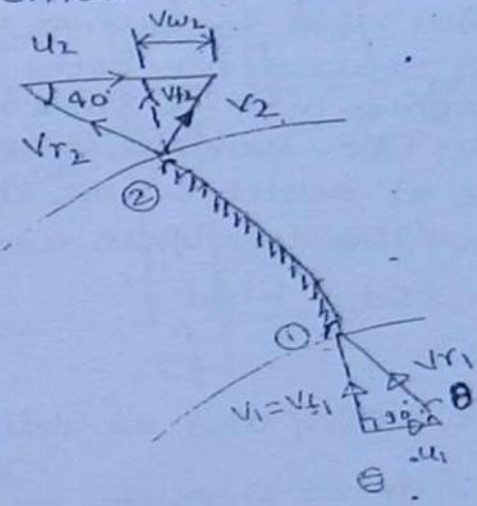
$$Q = A_{f2} \times V_{f2} = (\pi d_2 b_2) \times V_{f2} = 0.237 \text{ m}^3/\text{sec}$$

$$\text{Absolute velocity at exit } V_2 = \sqrt{V_{f2}^2 + V_{w2}^2} \\ = \sqrt{(5.02)^2 + (11.63)^2} \\ = 12.33 \text{ m/sec}$$

- A centrifugal pump having outer dia equal to two times the inner dia & running at 1000 rpm works against a total head of 40m. The velocity of flow through impeller is constant & at 2.5 m/sec. The vanes are set back at an angle of 40° at outlet. At the outer dia of impeller is 150 cm & width of outlet is 5 cm. then determine
 a) Vane Angle at inlet
 b) work done by impeller/sec in water
 c) Manometric efficiency

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→ water enters into pump without whirl & shock. $[v_{w1}=0]$



$$\tan 40^\circ = \frac{v_f}{u_2 - v_{w2}}$$

$$\Rightarrow v_{w2} = 23.19 \text{ m/sec}$$

$$B) \tan \theta = \frac{v_{f1}}{u_1}$$

$$\Rightarrow \theta = 10.81^\circ$$

$$\begin{aligned}
 A) u_2 &= \frac{\pi d_2 N}{60} \\
 &= \frac{\pi \times 0.5 \times 1000}{60} \\
 &= 26.17 \text{ m/sec}
 \end{aligned}$$

$$\begin{aligned}
 C) \text{ Manometric efficiency} &= \frac{g H_m}{v_{w2} u_2} \\
 &= \frac{\text{Manometric head}}{\text{I.P.}}
 \end{aligned}$$

$$\begin{aligned}
 Q &= A_{f2} \times v_{f2} = (\pi d_2 b_2) \times v_{f2} \\
 &= \pi \times 0.5 \times 0.05 \times 2.5 \\
 &= 0.1966 \text{ m}^3/\text{sec}
 \end{aligned}$$

work done by imp/sec. (I.P.)

$$\begin{aligned}
 &= \frac{W}{g} [v_{w2} u_2] = \frac{9.81 \times 0.1966 \times 23.19}{9.81} \\
 &= 49.9 \text{ kW}
 \end{aligned}$$

$$(1) \left(\frac{N \sqrt{Q}}{H_m^{3/4}} \right)_m = \left[\frac{N \sqrt{Q}}{(H_m)^{3/4}} \right]_p$$

$$(2) \left(\frac{H_m}{D^2 N^2} \right)_m = \left(\frac{H_m}{D^2 N^2} \right)_p$$

$$(3) \left(\frac{Q}{N D^3} \right)_m = \left(\frac{Q}{N D^3} \right)_p$$

$$(4) \left(\frac{P}{N^3 D^5} \right)_m = \left(\frac{P}{N^3 D^5} \right)_p$$

9) specific speed of pumps: ↓

$N_s = \frac{N \sqrt{Q}}{(H_m)^{3/4}}$

output →

↓

Turbine

$N_s = \frac{N \sqrt{Q}}{(H)^{5/4}}$

10) cavitation & ~~Thoma~~ Number: ↓

In centrifugal pump the pressure is minimum on the under side of vane at entry where vapour pressure may be formed, these vapour pressure carried to a region of high pressure to exit where bubble collapse causing pit & severe damage to metal surface. ~~appear~~ entry vane tips at exit are the most susceptible for water hammer attack. The harmful effect of cavitation are:

- 1) pitting & erosion of surface due to continuous hammering
- 2) sudden drop in head & decrease in the efficiency in the pump
- 3) Noise & vibration
- 4) corrosion problem

6) Mean speed required to start the pumping of water

Minimum speed should be such that head developed should be greater than H_m .

(27)

$$\frac{\omega^2 r_2^2}{2g} - \frac{\omega^2 r_1^2}{2g} \geq H_m$$

get

$\omega \rightarrow \text{Rad. sec}^{-1}$

$r_1 \rightarrow d_1/2$

$r_2 \rightarrow d_2/2$

especial:

For a given value of ω & H_m it is possible work-out minimum dia. of impeller (outer dia) which will be required for pumping of water

$$\Rightarrow \frac{\omega^2}{2g} \left[\frac{d_2^2}{4} - \frac{d_1^2}{4} \right] \geq H_m$$

$$\left[\begin{array}{l} \text{Take } d_2 = 2d_1 \\ \Rightarrow d_1 = d_2/2 \end{array} \right]$$

$$\therefore \frac{d_2^2}{4} [1 - 1/4] \geq \frac{2gH_m}{\omega^2}$$

$$\Rightarrow d_2 = \frac{10.23}{\omega} \sqrt{H_m}$$

get

7) Multi-stage centrifugal pump:

(a) To produce high head, impeller should be connected in series. If n no. of impellers are connected in series & each lifts manometric heads equal to (H_m) then

Series connection
 $Q = \text{const.}$

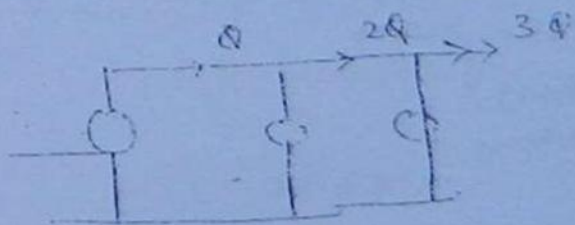
Total head Lifted = $n H_m$

Here total discharge remains constant

(b) Parallel connection: - Impellers or pumps are now parallel to increase the discharge
 \uparrow Head = const.

Total discharge = nQ

Total head remains constant

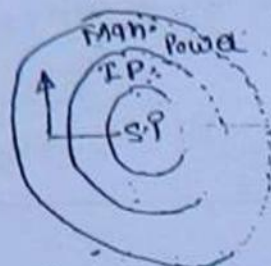


→ Manometric power = $\omega Q H_m$

1. Efficiency
Mechanical ~~power~~ (η_{cy}): ↓

$$\eta_{Mech.} = \frac{I.P.}{S.P.}$$

(220)



[Reverse calculation]

2. Manometric efficiency: ↓

$$\eta_{Mano.} = \frac{\text{Mano. power}}{\text{Imp. power}}$$

$$= \frac{\omega Q H_m}{\frac{\omega Q}{g} [V w_2 u_2]}$$

$$= \frac{g H_m}{V w_2 u_2}$$

define

$\left(\begin{array}{l} \eta_o \rightarrow \text{man} \\ \eta_m \rightarrow \end{array} \right.$

3. overall efficiency: ↓

$$\eta_o = \frac{\text{Man. power}}{S.P.}$$

$$= \eta_{Mech.} \times \eta_{Mano.}$$

NOTE: If Leakage losses are also accounted than Volumetric efficiency

$$\eta_v = \frac{\text{Discharge at delivery point}}{\text{Discharge at Inlet}}$$

$$= Q / (Q + \Delta Q)$$

ΔQ = Loss of water through casing

It may be noted that Loss of water takes after leaving the impeller plates

overall efficiency

$$\eta_o = \eta_{Mech.} \times \eta_{Mano.} \times \eta_{Volumetric}$$

Manometric head H_m

def

— This is the head against which pump has to work.

work done by pump / sec / unit wt of water = Manometric head + Losses

$\Rightarrow A) \quad \boxed{\frac{V_{w2} U_2}{g} = H_m + \text{Losses}}$ def

[Assuming $V_{w1} = 0$]

— If Losses are negligible

$\boxed{H_m = \frac{V_{w2} U_2}{g}}$

(269)

B)

$H_m = \text{Total energy head at outlet} - \text{Total energy head at Anlet}$

$= \left[\frac{p_o}{\rho} + Z_o + \frac{V_o^2}{2g} \right] - \left[\frac{p_i}{\rho} + Z_i + \frac{V_i^2}{2g} \right]$

$V_o \rightarrow V_d$ (velocity in delivery pipe)

$V_i \rightarrow V_s$ (velocity in suction pipe)

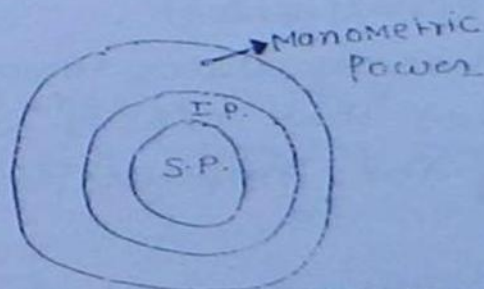
$\boxed{Z_o \approx Z_i}, \quad \boxed{p_o \approx p_s}$

C)

$H_m = (p_s + p_d) + (h_{fs} + h_{fd}) + \left(\frac{V_d^2}{2g} \right)$
 \downarrow Neglig

$\boxed{H_m = (p_s + p_d) + (h_{fs} + h_{fd})}$ def
 \uparrow Static head (H_s) \uparrow Friction head

6) Efficiency & Powers of Pump: ↓



$\boxed{S.P. > I.P. > \text{Manometric power (water)}}$ def

$$= - \left[\text{work done per second in case of Turbine} \right]$$

$$= - \frac{\omega Q}{g} \left[v_{w1} u_1 - v_{w2} u_2 \right]$$

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$$\therefore \text{work/sec.} = \frac{\omega Q}{g} \left[v_{w2} u_2 - v_{w1} u_1 \right]$$

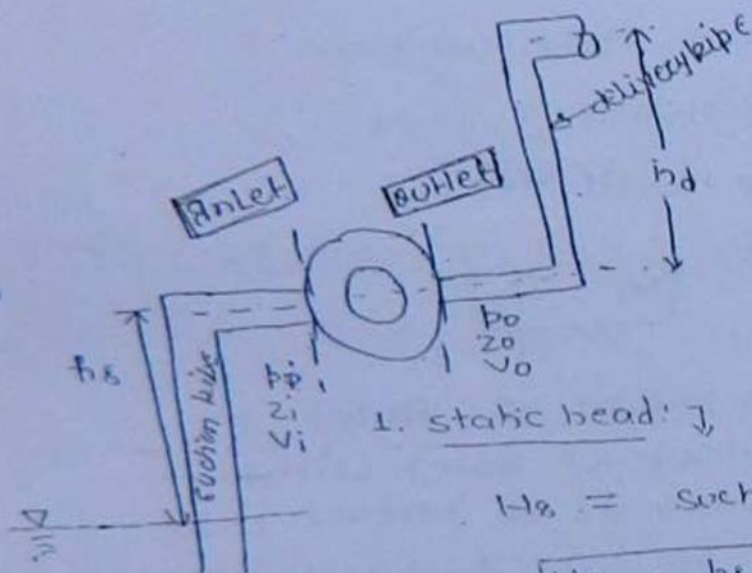
For maxm power $v_{w1} = 0$

work done by ampeller on water/sec

$$= \frac{\omega Q}{g} \left[v_{w2} u_2 \right]$$

= ampeller power / Rotor power

Important definitions: ↓



$H_s = \text{suction head} + \text{delivery head}$

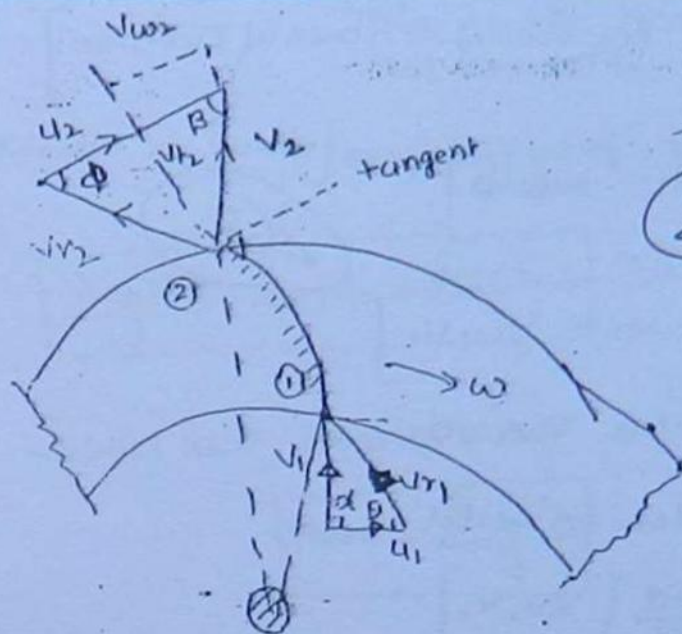
$$H_s = h_s + h_d$$

2. Friction head: ↓
(h_f)

$$\text{Total friction head} = h_{fs} + h_{fd}$$

↓
friction head in
suction pipe

4) velocity diagram :



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u_1 & $u_2 \rightarrow$ Tangential or peripheral velocities

$\phi \rightarrow$ Vane angle at exit

$\phi < 90^\circ$: vanes called backward curved

$\phi = 90^\circ$: vanes are radial at exit

$\phi > 90^\circ$: vanes forward curved

$\alpha = 90^\circ$ { Angle betn u_1 & V_{r1} }

Then

$$V_{w1} = 0$$

$$V_1 = V_{f1}$$

\rightarrow Hence discharge is radial at outlet, it means water enters at inlet without whirl and if there is no impact loss then discharge is said to be without whirl & shock

Let b_2 be the outlet width
 Following points may be noted: ↓

(1) Area of flow:

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$$A_{f1} = \pi d_1 b_1 \rightarrow \text{when thickness of vanes is negligible}$$

$$= \pi d_1 b_1$$

$$= (\pi d_1 - n\ell) b_1 \rightarrow \text{if thickness of each vane is } \ell \text{ \& there are } n \text{ vanes}$$

similarly at exit

$$A_{f2} = \pi d_2 b_2$$

$$= (\pi d_2 - n\ell) b_2$$

(2) Tangential velocity or peripheral velocity:

$$u_1 = \frac{\pi d_1 N}{60}$$

$$u_2 = \frac{\pi d_2 N}{60}$$

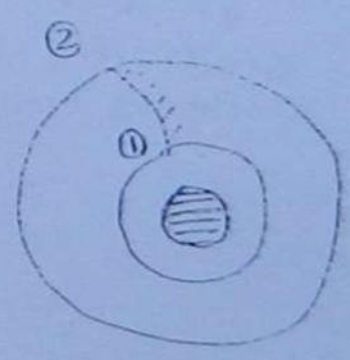
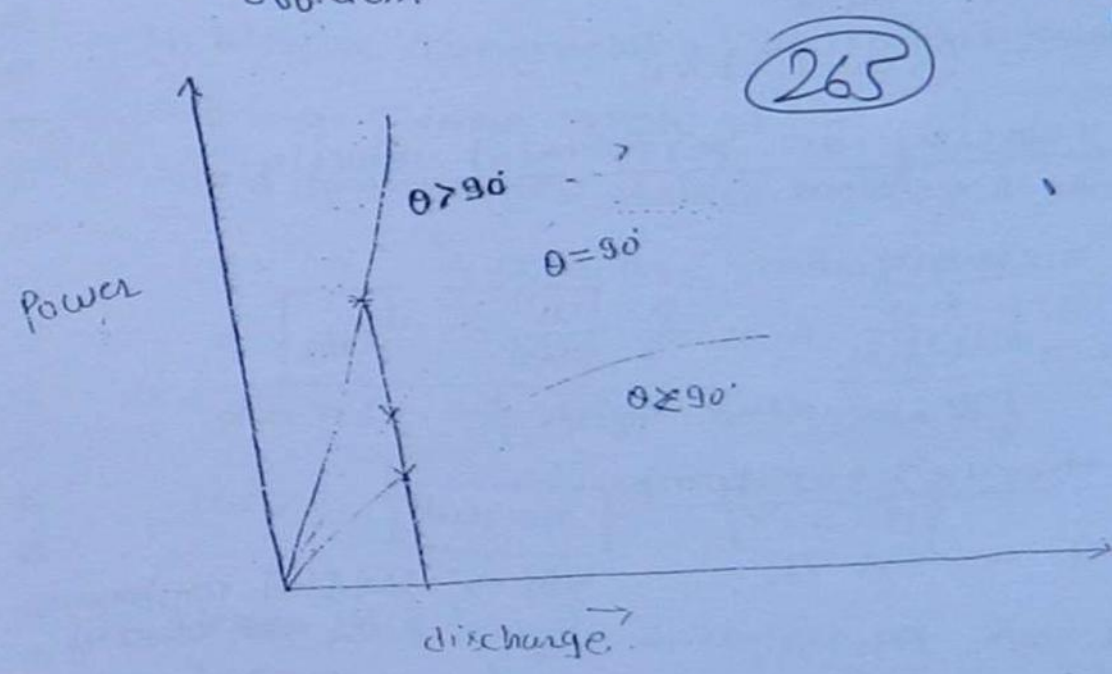
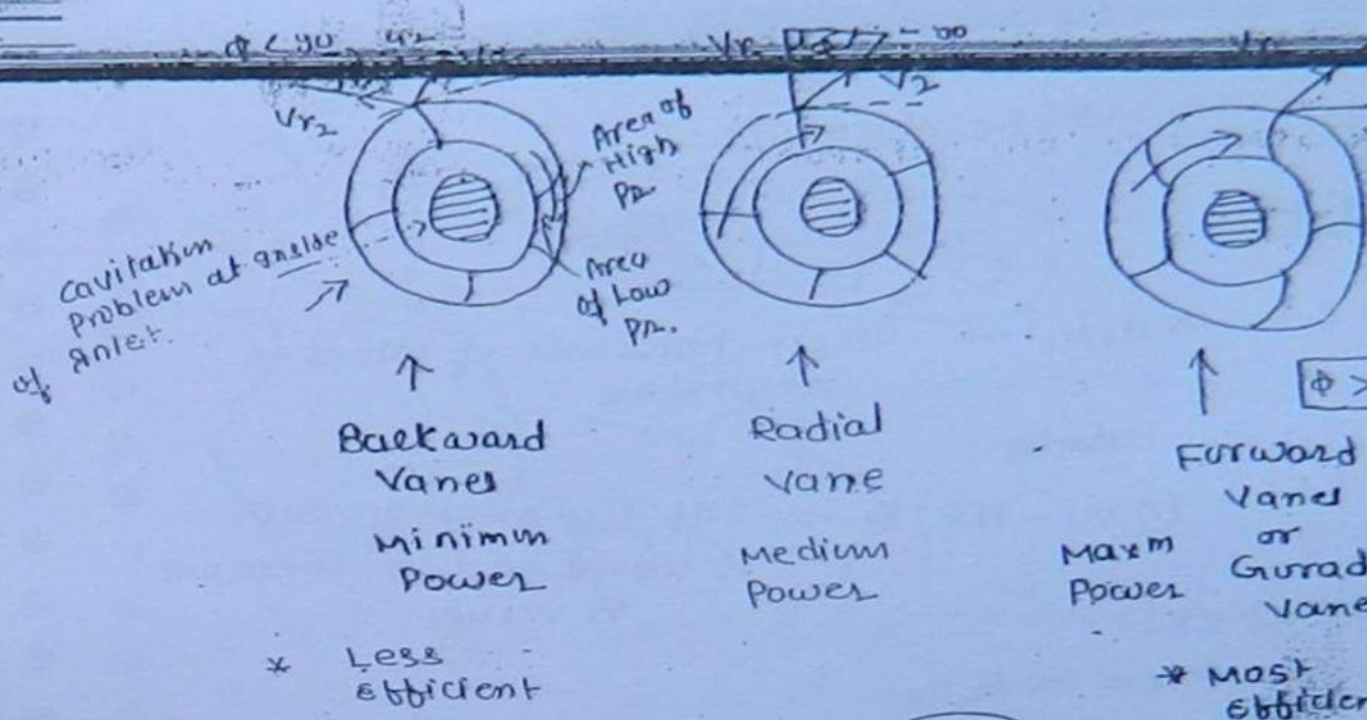
$$\Rightarrow \frac{u_1}{u_2} = \frac{d_1}{d_2}$$

(3) Discharge through the pump:

$$Q = A_{f1} \cdot V_{f1}$$

$$= A_{f2} \cdot V_{f2}$$

$V_{f1} \rightarrow$ Radial component of abs. vel
 V_{f2}



$d_1 \rightarrow$ inner dia. (inlet)
 $d_2 \rightarrow$ Exit dia. (outer dia)

Generally

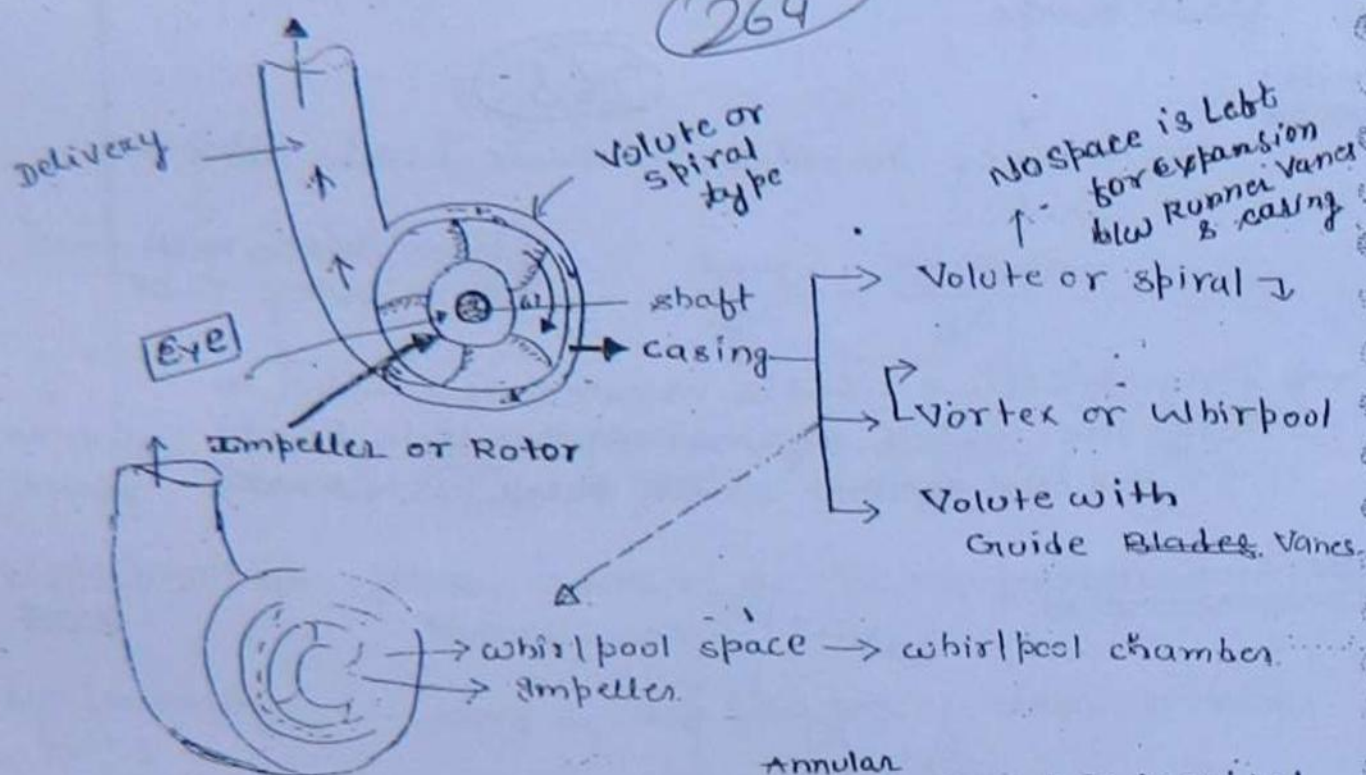
$$d_1 \leq \frac{1}{2} d_2$$

* size of impeller means outer dia $\approx d_2$

hammer problem. ($p_{min} = p_s$)

Imp parts of pump: ↓

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— In whirlpool chamber an ~~annular~~ ^{annular} spaced is provided b/w casing & impeller, this arrangement prevent the formation of eddies and gives an improved performance.

* In case of Volute with Guide Vanes, guides are provided to divert the water, properly such pumps are called diffuser pump or turbine pump. These are adopted when pumps ^{impeller} are connected in series for multistage pumping.

→ These have maximum efficiency but less satisfactory when operating conditions are fluctuating (Power or head)

* Impeller: ↓

It is the rotating unit of a pump similar to runner unit of turbine

→ Impeller has 6 to 12 curved Vanes.

* The No. of vanes on rotating unit are 6 to 12.

* centrifugal pumps are exactly inverse of Francis turbine. It means these are outward radial flow pumps.

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Principle: ↓

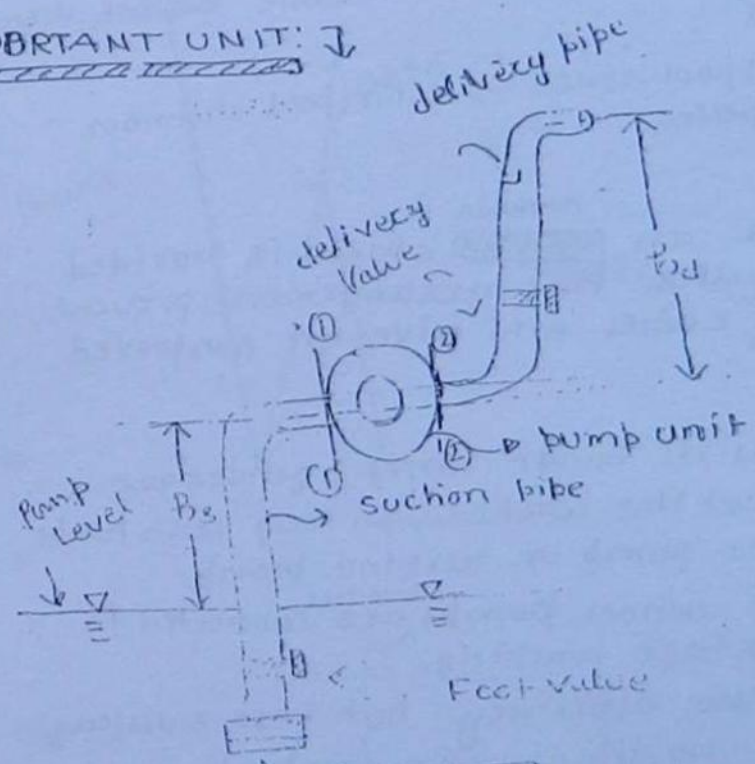
By rotating impeller, pressure head difference is created.

$$= \frac{\omega^2 r_2^2}{2g} - \frac{\omega^2 r_1^2}{2g}$$

$r_1 \rightarrow$ inner radii
 $r_2 \rightarrow$ outer radii

→ This pressure head is created is utilized to lift the water against manometric head (g + the head against which pump has to work)

IMPORTANT UNIT: ↓



strainer → used to prevent entry of blockage material

$$B_{ts} = B_s + B_d$$

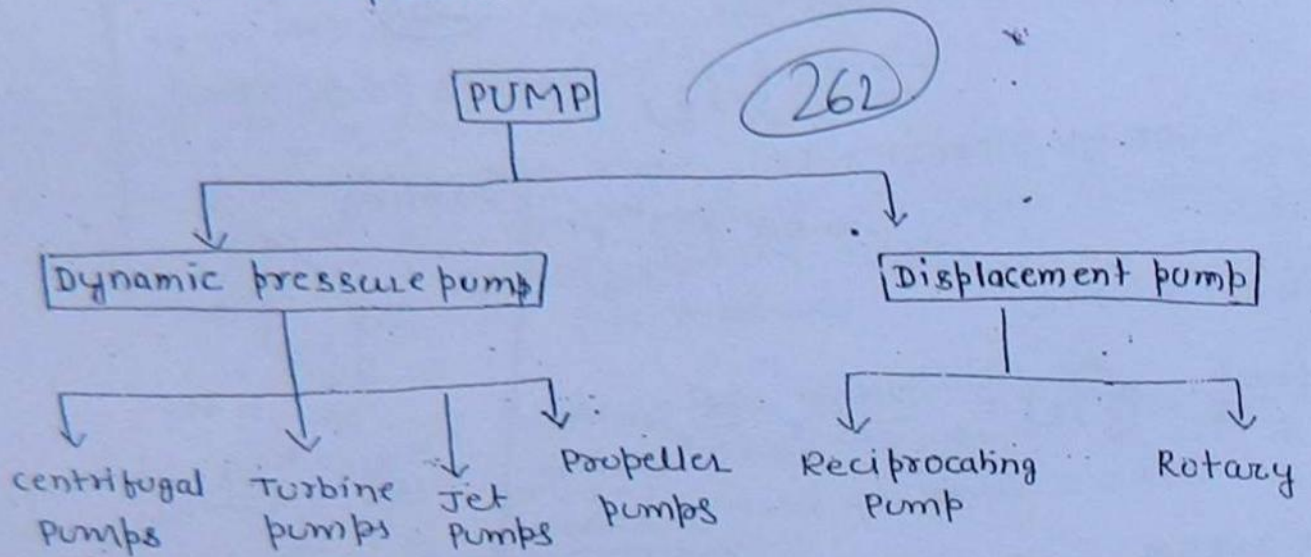
Static head = suction head + delivery head

Pressure in suction pipe is always below atmospheric

→ minimum pressure exist at (1)-(1) i.e. entry of pump

→ This min should not fall vapour pressure of water

- * These utilize mechanical power supplied by the shaft or utilize Man-power to convert it into hydraulic power or water power.



centrifugal pump → These work on the principle of forced Vortex - Motion

Reciprocating pump → works on the principle of suction pressure

* Centrifugal pump: ↓

1) These have high output & high efficiency & are used for Low head & high discharge.

a) When Head is Less than 15m these are called Low Head pump

b) $15\text{m} < H < 45\text{m}$ → medium Head pump

c) $H > 45\text{m}$ → High Head pump

→ Head here means suction Head primarily

NOTE: when $\boxed{\text{Head is } > 40\text{m}}$, single stage centrifugal pump is not desirable therefore Multistage pump in series should be used. It is practically observed that when Head is $\boxed{\text{b/w } 12 \text{ to } 8\text{ m}}$, centrifugal pump are most efficient.

$$= 0.04 \left(\frac{V_1^2}{2g} \right) = 0.04 H$$

(26)

Loss in Runner Vanes

$$= 0.06 \left(\frac{V_2^2}{2g} \right) = 0.06 [0.96 H]$$

Loss in Draft tube

$$= 0.05 \left(\frac{V_3^2}{2g} \right)$$

Apply B.E. n B.W (2) & (3)

$$\left(\frac{V_2^2}{2g} + \frac{P_2}{\omega} + 3 \right) = \frac{P_{atm}}{\omega} + \frac{V_3^2}{2g} + 0 + h_L + 0.05 x$$

$$\Rightarrow 0.95 x = \frac{P_{atm}}{\omega} + \frac{V_3^2}{2g}$$

$$= 10.3 + \frac{2.2^2}{2 \times 9.81}$$

$$\Rightarrow x = \frac{10.5}{0.95} = 11.10$$

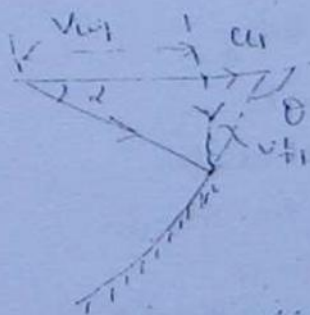
∴ Total head at the exit diameter = 11.10

B.E. n B.W inlet & outlet of Runner

Head at inlet = Head at outlet + work done by Runner / by

$$Y = 11.10 + \frac{\omega \Delta (V_{w1} U_1)}{g} + 0.06 y$$

$$0.94 y = 11.10 + \frac{V_{w1} U_1}{g}$$



$$\Rightarrow \frac{V_{w1} U_1}{g} = 24 - 11.10$$

$$\Rightarrow V_{w1} = 10.409 \text{ m/sec}$$

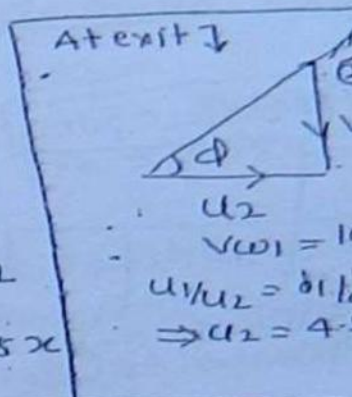
Since $V_{w1} < U_1 \Rightarrow \theta > 90^\circ$

$$\tan \theta = \frac{V_{f1}}{V_{w1}} = \frac{3.3}{10.409}$$

$$\tan (180 - \theta) = \frac{V_{f1}}{U_1 - V_{w1}}$$

$$\Rightarrow \theta = 17.59^\circ$$

$$V_1 = \sqrt{V_{f1}^2 + V_{w1}^2}$$



Total Head at inlet

$$24 = 2 + \frac{P_1}{\omega} + \dots$$

$$= 3 + \frac{P_{atm}}{\omega} + \frac{10.5}{2}$$

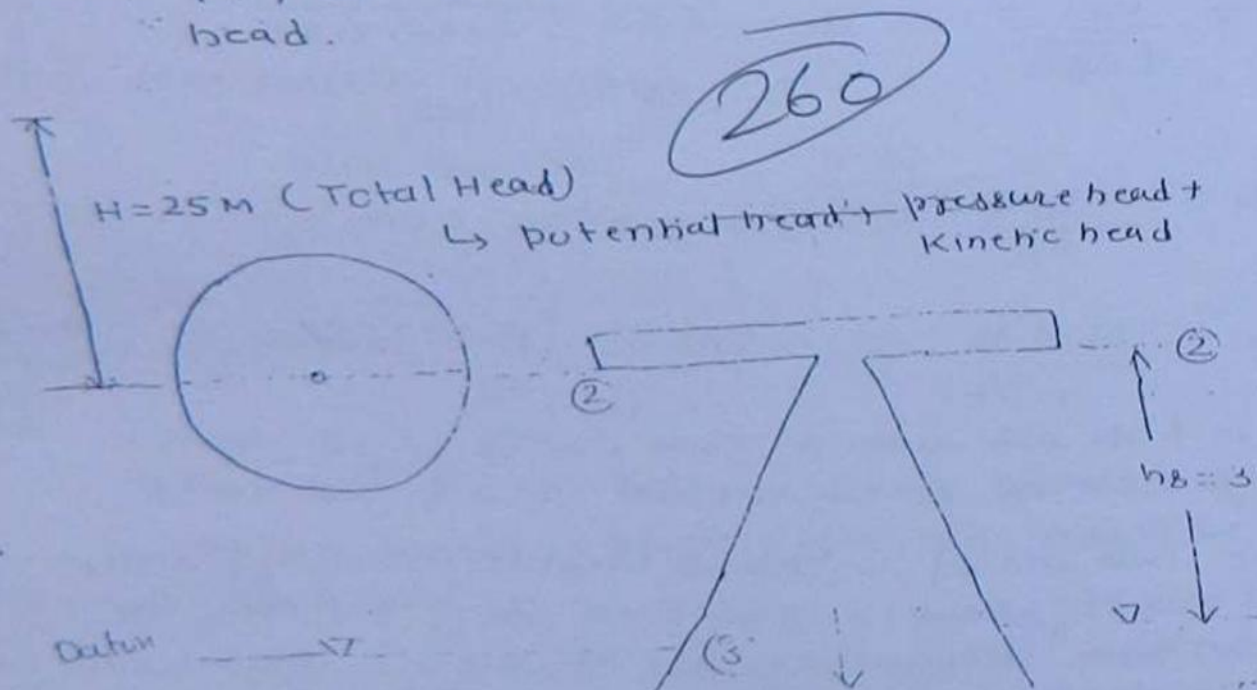
$$\therefore \frac{P_1}{\omega} = 14.92$$

total head of 25M. The centre line of the machine is 3M above the water level, on the tail race. The abs. velocity of flow leaving the vanes on the runner wheel is in the radial direction. the outlet dia = $0.45 \times \text{inlet dia}$. The tangential velocity at exit the rim is 10.8 m/sec & velocity of runner at entry is 3.3 m/sec. And velocity at the exit of the draft tube is 2.2 m/sec. Assuming that flow enters without shock (the runner wheel)

determine

- 1) outlet angle of the guide blades (α)
- 2) inlet & outlet angle (θ & ϕ)
- 3) The pressure head at inlet & outlet of the runner

Assume that LOS due to friction in the guide blade, runner blade plate & the draft tube 4%, 6% & 5% respectively of the available head.



$$H = 25\text{M}$$

$$H_2 = 0$$

$$H_1 = 0.45 \times H_2$$

$$V_1 = 10.8\text{ m/sec}$$

$$V_2 = 3.3\text{ m/sec}$$

$$\left[\begin{array}{l} \alpha = ? \\ \theta = ? \\ \phi = ? \\ p_{1/r} \& p_{2/r} = ? \end{array} \right]$$

$$V_3 = 2.2\text{ m/sec}$$

$$V_{f1} = \frac{Q}{\frac{\pi}{4} (D_o^2 - D_b^2) (\eta_{f1})} = 6.12 \text{ m/sec}$$

$$= V_{f2} = V_2$$

$$H = \frac{V_2^2}{2g} + \frac{V_{w1} u_1}{g}$$

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$$\Rightarrow H = \frac{6.12^2}{2 \times 9.81} + 0.9H$$

$$\Rightarrow H = 19.1 \text{ m}$$

$$\text{S.P.} = \frac{H \cdot P \cdot \eta_o}{1000} = \omega \rho H \times \eta_A \times \eta_m$$

$$\text{S.P.} = 10,972 \text{ kW}$$

$$\phi = \frac{u_1}{\sqrt{2gh}} \Rightarrow u_1 = 2.0 \times \sqrt{2 \times 9.81 \times 19.1}$$

$$u_1 = 38.71 \text{ m/sec}$$

$$u_1 = \frac{\pi D_o N}{60} \Rightarrow N = 184.76 \text{ rpm}$$

$$N_s = \frac{N \sqrt{P}}{H^{5/4}} = \frac{184.76 \sqrt{10,972}}{(19.1)^{5/4}} = \underline{\underline{454.31}}$$

$$N_s (\text{specific speed}) = 454.31 \text{ (SI unit)}$$

The hub dia. of a Kaplan turbine working at a head of 12 m is 0.35 times dia. of runner. The turbine is running at 150 rpm. If the angle of extreme edge at outlet is 15°. Flow ratio is 0.26.

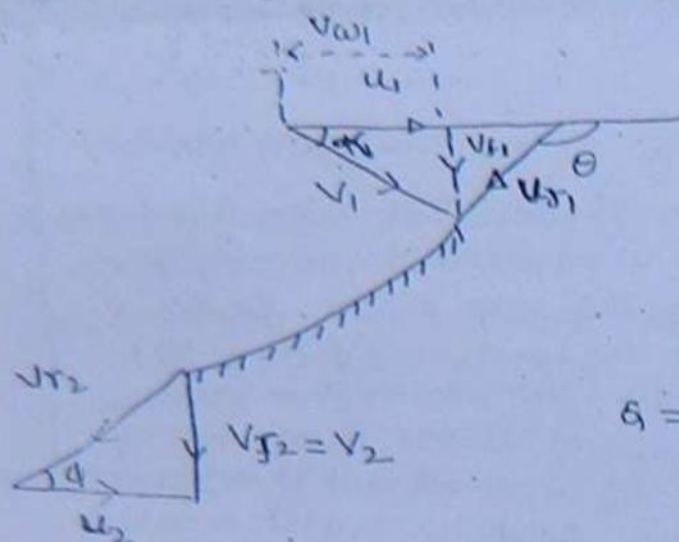
$$\text{a) Find dia. of runner} \rightarrow D_o = 6.55 \text{ m}$$

$$\text{b) Find dia. of boss} \rightarrow$$

$$\text{c) discharge through runner} \rightarrow 231.77 \text{ m}^3/\text{sec}$$

$$\text{d) Assume velocity of vort at outlet is}$$

Prob 3



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$$Q = \frac{\pi}{4} (D_b^2 - D_h^2) \times V_{f1}$$

$$\Rightarrow V_{f1} = 8.90 \text{ m/sec}$$

$$V_{f1} = V_{f2} = V_2 = 8.90 \text{ m/sec.}$$

$$\tan(180 - \theta) = \left(\frac{V_{f1}}{u_1 - V_{w1}} \right) = \frac{8.90}{(25.525 - 7.58)}$$

$$\therefore \tan(180 - \theta) =$$

$$\Rightarrow \boxed{\theta = 153.6^\circ}$$

From exit velocity diagram

$$\tan \phi = \frac{V_{f2}}{u_2} = \frac{8.95}{25.525}$$

$$\Rightarrow \boxed{\phi = 19.22^\circ}$$

Case IInd

Repeat the calculation for $d = D_h$

NOTE

When θ is given that runner dia is d m then θ is the ^{outer} dia. of the runner. & in case of the Pelton wheel pitch circle dia. should be taken

runner

Prob 2

A Kaplan turbine has a dia. of 4 m. And hub dia. ^{9.2} meter. The discharge through turbine is 70 m³/sec. The η_B & η_m can be taken as 0.9 & 0.85 respectively. Assuming absence of whirl at outlet & discharge is free from friction estimate the Net head available in the turbine & the power developed. Speed Ratio is 2. Also estimate specific speed.

5) Under assumptions B: Eqn may be applied at inlet and exit point

when $v_{r2} = 0$, friction on the blades is negligible

Prob 1

A propeller runner turbine runner has outer dia 4.5m & dia of hub is 2m. It is required to develop power 20600 kW when running at 150 r.p.m. under a head of 21 M. Assuming hydraulic efficiency 94% & overall efficiency of 88%. Determine the runner vane angle at inlet & outlet at the mean exit of vane. Assume velocity of whirl at outlet is zero. Also determine the vane angle at outlet & inlet at outer dia.

Calculation at Mean diameter: \downarrow

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$$D_m = \frac{D_o + D_b}{2}$$

$$= 3.25 \text{ m}$$

$$u_1 = u_2 = \frac{\pi D_m N}{60}$$

$$= \frac{\pi \times 3.25 \times 150}{60}$$

$$= 25.525 \text{ m/sec.}$$

$$\text{Shaft Power} = 20600 \text{ kW}$$

$$\eta_o = 0.88$$

$$\eta_o = \frac{S.P.}{H.P.}$$

$$WQH = \frac{20600}{0.88}$$

$$\Rightarrow 4.815 \times 21 = \frac{20600}{0.88}$$

$$\Rightarrow Q = 113.63 \text{ m}^3/\text{sec.}$$

$$\text{Hydraulic efficiency } (\eta_h) = \frac{v_{w1} u_1}{gH} = 0.94$$

$$\Rightarrow v_{w1} = 7.58 \text{ m/sec.}$$

$$A_1 V_1 = A_2 V_2 = \frac{\pi D^2}{4} V$$

$$\begin{aligned} \text{ii)} \quad Q &= A_1 \cdot V_1 \\ &= A_2 \cdot V_2 \\ \Rightarrow V_1 &= V_2 \end{aligned}$$

(256)

iii) If friction losses are negligible then

$$V_{r1} = V_{r2}$$

iv) Vane angle from D_o to D_b may change.

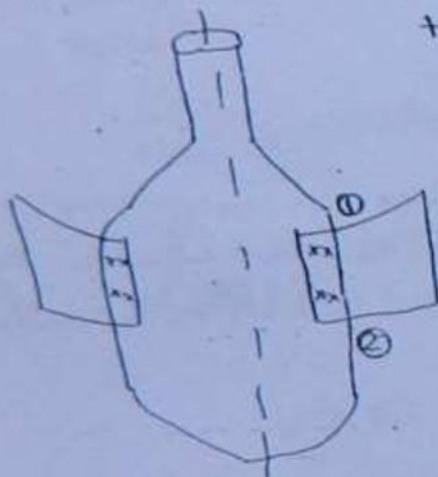
$$u_1 = u_2 = \frac{\pi D N}{60}$$

u_1 at $D = D_o$

u_2 at $D = D_b$

$$\left\{ D = D_{avg} = \frac{D_o + D_b}{2} \right\}$$

* Power & Efficiency calculation are similar to Francis turbine & velocity diagram also similar to Francis turbine.



$$H = \frac{V_o^2}{2g} + \frac{V_{w1} u_1}{g}$$

NOTE:

1) If entry & exit points are not clearly maintain then calculations should be made at outer dia. $d = D_o$

2) Velocity of flow through runner remains constant. $A_1 = A_2$

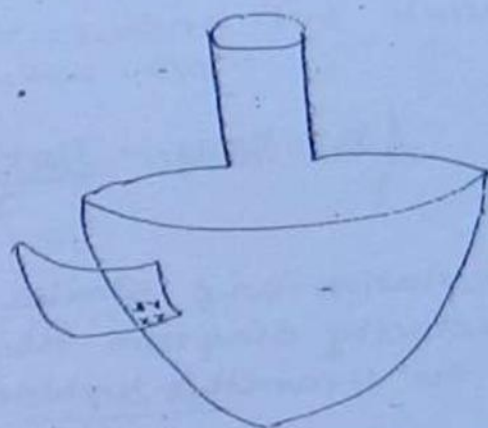
3) The Inlet Vane Angle θ , normally $> 90^\circ$. No. of Blades is Normally 3 to 6

4) Ignoring friction resistance, $V_{r1} = V_{r2}$ may be taken

3) Kaplan & Propeller Turbines

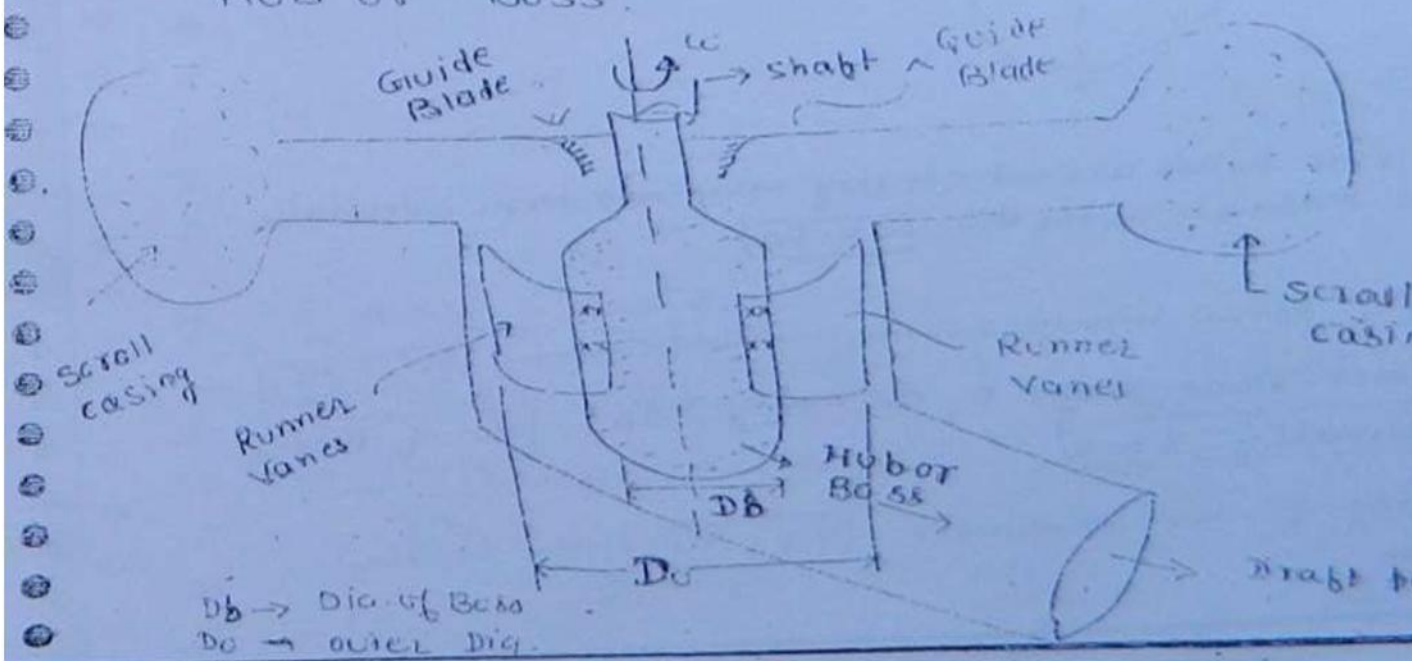
(255)

- Axial flow Rxn turbines
- Propeller turbines & Kaplan turbines are similar in principle except that Kaplan has adjustable blades over propeller w.r.t. direction of the blades are flexible i.e. Propeller turbines have fixed vanes connected through rivets permanently where Kaplan has provided bolts & an additional bolt hole.



Important unit of Propeller turbines: ↓

- a) For the axial flow reaction turbine the shaft of the turbine is vertical, the lower end of the shaft is made larger which is known as Hub or Boss.



for a turbine to run under a head of 1m

$$Q_u = \frac{Q}{\sqrt{H}}$$

(254)

> Unit power: It is the theoretical power which a turbine would produce under a head of 1m.

$$P_u = \frac{P}{H^{3/2}}$$

Model quantities: ↓

When the results obtained from the experiments conducted on the model are applied to the prototype in the field for similarity exists:

$$a) \frac{H}{N^2 D^2} = \text{constant}$$

$$i.e. \frac{H_m}{N_m^2 D_m^2} = \frac{H_p}{N_p^2 D_p^2}$$

$$b) \frac{P}{N^3 D^5} = \text{constant}$$

$$\Rightarrow \frac{P_m}{N_m^3 D_m^5} = \frac{P_p}{N_p^3 D_p^5}$$

$$c) \frac{Q}{N D^3} = \text{constant}$$

$$\Rightarrow \frac{Q_m}{N_m D_m^3} = \frac{Q_p}{N_p D_p^3}$$

Specific speed for various turbines:

a) Pelton wheel

→ Single Jet $N_s = 10 \text{ to } 30$
→ Multi Jet $= 30 \text{ to } 60$

(253)

b) Francis turbine

$N_s \Rightarrow 60 \text{ to } 300$

c) Kaplan turbine

$N_s \Rightarrow 300 \text{ to } 1000$

Unit quantities: ↓

A turbine operates most efficiently at its design points at a particular combination of H , Q & N . But in practice these variables do not remain constant therefore unit quantity is important to

- (i) Predict the behaviour of turbine working at different condition
- (ii) To make the comparison of performance turbine of the same type but diff. size
- (iii) It may be used to compare the performance of turbine of different type
- (iv) To correlate the use of experimental data

a) unit speed: It is the theoretical speed at which a given turbine would operate under a head of 1m.

$$N_u = \frac{N}{\sqrt{H}}$$

$$N_s = \frac{N \sqrt{P}}{H^{5/4}}$$

$P \rightarrow H \cdot P$ Power
 $H \rightarrow$ Head

$$L = 1200 \text{ m}$$

$$\text{no. of turbine} = 3$$

$$H = 360 \text{ m}$$

$$f = 0.02$$

NOTE:-

For multiple jet
 pelton wheel the
 sp. speed is Base
 on Break Power
 per jet.

$$\eta = 0.85 = \frac{\text{S.P.}}{\text{water available at the base of nozzle}}$$

$$0.85 = \frac{55.06}{WQH}$$

$$\Rightarrow Q = 1.834 \text{ m}^3/\text{sec.}$$

Total discharge supplied by pipe = 18.34

\therefore Discharge through each nozzle

$$= \frac{18.34}{3} = 6.11 \text{ m}^3/\text{sec}$$

$$V_1 = C_v \sqrt{2gH} = 0.98 \sqrt{2 \times 9.81 \times 360}$$

$$\therefore V_1 = 82.36 \text{ m/sec}$$

discharge dia. of each ^{jet} nozzle is

$$\frac{\pi d^2}{4} \times V_1 = \frac{6.11}{82.36} \Rightarrow d = 0.097 \text{ m} = 97 \text{ mm}$$

$$B_f = \frac{f L Q^2}{12.1 D^5} = \frac{0.02 \times 1200 \times 18.34^2}{12.1 D^5}$$

$$= 12$$

$$\Rightarrow D = 0.88 \text{ m}$$

$$\text{Power at Nozzle} = \rho g Q H = 981 \times 700 \times 0.1 = 68670 \text{ W} = 68.67 \text{ kW}$$

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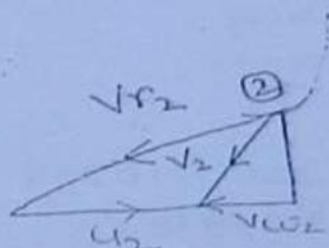
$$\eta_B = \frac{V_{w1} u_1 + V_{w2} u_2}{gH} \quad \left\{ \begin{array}{l} \text{Losses at} \\ \text{Nozzle are} \\ \text{negligible} \end{array} \right.$$

$$V_{w1} = V_1 = 117.19$$

$$V_{r1} = V_1 - u_1 = 117.19 - 52.35 = 64.83$$

Incoming bucket to be frictionless

$$V_{r2} = V_{r1} = 64.83 \text{ m/sec.}$$



$$|u_2| + |V_{w2}| = |V_{r2}| \cos \alpha$$

$$= 64.83 \cos \alpha$$

$$\therefore V_{w2} = 10.261 \text{ m/s}$$

$$\therefore \eta_B = 97.18\%$$

Prob 4

A pipeline 1200 m long supply water to three single pelton wheel. The head at the nozzle is 360 M. $C_v = 0.98$, friction factor for the pipe = 0.02. The turbine efficiency based on the head at the nozzle is 0.85. If specific speed at the base of nozzle is 15.3 [Ns].

such that $n = \text{r.p.m.}$, $P = \text{KW}$, $H = \text{m}$. If head loss due to friction in the line is 12M and operating speed of turbine is 300 r.p.m. determine

- Total power developed
- The dia of pipeline
- The dia of pipe each nozzle.
- Discharge

$$= \frac{\omega Q}{g} [v_{w1} u_1 + v_{w2} u_2]$$

$$= \frac{9.81 \times 0.75}{9.81} [25.68 + 0.855] \times 12$$

256

Efficiency: ↓

$$\text{Hydraulic efficiency } \eta_{cy} = \frac{R.P.}{K.E./\text{second}}$$

$$K.E./\text{second} = \frac{1}{2} \times \left(\frac{\omega \Delta}{g} \right) \times v_1^2$$

$$= \frac{1}{2} \times \frac{9.81 \times 0.75}{9.81} \times 25.68^2$$

$$= 247.29 \text{ kW}$$

$$\therefore \eta_R = \frac{238.8}{247.29} \times 100 \%$$

$$= 96.56\%$$

prob 3
sp2004

A pelton wheel has mean bucket dia. of 1m and is running at 1000 R.P.M. The net head on the pelton wheel is 700m. If the side clearance angle is 15° and discharge through the nozzle $0.1 \text{ m}^3/\text{sec}$. Find

a) Power available at the nozzle

b) Hydraulic efficiency of turbine, take $C_v = 1.0$

$$u_1 = u_2 = \frac{\pi D N}{60} = \frac{\pi \times 1 \times 1000}{60} = 52.35 \text{ m/s}$$

$$\text{Net head on the pelton wheel} = 700 \text{ m} \quad (1)$$

$$\phi = 15^\circ$$

$$Q = 0.1 \text{ m}^3/\text{sec}, \quad C_v = 1.0$$

$$V = C_v \sqrt{2gH}$$

$$= 1.0 \sqrt{2 \times 9.81 \times 700}$$

$$= 117.19 \text{ m/s}$$

$u = \frac{\omega r}{60}$
 $\phi = 15^\circ$
Can not be applied to calculate η
6/2/2004

A Pelton wheel has a mean bucket speed of 12 m/s. It is supplied with water at a rate of 750 l/sec under a head of 35 m. If the buckets deflect the jet by an angle of 160° . Find the power & efficiency of the bucket taking $C_v = 0.98$ & Neglecting the friction in the

$$u_1 = u_2 = 12 \text{ m/sec}$$

$$Q = 750 \text{ l/sec}$$

$$= 0.75 \text{ m}^3/\text{sec}$$

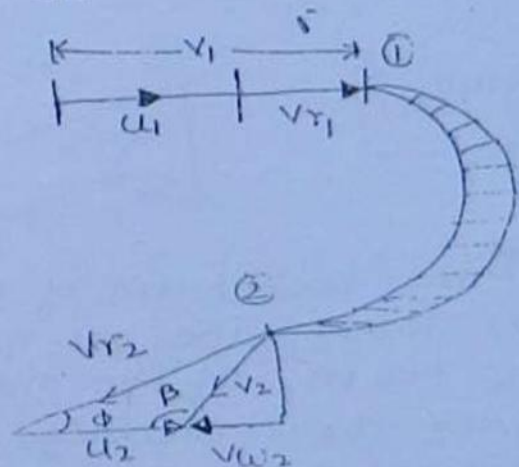
$$H = 35 \text{ m}$$

$$\text{Deflection Angle} = 180^\circ - \phi$$

$$= 160^\circ$$

$$\therefore \text{side clearance angle or runner vane angle } (\phi) = 20^\circ$$

$$C_v = 0.98$$



$$V_{r1} = V_1 - u_1$$

$$= 25.68 - 12$$

$$= 13.68 \text{ m/sec}$$

$$V_1 = C_v \sqrt{2gH}$$

$$= 0.98 \sqrt{2 \times 9.8 \times 35}$$

$$= 25.68 \text{ m/s}$$

$$V_{r1} = V_{r2} = 13.68 \text{ m/sec}$$

$$\frac{20^\circ}{20^\circ} [V_{r2} \cos \phi] = [u_2] + [V_{w2}]$$

$$\Rightarrow 13.68 \cos 20^\circ = 12 + V_{w2}$$

$$\Rightarrow V_{w2} = 0.885 \text{ m/sec}$$

700
ES/2001

Pelton wheel

- i) Head at the base of Nozzle = 32 m (H)
- ii) discharge of the Nozzle = $0.18 \text{ m}^3/\text{sec}$ (Q)
- iii) area of Jet = 7500 mm^2 (a)
- iv) Power available at the shaft = 44 kW (S.P.)
- v) Mech. efficiency = 94% ($\eta_m = 0.94$)

calculate the power loss

- a) In the Nozzle
- b) In the Runner
- c) In the Mech. friction

$$\eta_m = \frac{\text{S.P.}}{\text{R.P.}}$$

(248)

$$\text{R.P.} = \frac{\text{S.P.}}{\eta_m} = \frac{44}{0.94} = 46.8 \text{ kW}$$

$$\text{velocity of Jet} = \frac{Q}{a} = \frac{0.18}{7500 \times 10^{-6}} = 24.0 \text{ m/sec}$$

$$\begin{aligned} \text{K.E. per second of Jet} &= \frac{1}{2} \times \left(\frac{WQ}{g} \right) \times V_1^2 \\ &= \frac{1}{2} \times \frac{9.81 \times 0.18}{9.81} \times (24)^2 = 51.84 \text{ kW} \end{aligned}$$

Power available at the base of Nozzle

$$= WQH$$

$$= 9.81 \times 0.18 \times 32 = 56.5 \text{ kW}$$

Loss at Nozzle

$$= 56.5 - 51.84 = 4.66 \text{ kW}$$

Loss in Runner

$$= 51.84 - 46.8 = 5.04 \text{ kW}$$

$$\text{Mechanical Losses} = 46.8 - 44 = 2.8 \text{ kW}$$

$$= 44 - \text{loss}$$

$$= 44 - \text{shaft power}$$

VOLUMETRIC EFFICIENCY:

If entire volume of Jet is not striking to the Vane than Volumetric efficiency

$$\eta_v = \frac{Q'}{Q}$$

Q' = total volume of water striking the Jet per sec.

$Q =$

Specification for design of Pelton wheel: ↓

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(a) speed ratio $\phi = \frac{u_1}{\sqrt{2gH}}$

[ϕ is also called side clearance angle]

$$= 0.45 \text{ to } 0.47$$

(b) Angle of deflection of Jet through the bucket is $\approx 165^\circ$ [if not given]

(c) Jet ratio = $\frac{\text{Dia. of pitch circle}}{\text{Dia. of Jet}} = D/d = m$
 $\approx 10 \text{ to } 15$ & commonly 9 is adopted 12.

(d) No. of buckets in the runner should be as less as possible in order to minimize friction losses. But in order to maximize volumetric efficiency optimum No. of Buckets are given by

$$Z = 15 + \frac{D}{2d} = 15 + \frac{0.5m}{2} \quad [m=12]$$

$$\approx 21 \quad [\text{Range } 18 \text{ to } 25]$$

(e) No. of Jets $(n) = \frac{\text{Total discharge}}{\text{Discharge through one jet}} \rightarrow 6$

(f) width of bucket may be taken as height times dia. of Jet & vertical depth of bucket may be taken as 1.5 times dia. of Jet.

a) Efficiency of Nozzle: ↓

$$\eta_N = \frac{K.E./\text{sec}}{\text{Power at the base of Nozzle}} = \frac{\frac{1}{2} \frac{wQ}{g} \times V_1^2}{wQH}$$

(246)

$$= \frac{V_1^2}{2gH} = \frac{C_v^2 (2gH)}{(2gH)} = C_v^2$$

∴ ~~η_N = C_v²~~

$$\eta_N = C_v^2$$

b) Hydraulic efficiency: ↓

$$\Rightarrow \eta_H = \frac{R.P.}{K.E./\text{sec of Jet}}$$

$$= \frac{\frac{wQ}{g} [V_{w1} u_1 + V_{w2} u_2]}{\frac{wQ}{2g} \times V_1^2}$$

NOTE! -

If hydraulic efficiency is calculated on the basis of power available at the base of nozzle then

$$\eta_H = \frac{R.P.}{\text{Power available at the base of Nozzle}}$$

$$= \frac{\frac{wQ}{g} [V_{w1} u_1 + V_{w2} u_2]}{wQH}$$

$$\eta_H = \frac{V_{w1} u_1 + V_{w2} u_2}{gH}$$

It means Energy loss at the base of nozzle is neglected

c) Mechanical efficiency ↓

$$\eta_M = \frac{S.P.}{R.P.}$$

d) Overall efficiency

Accounting the Nozzle Loss

$$\eta_o = \eta_N \times \eta_H \times \eta_M$$

Work done by water on the runner: (per second)

$$= \frac{\omega Q}{g} [v_{w1} u_1 - [-v_{w2}] u_2]$$

$$= \frac{\omega Q}{g} [v_{w1} u_1 + v_{w2} u_2]$$

Discharge through turbine: \downarrow

(245)

$$= n \times \left(\frac{\pi d^2}{4} \right) \times v_1$$

where $n \rightarrow$ No. of vanes
 $d \rightarrow$ dia. of Jet
 $v_1 \rightarrow$ velocity of Jet
 $= C_v \sqrt{2gH}$
 $H \rightarrow$ Net head at

Power available at the base of Nozzle: \downarrow base of nozzle

(1) $\frac{W.P.}{(H.P.)} = \omega Q H$ [Power available at the base of nozzle]

(3) $R.P. = \frac{\omega Q}{g} [v_{w1} u_1 + v_{w2} u_2]$ [$u_1 =$]

~~245~~ ²⁴⁵ Power available at the inlet of Jet/vane

= K.E. of Jet per second

$$= \frac{1}{2} \times \text{mass flowing per second} \times v_1^2$$

$$= \frac{1}{2} \times \rho Q \times v_1^2$$

$$= \frac{1}{2} \left(\frac{\omega}{g} \right) (a v_1) \times v_1^2$$

[$a \rightarrow$ area of Jet]

$$= \frac{\omega a v_1^3}{2g}$$

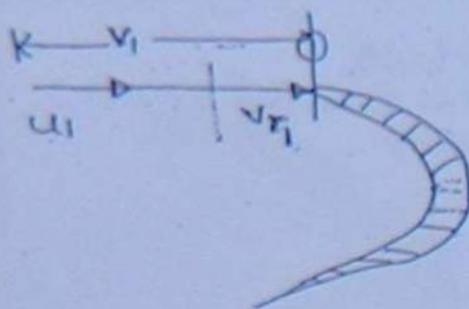
11. 1/1/11

(4) Power available at the shaft

S.P. or Break Power

$$= R.P. - \text{Losses}$$

Exit Velocity Diagram:



$$V_{r1} = V_1 - u_1$$

$$\theta = 0^\circ$$

$$\alpha = 0^\circ$$

for

$$V_{w1} = V_1 \cos \alpha = V_1$$

$V_{f1} = 0$ { radial velocity at inlet }

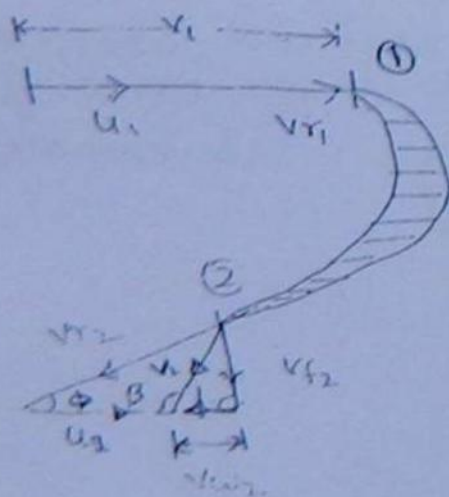
244

If Loss of friction on the surface of vane is negligible then $V_{r1} = V_{r2}$

$V_{r2} = V_{r1}$ ----- when friction loss is negligible

$V_{r2} = K V_{r1}$ ----- If friction losses are accounted.
($K < 1$)

Exit velocity diagram: ↓



$$V_{r2} + u_2 = V_2$$

$$u_2 = u_1 = \frac{\pi D N}{60}$$

$$\tan \phi = \frac{V_{f2}}{u_2 - V_{w2}}$$

where $V_{w2} = -ve$

$$u_1 = u_2 = \frac{\pi D N}{60}$$

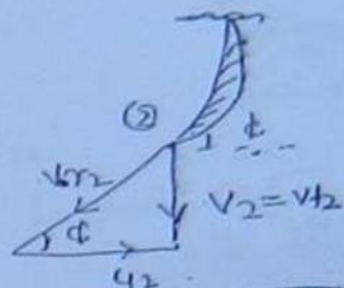
$d = \text{dia of jet}$
 $D = \text{dia of bucket (mean)}$

\vec{V}_r = Relative velocity at inlet
 $= V_1 - u_1$

$$\vec{V}_r + \vec{u}_1 = \vec{V}_1$$

At exit

$$\vec{V}_r + \vec{u}_2 = \vec{V}_2$$



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$$\text{Def'n Angle} = 180 - \phi$$

ϕ = Vane angle at exit

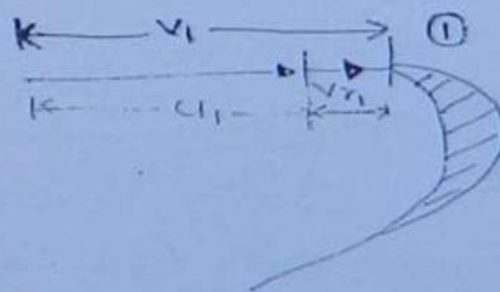
θ = Angle b/w u_1 & \vec{V}_r

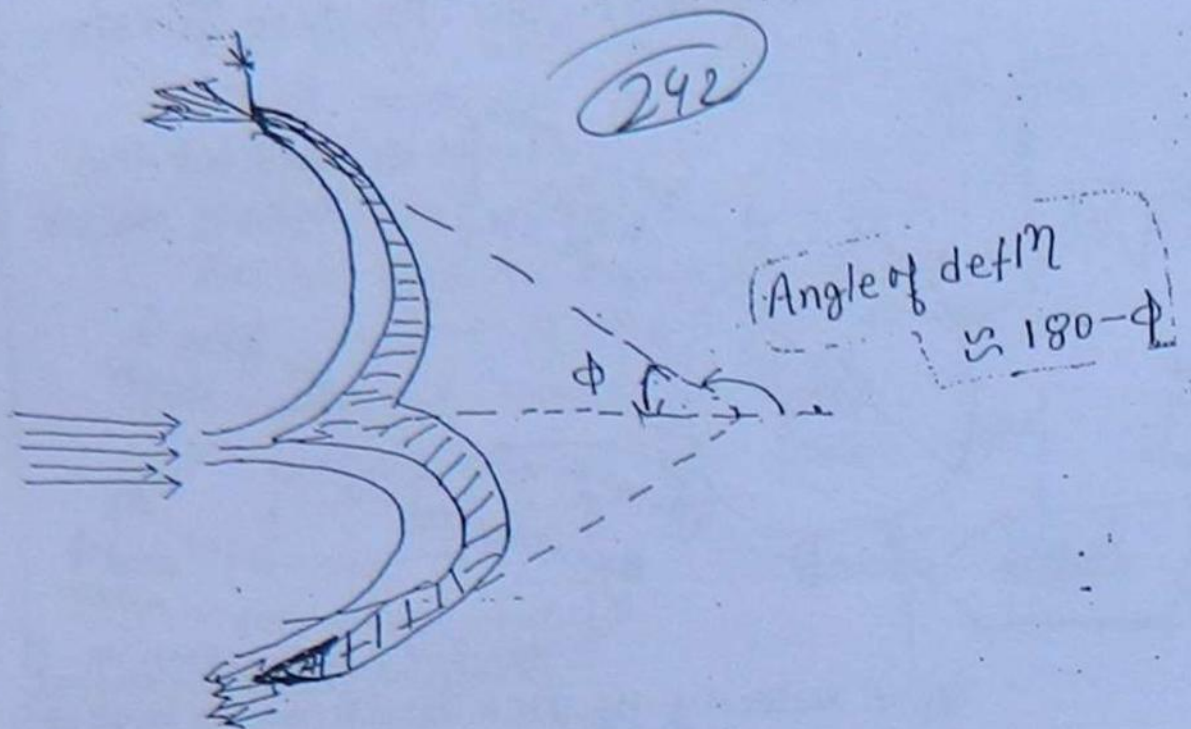
θ = Vane angle at inlet

Velocity triangle at inlet: ↓

Since u_1, \vec{V}_r, V_1 are in same direction

∴ Inlet vel. triangle reduces into a straight line.

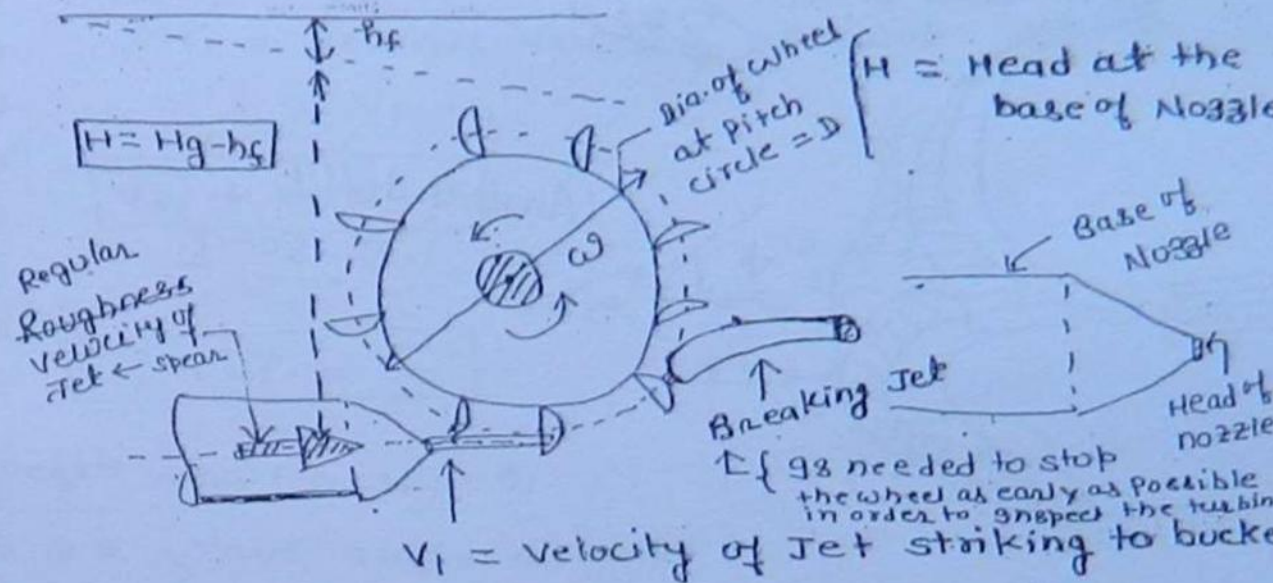




- Theoretically deflⁿ. angle should be 180° in order to produce maxm. work-done but practically it causes retardation of coming vanes by hitting on the vanes hence appropriate angle of deflⁿ back of $(160^\circ \text{ to } 165^\circ)$ is desirable. In order to minimize friction losses vanes are polished on inside.
- Two hemispherical buckets are provided because it will cancel the y-comp of force and alignment of bucket can be preserved.

2. PELTON WHEEL

Tangential flow impulse turbine



$$V_1 = C_v \sqrt{2gH}$$

$$C_v \approx 0.97 \text{ to } 0.98$$

(241)

→ When wheel is vertical and shaft is horizontal then it is called 'horizontal alignment', whereas 'vertical alignment' is that wheel is horizontal & shaft is vertical.

→ In the Pelton wheel hemispherical bucket are mounted on the pitch circle which may be 15 to 25 in number.

→ As far as possible less no. of vanes should be provided so as to minimize friction losses but specified minimum no. of vanes should be provided so as to prevent the loss of discharge without hitting the vanes. No. of vanes required depends upon Jet ratio which is defined as dia. of pitch circle to dia. of Jet.

$$m = \text{Jet Ratio} = \frac{\text{Dia. of Pitch circle}}{\text{Dia. of Jet}} = D/d$$

$$\text{No. of Vanes} = 15 + 0.5m$$

where m is generally 14 to 15 & 12

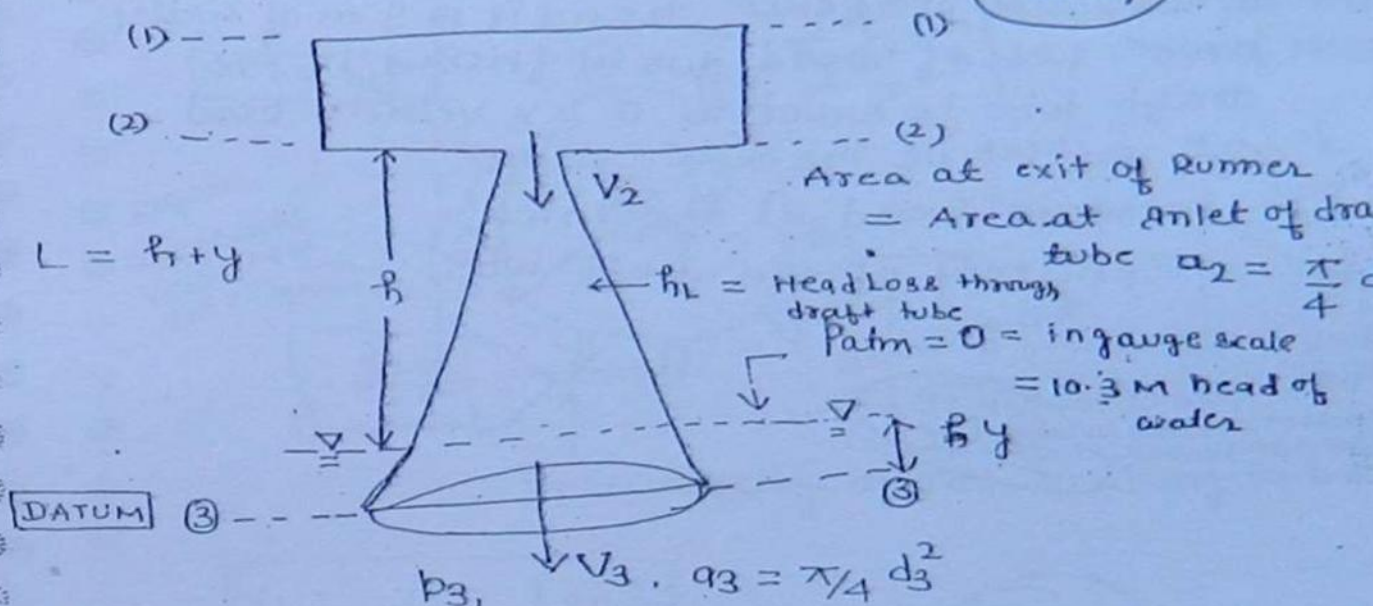
A vertical draft tube having inlet & outlet dia. 1m & 1.5m discharges water at outlet with velocity of 2.5 m/sec. The total length of the draft tube is 6m & 120m. Length of the draft tube is immersed in water. If the atmospheric pressure head is 10.3 m of water and loss of head due to friction in the draft tube is equal to $0.2 \times$ velocity head at outlet of the tube. Find

- (i) Pressure head at the inlet.
- (ii) Efficiency of the draft tube.

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DRAFT TUBE 7

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Apply B. Eqn b/w (2) & (3)

$$(h+y) + \frac{V_2^2}{2g} + \frac{p_2}{\rho g} = 0 + \frac{V_3^2}{2g} + \frac{p_3}{\rho g} + h_L$$

$$= \frac{V_3^2}{2g} + \frac{p_{atm}}{\rho g} + y + h_L$$

NOTE: Since p_2 is below atmosphere than it should be such that it should not fall below vapour pressure.

Efficiency of draft tube:

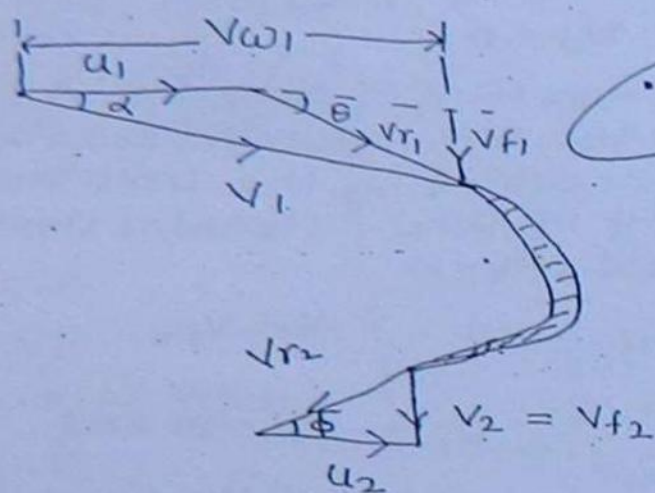
$$\eta = \frac{\text{Actual converging of K.H. into pressure head}}{\text{original K.H.}}$$

$$\eta = \frac{\frac{V_2^2}{2g} - \frac{V_3^2}{2g} - h_L}{\frac{V_2^2}{2g}}$$

$$\frac{1}{60} = \frac{1 \times 0.6 \times 150}{60} = 23.56 \text{ m/s}$$

$$\eta_B = \frac{V_{w1} u_1}{gH} \Rightarrow V_{w1} = \frac{0.84 \times 9.81 \times 68}{23.56} = 26.61 \text{ m/sec.}$$

since $u_1 < V_{w1} \Rightarrow \theta < 90^\circ$



$$\tan \theta = \frac{V_{f1}}{V_{w1} - u_1} \Rightarrow \theta = 60.8^\circ$$

$$\tan \phi = \frac{V_{f2}}{u_2} = \frac{V_{f1}}{u_1/2} \Rightarrow \phi = 24.9^\circ$$

$$\tan \alpha = \frac{V_{f1}}{V_{w1}} \Rightarrow \boxed{\alpha = 11.64^\circ}$$

Sheet 1 of 1
Pavb17

Design a Francis turbine runner with the following data:

Net head 68 m
Speed $N = 750 \text{ r.p.m}$
Output power = 330 kW (S.P.)
 $\eta_B = 94\%$

(237)

η_{cy} (Overall efficiency) = 85% = η_o
 $\psi = \text{Flow ratio} = 0.15 \Rightarrow \psi = \frac{V_{f1}}{\sqrt{2gH}} = 0.15$

width ratio $\eta = b/d = 0.1$

Inner dia. of runner is $\frac{1}{2}$ of outer dia. Also assume 6% of the circumferential area of runner to be occupied by the thickness of vanes. Velocity of flow remains constant & flow is radial at exit.

Given

$K = 0.94$, $d_2 = d_1/2$, $b_1/d_1 = 0.1$, $V_{f1} = V_{f2}$

$$\eta_o = \frac{S.P.}{H.P.} = \frac{330}{H.P.} \Rightarrow H.P. = 388.235 \text{ kW}$$

$$= \omega Q H \quad \left[\omega = 10 \text{ rev/sec} \right]$$

$$\Rightarrow Q = 0.58 \text{ m}^3/\text{sec}$$

$$\frac{V_{f1}}{\sqrt{2gH}} \Rightarrow V_{f1} = 5.48 \text{ m/sec}$$

$$Q = K \cdot (\pi d_1 b_1) V_{f1} = 0.94 \times \pi \times d_1 b_1 \times 5.48$$

$$b_1 d_1 = 0.036 \text{ m}^2 \quad \text{--- (11)}$$

$$b_1/d_1 = 0.1 \quad \text{--- (1)}$$

$$b_1 = 0.1 d_1$$

$$\Rightarrow d_1 \times 0.1 d_1 = 0.036$$

$$\Rightarrow 0.1 d_1^2 = 0.036$$

$$\Rightarrow d_1 = 0.6 \text{ m} = 600 \text{ mm}$$

$$b_1 = 0.06 \text{ m} = 60 \text{ mm}$$

$$Q = 0.94 \pi d_1 b_1 V_{f1} = 0.94 \pi d_2 b_2 V_{f2}$$

$$b_2 = \frac{d_1 b_1 V_{f1}}{d_2 V_{f2}}$$

$$\Rightarrow b_2 = 120 \text{ mm}$$

9/2001

at an avg. head of 160 m with a discharge of $80 \text{ m}^3/\text{sec}$. The inlet and outlet dia. are 4 m & 2 m respectively. The runner blade angle is 120° . Radial discharging velocity at outlet is 15 m/sec . Assuming constant width of wheel and 80% hydraulic efficiency. Determine H.P. produced, in MW. & R.P.M. of machine.

Given

$$H = 160 \text{ m}$$

$$Q = 80 \text{ m}^3/\text{sec}$$

$$d_1 = 4 \text{ m}$$

$$d_2 = 2 \text{ m}$$

$$\theta = 120^\circ$$

$$V_{f2} = V_2 = 15 \text{ m/sec}$$

$$V_{w2} = 0$$

$$b_1 = b_2 = \text{const.}$$

$$\eta_H = 0.8$$

$$\text{H.P.} = ?$$

$$N = ?$$

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$$u_1 = \frac{\pi d_1 N}{60}$$

$$Q = \pi d_2 b_2 V_{f2}$$

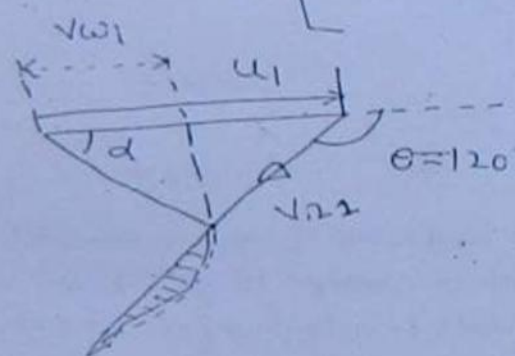
$$80 = \pi \times 2 \times b_2 \times 15$$

$$\Rightarrow b_2 = 0.84 \text{ m}$$

$$b_1 = b_2 = 0.84 \text{ m}$$

$$Q = V_{f1} \cdot \pi d_1 b_1$$

$$\Rightarrow V_{f1} = 7.5 \text{ m/sec}$$



$$\therefore \tan(180^\circ - \theta) = \frac{V_{f1}}{u_1 - V_{w1}}$$

$$\eta_H = \frac{V_{w1} u_1}{g H}$$

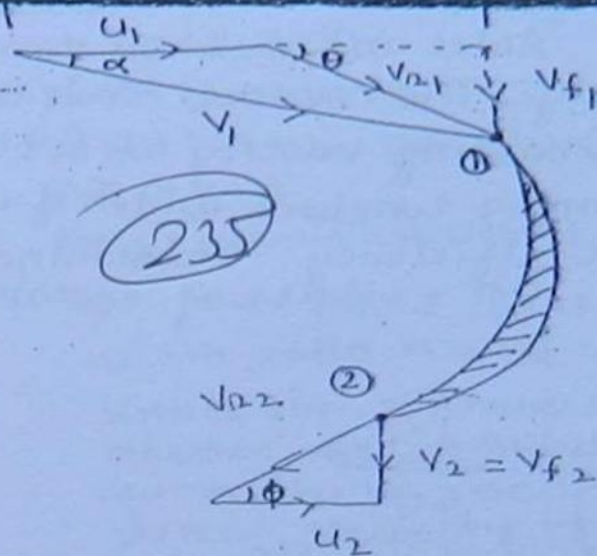
$$\Rightarrow V_{w1} \cdot u_1 = 0.8 \times 9.81 \times 160 \quad \text{--- (i)}$$

By solving eqn (i) & (ii)

$$u_1 \cdot V_{w1} = 39.81 \text{ m/sec} = \frac{\pi d_1 N}{60}$$

$$u_1 = 35.96 \text{ m/sec} \Rightarrow N = 190 \text{ R.P.M.}$$

$$\begin{aligned} \text{Hydraulic Power} &= \rho Q g H \\ &= 9.81 \times 80 \times 160 \text{ kW} \\ \therefore \text{H.P.} &= 125.5 \text{ MW} \end{aligned}$$



(235)

$$\eta_B = \frac{V_{w1} u_1}{gH} \quad \text{--- (iii)}$$

Apply B. eqn at (1) & (2)

$$H = \frac{V_2^2}{2g} + \frac{V_{w1} u_1}{g}$$

$$= \frac{V_2^2}{2g} + \frac{V_{w1} u_1}{g}$$

$$\frac{V_{w1} u_1}{g \eta_B} = \frac{V_2^2}{2g} + \frac{V_{w1} u_1}{g}$$

$$\Rightarrow \frac{1}{\eta_B} = \frac{V_2^2}{2 u_1 V_{w1}} + 1$$

$$\eta_B = \frac{1}{1 + \frac{V_2^2}{2 u_1 V_{w1}}}$$

$$= \frac{1}{1 + \frac{V_{w1}^2 \tan^2 \alpha / 2}{V_{w1} \times V_{w1} \left[1 - \frac{\tan \alpha}{\tan \theta} \right]}}$$

$$\therefore \eta_B = \frac{1}{1 + \left[\frac{\tan^2 \alpha / 2}{1 - \frac{\tan \alpha}{\tan \theta}} \right]}$$

$$\tan \alpha = \frac{V_{f1}}{V_{w1}}$$

$$\Rightarrow V_{f1} = V_{w1} \tan \alpha$$

$$V_{2,} = V_{f2} = V_{f1} = V_{w1} \tan \alpha$$

$$\tan \theta = \frac{V_{f1}}{V_{w1} - u_1}$$

$$\therefore \tan \theta = \frac{V_{w1} \tan \alpha}{V_{w1} - u_1}$$

$$\therefore V_{w1} - u_1 = V_{w1} \frac{\tan \alpha}{\tan \theta}$$

$$\therefore u_1 = V_{w1} \left[1 - \frac{\tan \alpha}{\tan \theta} \right]$$

$$Q = \pi d_1 b_1 \cdot V_{f1}$$

$$= \pi \times 0.6 \times 0.15 \times 1.5$$

$$= 0.424 \text{ m}^3/\text{sec}$$

$$= \pi d_2 b_2 \cdot V_{f2}$$

$$\Rightarrow b_2 = 0.3 \text{ m} = 300 \text{ mm}$$

$$\sin \theta = \frac{V_{f1}}{V_{r1}} \Rightarrow V_{r1} = \frac{V_{f1}}{\cos \theta} = 2.89 \text{ m/sec}$$

$$R.P. = \frac{\omega G}{g} [V_{w1} u_1 - V_{w2} u_2]$$

$$= 21.74 \text{ kW}$$

• Apply Master Formula { Apply B. eqn b/w ① & ② }

$$H = \frac{V_2^2}{2g} + \frac{V_{w1} u_1}{g} = 5.34 \text{ m}$$

Assuming
minor losses
to be neglected

$$\eta_H = \frac{V_{w1} u_1}{g H} = 95.1\%$$

Prob. 4
CS/2003

An inward flow reaction turbine discharges at the velocity of flow through the runner is such that hydraulic efficiency is given by

$$\eta_H = \frac{1}{1 + \frac{V_2 \tan^2 \alpha}{1 - \left(\frac{\tan \alpha}{\tan \theta} \right)}}$$

where α is the guide blade angle & θ is runner vane angle at inlet

Assume there is no friction on the blades

19 An inward reaction flow turbine runs at 192 r.p.m. the dia & width at inlet are 600 mm & 150 mm & the outlet dia. is 300 mm. The velocity of flow to the runner is constant at 1.5 M/sec. If the guide blades are 10° to the wheel tangent. Draw the inlet & outlet velocity diagram, if velocity of whirl at outlet is zero. Determine

- Runner Blade Angle
- Abs. Velocity of water leaving the guide blade
- Rel. velocity of water at inlet
- Width of the wheel at outlet
- Discharge through turbine
- Head supplied
- R.P. supplied (developed)
- Hydraulic Efficiency.

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Given $N = 192 \text{ r.p.m.}$
 $d_1 = 600 \text{ mm}$
 $b_1 = 150 \text{ mm}$
 $d_2 = 300 \text{ mm}$
 $V_{f1} = V_{f2} = 1.5 \text{ M/sec}$
 $\alpha = 10^\circ$
 $V_{w2} = 0$
 $\phi \text{ \& } \psi = ?$

$$u_1 = \frac{\pi d_1 N}{60} = 6.03 \text{ M/sec}$$

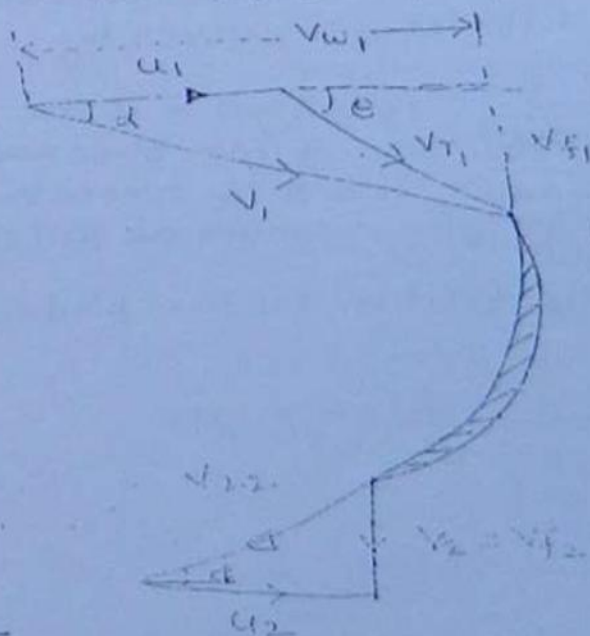
$$\tan \psi = \frac{V_{f1}}{V_{w1}}$$

$$\Rightarrow V_{w1} = 8.507 \text{ M/sec}$$

$$V_1 = \sqrt{V_{f1}^2 + V_{w1}^2}$$

$$= 8.63 \text{ M/sec}$$

Since $u_1 < V_{w1} \Rightarrow \psi < 90^\circ$



$$\tan \psi = \frac{V_{f1}}{V_{w1} - u_1}$$

$$\Rightarrow \psi = 31.19^\circ$$

$$\frac{u_1}{u_2} = \frac{d_1}{d_2}$$

$$\Rightarrow u_2 = u_1 \times \frac{d_2}{d_1} = 3.015 \text{ M/sec}$$

$$\tan \phi = \frac{V_{f2}}{u_2}$$

$$\Rightarrow \phi = 26.45^\circ$$

→ $N = 450 \text{ R.P.M.}$

$H = 120 \text{ M}$

$d_1 = 1.2 \text{ M}$

$A_{f1} = 0.4 \text{ M}^2$

$\alpha = 20^\circ$

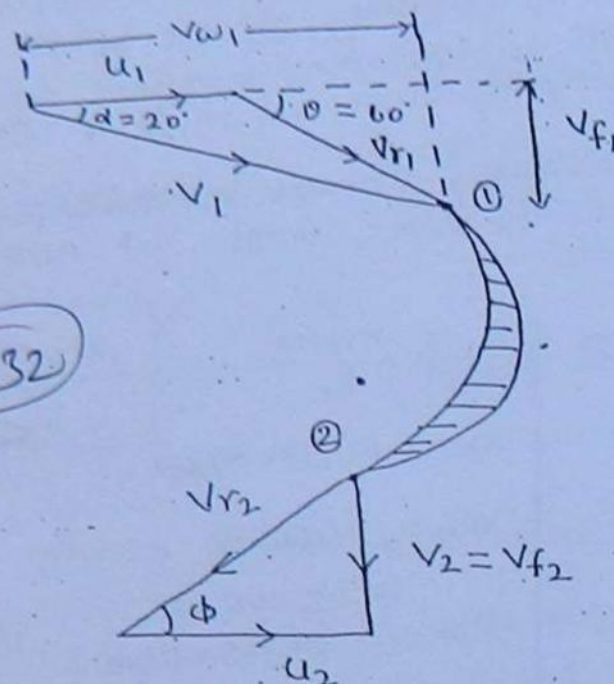
$\theta = 60^\circ$

$V_{w2} = 0$

$Q = ?$

$R.P. = ?$

$\eta_B = ?$



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$$u_1 = \frac{\pi d_1 N}{60} = 28.27 \text{ M/sec.}$$

$$\tan \theta = \frac{V_{f1}}{V_{w1} - u_1} \quad \text{--- (1)}$$

$$\tan \alpha = \frac{V_{f1}}{V_{w1}} \quad \text{--- (2)}$$

By eqn (1) & (2)

$$V_{w1} = 35.78 \text{ M/sec}$$

$$V_{f1} = 13.03 \text{ M/sec.}$$

$$Q = A_{f1} \cdot V_{f1}$$

$$= 0.4 \times 13.03$$

$$Q = 5.21 \text{ m}^3/\text{sec}$$

$$\eta_B = \frac{V_{w1} u_1 - V_{w2} u_2}{gH} = 85.96\%$$

$$\text{Runner power} = \frac{\omega Q}{g} [V_{w1} u_1 - V_{w2} u_2]$$

$$= 5270 \text{ kW}$$

have an external diameter of 700mm & a width of 180mm. If the guide vane are at 20° to the wheel tangent and the abs. velocity of water at inlet is 25m/sec. Then find

- Discharge through the turbine
- Runner Vane angle at inlet

$$N = 500 \text{ r.p.m.}$$

$$d_1 = 700 \text{ mm} \\ = 0.7 \text{ m}$$

$$b_1 = 180 \text{ mm} \\ = 0.18 \text{ m}$$

$$\alpha = 20^\circ$$

$$V_1 = 25 \text{ m/sec.}$$

$$\theta = ?$$

$$\phi = ?$$

$$u_1 = \frac{\pi d_1 N}{60}$$

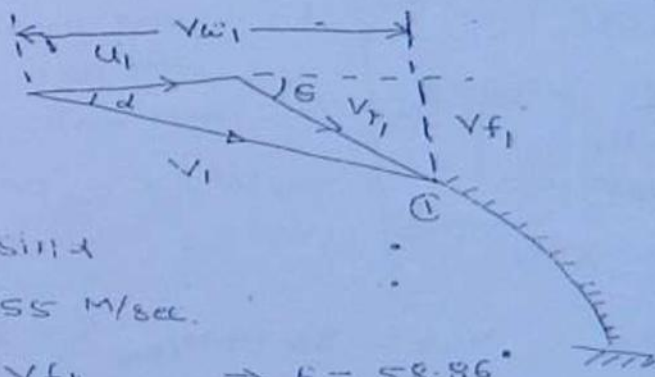
$$u_1 = 18.326 \text{ m/sec}$$

$$V_{w1} = V_1 \cos \alpha$$

$$= 25 \cos 20^\circ$$

$$= 23.49 \text{ m/sec}$$

$$\text{since } u_1 < V_{w1} \Rightarrow \theta < 90^\circ$$



$$V_{f1} = V_1 \sin \alpha$$

$$= 8.55 \text{ m/sec.}$$

$$\tan \theta = \frac{V_{f1}}{V_{w1} - u_1} \Rightarrow \theta = 58.86^\circ$$

$$Q = A_{f1} \cdot V_{f1}$$

$$= (\pi d_1 b_1) V_{f1}$$

$$= \pi \times 0.7 \times 0.18 \times 8.55$$

$$= 3.354 \text{ m}^3/\text{sec}$$

Prob. 2

A reaction turbine works at 450 r.p.m. under a head of 120 M. If the dia. at inlet is 1.2 m & flow area at inlet is 0.4 m^2 . The angle made absolute and relative velocities at the inlet with the tangent at wheel is 60° & 20° respectively. Then determine

- Flow rate
- Runner power developed
- Hydraulic efficiency

$$= \frac{\left(\frac{\omega Q}{g}\right) [V_{w1} u_1 - V_{w2} u_2]}{\omega Q} + \frac{V_2^2}{2g} + h_L$$

If $V_{w2} = 0$ & Losses are Negligible
than

[pressure
energy
Negligible]

$$H = \frac{V_2^2}{2g} + \frac{V_{w1} u_1}{g} \quad \text{--- 230 --- Master Form}$$

(d) If speed ratio is given than it means

$$\phi = \frac{u_1}{\sqrt{2gH}}$$

[it is only significant
for inlet port
 ϕ lies between 0.5 and 1]

(e) If flow ratio is given than

$$\psi = \frac{V_{f1}}{\sqrt{2gH}}$$

[Also valid at inlet
 ψ lies b/w 0.15 and 0.5]

(f) If width ratio is given than

$$= b/d \quad \left\{ \frac{b_1}{d_1} \right\}$$

$$\eta_B = \frac{V w_1 u_1}{g H}$$

2) Mechanical efficiency

(229)

$$\eta_M = \frac{S.P.}{R.P.}$$

3) overall efficiency

$$\eta_o = \frac{S.P.}{H.P.} = \frac{S.P.}{R.P.} \times \frac{R.P.}{H.P.} = \eta_B \times \eta_M$$

$$\therefore \eta_o = \eta_B \times \eta_M$$

NOTE:

If Leakage in the turbine chamber are also considered than Volumetric efficiency may be considered as

$$\eta_v = \frac{Q'}{Q}$$

$Q' \rightarrow$ Discharge at exit
 $Q \rightarrow$ Discharge at inlet

However this is more imp. for centrifugal pump & Negligible for turbines.

If it is also accounted than overall efficiency

$$\eta_o = \eta_B \times \eta_M \times \eta_v$$

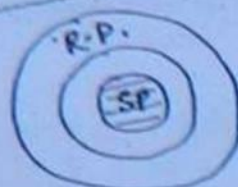
NOTE:

(a) If it is given that velocity through the runner is constant then $V_{f1} = V_{f2}$ and hence $A_{f1} = A_{f2}$

(b) In case of some data is missing than Bernoulli eqn may be applied b/w inlet & outlet of the runner by assuming No-Losses of head through the runner.

(c) Total Head at inlet = work done/sec/unit weight of water
+ Energy head at outlet + Losses

H.P. or W.P.



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a) Hydraulic Power (water power): It is the power available at the inlet of turbine.

$$H.P. (W.P.) = \rho g \cdot Q \cdot H \rightarrow \text{KW} \quad \left[\rho = 9.81 \text{ KN/m}^3 \right]$$

$$= \frac{\rho Q H}{75} \rightarrow \text{H.P.} \quad \{ \text{Horse Power} \}$$

b) Runner Power = work done by water on Runner

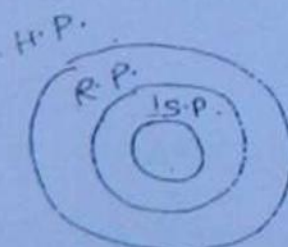
$$= \frac{\omega Q}{g} [V_{w1} u_1 - V_{w2} u_2]$$

c) Shaft-Power (Break Power): This is final power available at the shaft of turbine.

$$\therefore S.P. = R.P. - \text{Transmission Losses} \quad \{ \text{Mechanical Losses} \}$$

7) Efficiency of Turbines: ↓

(a)



$$\text{Hydraulic efficiency} = \eta_H = \frac{R.P.}{\text{water Power}}$$

$$\therefore \eta_H = \frac{\left(\frac{\omega Q}{g} \right) [V_{w1} u_1 - V_{w2} u_2]}{\omega Q H}$$

$$\therefore \eta_H = \frac{V_{w1} u_1 - V_{w2} u_2}{g H}$$

Ang. Momentum = mvr

$$\therefore \text{Ang. Momentum/sec} = \left(\frac{\text{mass}}{\text{sec}} \right) \cdot V \cdot r$$

$$\therefore T = \left(\text{mass/sec} \right) V_{w1} r_1 - \left(\frac{\text{mass}}{\text{sec}} \right) V_{w2} r_2$$

$$\therefore T = \rho Q [V_{w1} r_1 - V_{w2} r_2] \quad (22)$$

$$\begin{aligned} \therefore \text{work-done/sec.} &= T \times \omega \\ &= \rho Q [V_{w1} r_1 \omega - V_{w2} r_2 \omega] \\ &= \rho Q [V_{w1} u_1 - V_{w2} u_2] \end{aligned}$$

$$\therefore \boxed{\text{work-done/sec.} = \left(\frac{\omega}{g} \right) Q [V_{w1} u_1 - V_{w2} u_2]}$$

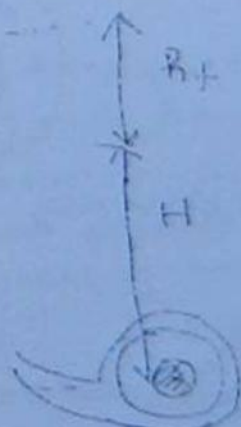
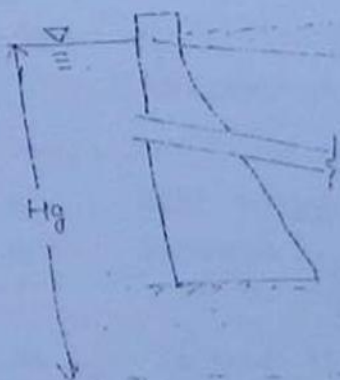
$[\omega \rightarrow \text{units of } \omega]$

For Francis Turbine

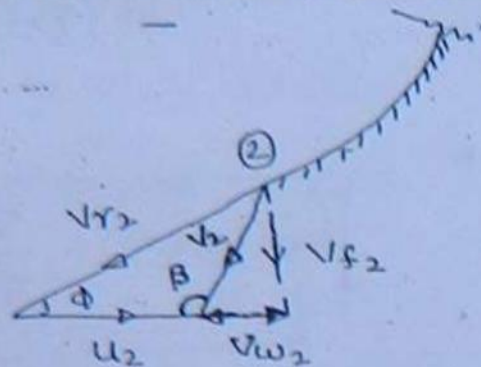
$$V_{w2} = 0$$

$$\begin{aligned} \therefore \text{work-done/sec.} &= \text{R.P. (Runner Power)} \\ &= \rho Q [V_{w1} u_1] \\ &= \frac{\omega Q}{g} [V_{w1} u_1] \end{aligned}$$

6) Power of Turbine



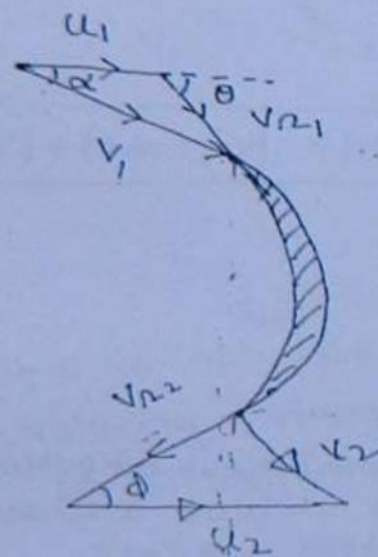
$H = \text{Net Head at inlet of turbine (Runner)} = H_g$



$$\tan \phi = \frac{V_{f2}}{|u_2| + |V_{w2}|}$$

$$|V_{r2} \cos \phi| = |u_2| + |V_{w2}|$$

5> Work-done per sec by the water on Runner
(Runner Power)

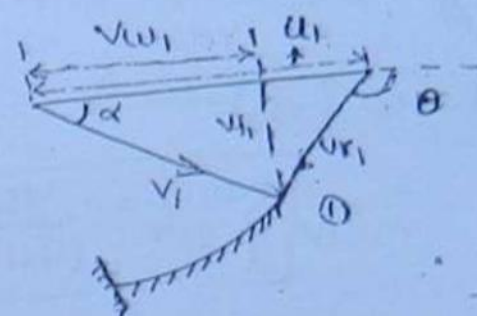


Rate of change of Angular momentum = T

$$\text{Work-done/sec} = T \times \omega$$

$$T = \frac{\text{Angular momentum/sec at inlet} - \text{Angular momentum/sec at outlet}}{\text{sec}}$$

case 3 when $\theta > 90^\circ$

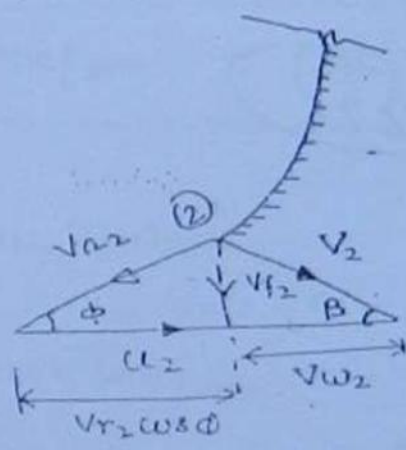


$$\tan(180 - \theta) = \frac{V_{f1}}{u_1 - V_{w1}}$$

(225)

4) velocity triangle at exit

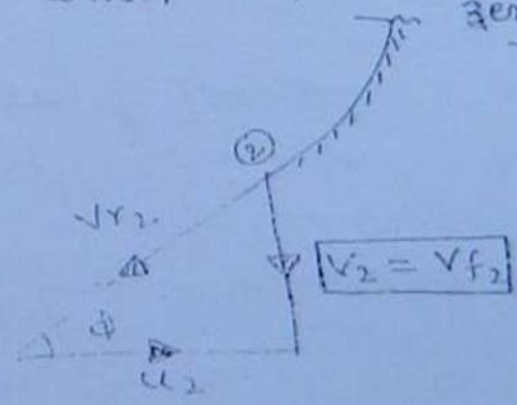
case 4) when V_{w2} is in direction of V_{w1}



$$|u_2| = |V_{r2} \cos \phi| + |V_{w2}|$$

VIMP case B)

when $V_{w2} = 0$, whirl component at outlet is zero or turbine discharges radially outward.



$$V_2 = V_{f2}$$

- $\beta = 90^\circ, V_{w2} = 0$
- $|u_2| = |V_{r2} \cos \phi|$
- $\tan \phi = \frac{V_{f2}}{u_2}$

common case of Francis Turbine.

$$u_1 = \frac{\pi d_1 N}{60} = r_1 \omega$$

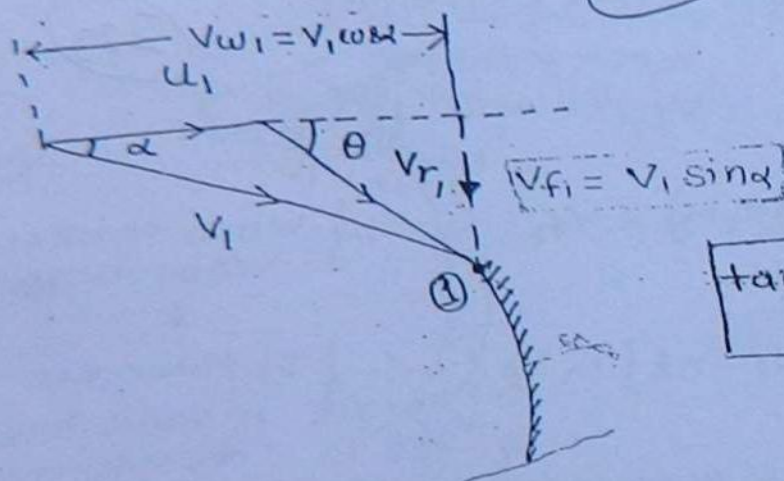
$$u_2 = \frac{\pi d_2 N}{60} = r_2 \omega$$

$$\therefore \boxed{u_1/u_2 = d_1/d_2}$$

(3) velocity triangle at inlet.

case A) when $\theta < 90^\circ$

(224)



$$\boxed{\tan \theta = \frac{V_{f1}}{V_{w1} - u_1}}$$

NOTE:

if θ is not given then in order to obtain velocity triangle compare magnitudes of u_1 & V_{w1} . if

$$u_1 < V_{w1} \Rightarrow \theta < 90^\circ$$

$$\text{if } u_1 = V_{w1} \Rightarrow \theta = 90^\circ$$

$$\text{if } u_1 > V_{w1} \Rightarrow \theta > 90^\circ$$

case B)

when $\theta = 90^\circ$ means vanes are set radially at inlet

$$\boxed{u_1 = V_{w1}}$$

$$V_{f1} > 0$$

$$0$$

$$V_2 \sin \beta = V_{f2} = \text{Radial (Flow) velocity at exit}$$

NOTE: If β is 90° then Vanes are said to be radial exit or if turbine discharges radially outward then $\beta = 90^\circ$.

In case of Francis Turbine in order to increase the efficiency β is purposely made 90° .

Following points may be noted w.r.t. Francis Turbine

(i) Discharge through Runner (Turbine)

(223)

$$Q = A_{f1} \times V_{f1}$$

$$= (\pi d_1 b_1) \times V_{f1} \quad \left\{ \begin{array}{l} \text{When thickness of vane is negligible} \end{array} \right.$$

$$= [\pi d_1 - n \epsilon] b_1 V_{f1} \quad \left\{ \begin{array}{l} \text{If there are } n \text{ vanes having } \epsilon \text{ thickness each} \end{array} \right.$$

$$= K \cdot \pi d_1 b_1 V_{f1} \quad \left\{ \begin{array}{l} K \text{ is coeff. which is account for reduced area occupied by thickness} \end{array} \right.$$

For eg: 5% area at circumference is occupied by Vane thickness then $K = 0.95$

At exit

$$Q = V_{f2} \times A_{f2}$$

$$= (\pi d_2 b_2) \times V_{f2}$$

$$= [\pi d_2 - n \epsilon] \times b_2 V_{f2}$$

$$= K \pi d_2 b_2 V_{f2}$$

$$Q = \pi d_1 b_1 V_{f1} = \pi d_2 b_2 V_{f2}$$

Blade Angle at Inlet

V_{a1} = Relative velocity at inlet (Rel. vel. of fluid with respect to Blade)

$$|\theta > 90^\circ, = 90^\circ, < 90^\circ|$$

if $\theta = 90^\circ$, Vanes are said to be radial at inlet

α = Guide blade angle or angle b/w u_1 & V_1

α = Angle of V_1 with the tangent of wheel / Runner at inlet [b/w 10 to 30°]

V_1 = Abs. velocity ^{of fluid} at inlet

$V_1 \cos \alpha$ = Tangential component of abs. velocity [wheel velocity at inlet u_{w1}]

This component is responsible for producing torque

Exit Point

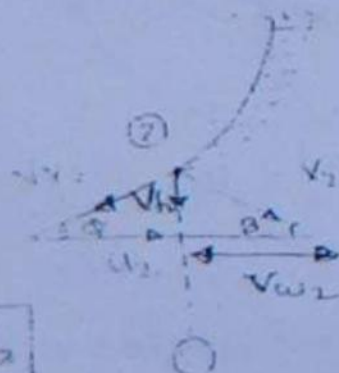
(222)

$$V_1 \sin \alpha = V_{f1}$$

From vector diagram

$$\vec{V}_1 = \vec{u}_1 + \vec{V}_{r1}$$

$V_1 \sin \alpha$ = Radial component of V_1
(Radial velocity at inlet)
(velocity of flow at inlet) [V_{f1}]



$$\vec{V}_{r2} + \vec{u}_2 = \vec{V}_2$$

V_{r2} = Relative velocity at exit which is in the direction of vane. In a turbine V_{r1} is not necessarily to be equal to V_{r2}

u_2 = Tangential velocity of the Runner at exit

V_2 = Abs. Velocity at exit

$\rightarrow |\beta \text{ may be } > 90^\circ, = 90^\circ, < 90^\circ|$

Noting the runner, transfer energy to shaft & fluid
 Shaft may be connected to generator.

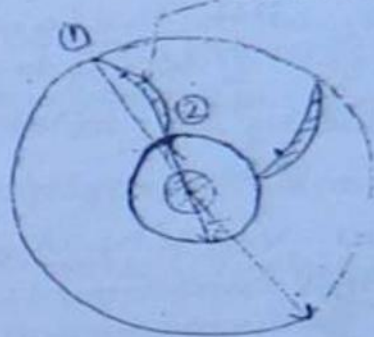
29/10/20

1) FRANCIS TURBINE: ↓

Runner ↓

(24)

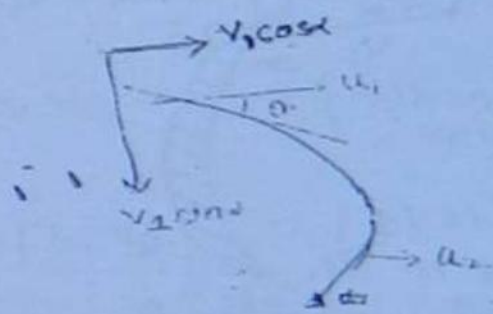
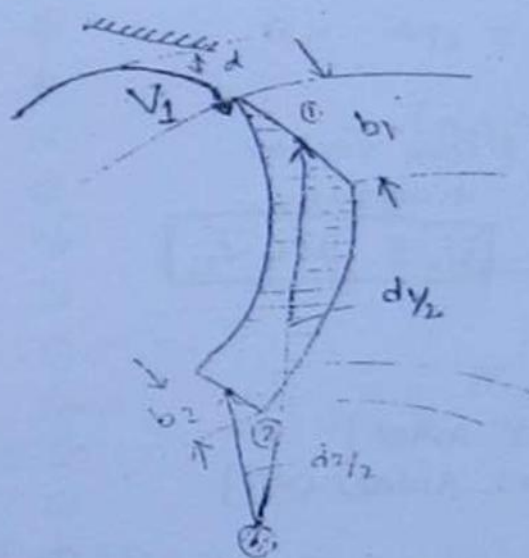
Runner Vane



① → Entry point
 ② → Exit point

Inner dia. = Dia. at
 = d_2

Outer diameter
 = Dia. at Inlet
 = d_1



Velocity triangle: ↓

Guide Blade (Fixed in the casing)

Tangential direction
 (direction of u_1)

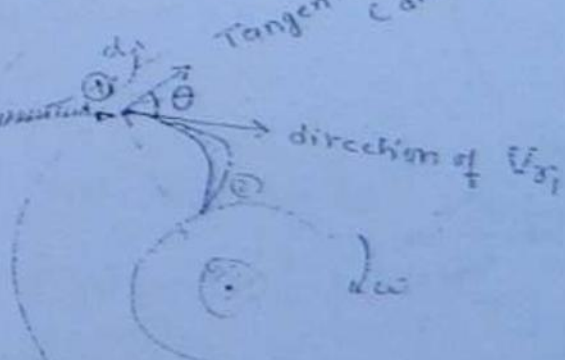
u = Tangential Velocity /
 Peripheral velocity
 of Runner at
 inlet

$$= r_1 \omega$$

$$= d_1 \omega$$

$$= d_1 \left(\frac{2\pi N}{60} \right)$$

$$u_1 = \frac{\pi d_1 N}{60} \text{ m/sec.}$$



energy is used to convert into work. Pressure at exit of turbine may fall below atmospheric therefore disposal of exit water directly into atmosphere is not safe hence a tube of gradually diverging ^{exit section} is used to carry water of turbine to tail rest & this is always submerged at some depth below tail rest level.

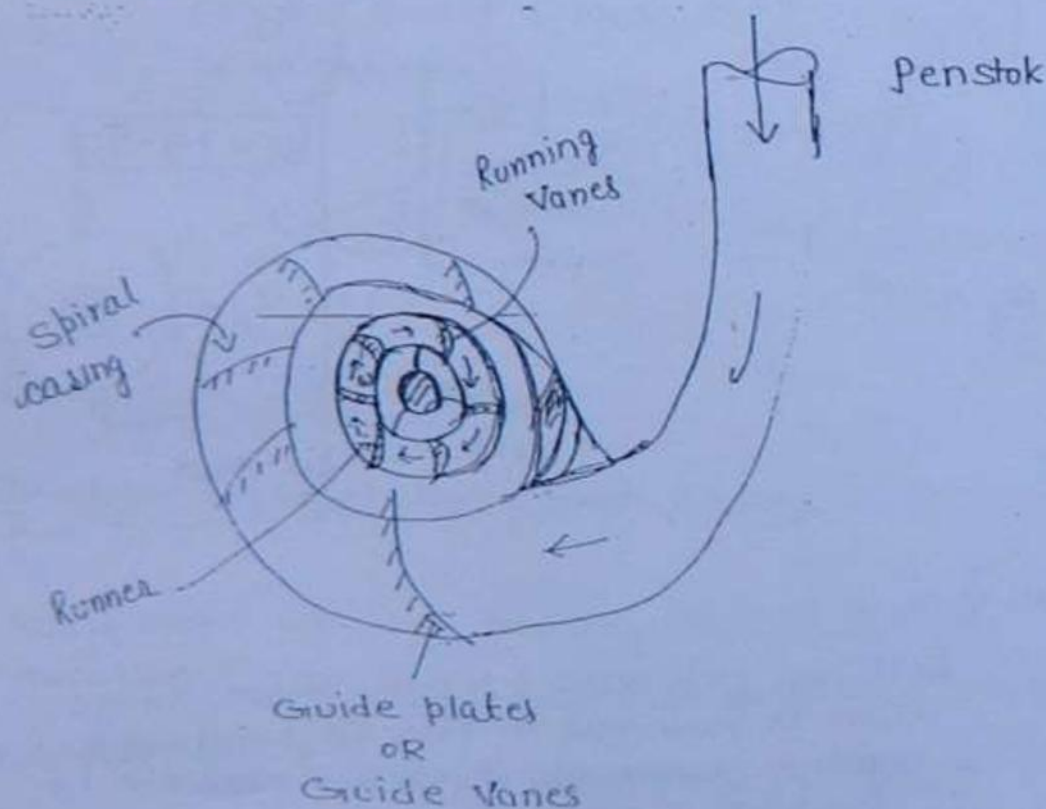
2. In case of impulse turbine, pressure remains const. hence draft tube is not essential.

d) Turbine units: ↓

1) Francis Turbine: ↓

(220)

* Inward Radial Flow Reaction turbine



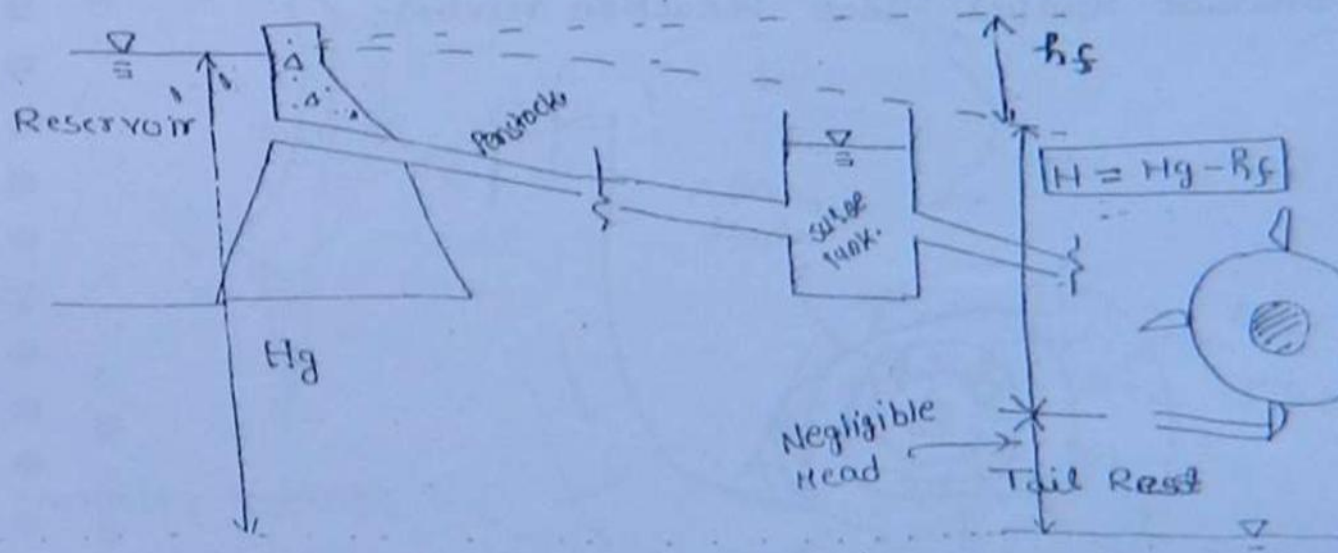
- Casing is a spiral chamber which is of gradually decreasing area in order to keep constant velocity at inlet of vanes. [As decrease, so A is decreased] somewhat entered into runner
- Guide plates are permanently attached to casing which

~~classification on the basis of which is the practical~~
 is combination of Last two classification.

- i) Francis Turbines \rightarrow Inward Radial flow Reaction turbine.
- ii) Pelton wheel / Turgo impulse wheel \rightarrow Tangential flow impulse turbine.
- iii) Kaplan / propeller \rightarrow Axial flow Reaction turbine.
- iv) Modern Francis \rightarrow Mixed Flow Reaction turbine.

Important units of hydropower plant: \downarrow

(2/9)



a) Surge Tank: \downarrow

Between the reservoir & turbine house, surge tank is provided in order to minimize water hammer pressure problem in penstock. Surge tank also helps in maintaining constant head at turbine.

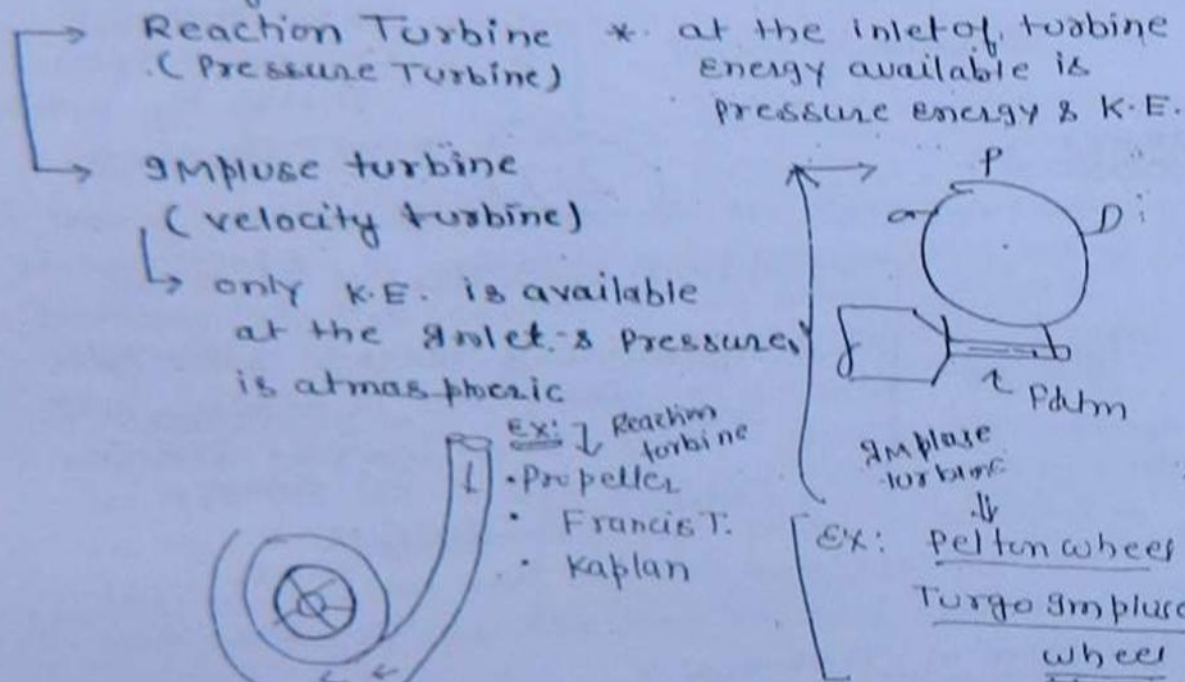
b) Penstock: \downarrow

It is the pipe through which water is brought from reservoir or from surge tank to the turbine chamber. There is always head loss at the turbine.

- High $[V_1 = 300 \text{ to } 1000]$
ex: Kaplan & propeller
- Medium sp-speed $[V_2 = 60 \text{ to } 300]$
- Low sp-speed $[V_3 < 60 \text{ m/sec}]$
ex: Pelton wheel

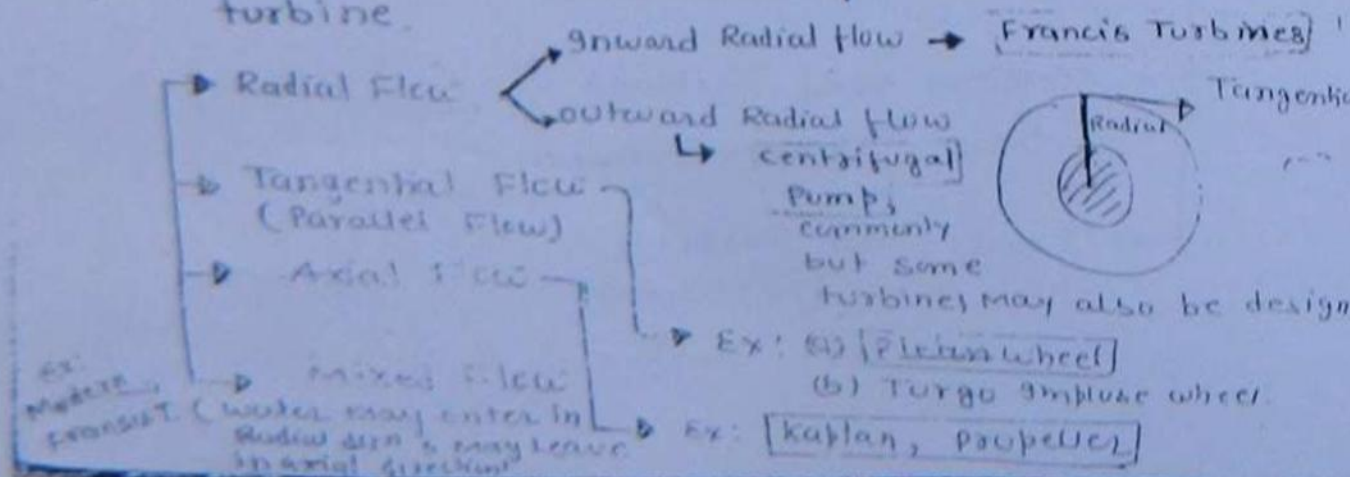
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c) classification on the basis of available energy at the inlet of turbines



• Reaction turbines have closed casing

d) classification on the basis of type of flow in the turbine



Turbines

* used to convert hydraulic energy into mechanical energy which is further converted into electric energy by means of generators.

* It extracts energy

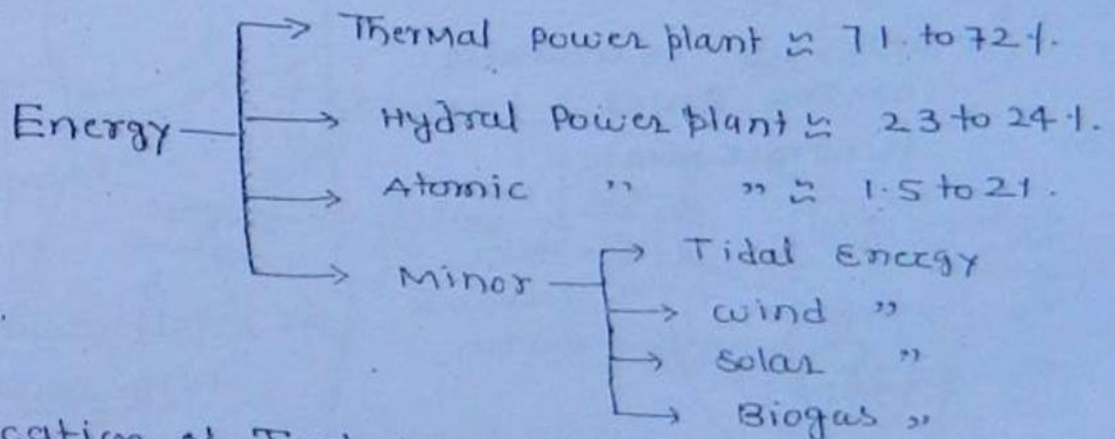
Pumps

* First electric energy is converted into mechanical energy by motor & mechanical energy is converted into pressure energy or hydraulic energy by pumps.

* It adds energy.

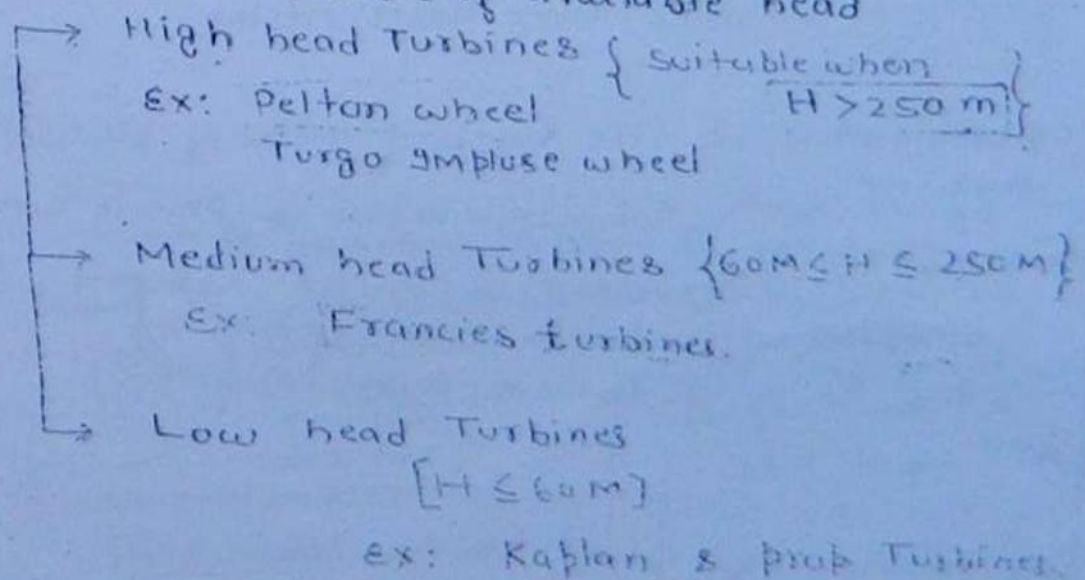
Turbines: ↓

(217)



classification of Turbines: ↓

a) classification on the basis of available head



$$[\cos \theta] = \frac{w}{2}$$

$$\therefore (\rho a v^2 \cos \theta) (x / \cos \theta) = w (L/2 \sin \theta) \quad \rightarrow \text{wt. of plate}$$

$$\therefore \sin \theta = \frac{2 \rho a v^2 x}{w L}$$

special case:

When $x = L/2$ [Jet strikes on the c.g.]

$$\therefore \sin \theta = \frac{\rho a v^2}{w}$$

(216)

Prob. 1

A Jet of water of dia. 25 mm dia strikes to a 20 cm square plate of uniform thickness with a vel. of 10 m/sec at the centre of the plate which is suspended vertically by a hinge on its top edge. The weight of the plate is 98 N/m. The Jet strikes normal to the plate. What force must be applied at lower edge of plate so that plate is kept vertical. If the plate is allowed to deflect ^{freely} what will be the angle of def'n with due to the force exerted by the Jet of water.

Ans: $\theta = 30^\circ$, $F = 28.5 \text{ N}$

Efficiency of Jet

$$\eta = \frac{\text{work-done/sec}}{\text{K.E./sec}}$$

$$\therefore \eta = \frac{\rho a v [v-u] \cdot u}{\frac{1}{2} \rho a v^3} = \frac{2 [vu - u^2]}{v^2}$$

For Maxm. Efficiency of Jet, $\frac{d\eta}{du} = 0$ (215)

$$\therefore \frac{2}{v^2} [v - 2u] = 0$$

\Rightarrow

$$u = \frac{v}{2}$$

tangential = $\frac{1}{2}$ J vel.

$$\therefore \eta_{\text{Max}} = \frac{2 \left[v \times \frac{v}{2} - \left(\frac{v}{2}\right)^2 \right]}{v^2} = 0.5$$

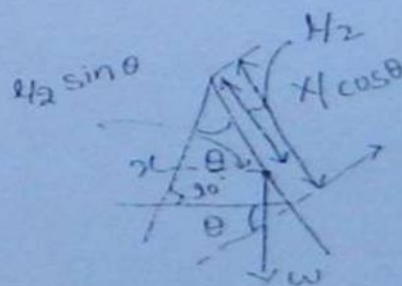
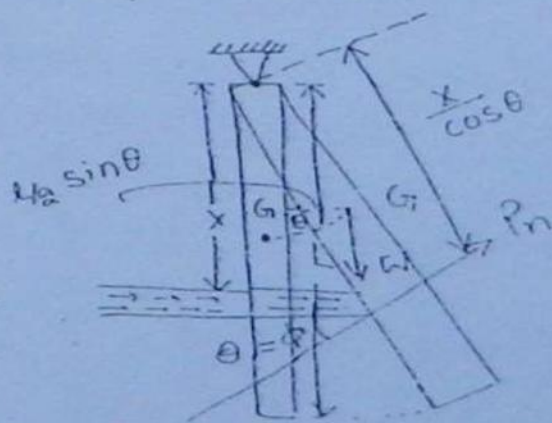
$$= 50\% \quad \text{qmb}$$

NOTE:

In above analysis vanes are flat & if vanes are made curved than efficiency may be further increased as in case of Pelton.

Case VIIIth:

Jet of water strikes normal to a hinged vertical plate



$$P_n = \rho g [v \cos \theta - 0]$$

$$P_n = \rho a v^2 \cos \theta$$

$a \rightarrow$ area of jet

$$= \rho Q$$

$$= \rho \cdot a [v-u]$$

Force exerted by the jet on the plate

$$P_n = \rho \cdot a (v-u) [v-u]$$

$$P_n = \rho a (v-u)^2$$

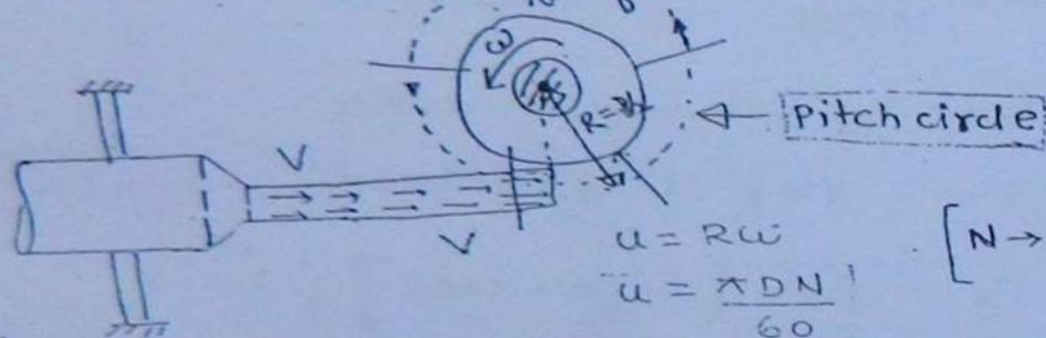
∴ Work-done per sec. by the jet

$$W = \rho a (v-u)^2 \times u$$

(214)

ase VIth;

Jet strikes on a series of ~~fixed~~ Vanes Mounted on a periphery of a wheel



Mass of water strikes the ~~fixed~~ ^{Vane} per sec.

$$= \rho a v = \rho Q$$

$$[Q = av]$$

Force exerted by Jet on Vane will be

$$= \rho a v [v-u]$$

$$= \rho a [v^2 - vu]$$

Work-done by Jet per sec. = $\rho a [v^2 - vu] \times u$

$$= \rho a [v^2 u - u^2 v]$$

$$W = \rho Q [v-u] \times u$$

$$K.E / \text{sec of Jet} = \frac{1}{2} \times \text{Mass Flowing / sec} \times v^2$$

$$K.E = \frac{1}{2} \rho a v^3$$

Force exerted along the plate = 0

$$F = \rho Q V \cos \theta - \rho Q_1 V - (\rho Q_2 (-V)) = 0$$

$$\Rightarrow Q \cos \theta - Q_1 + Q_2 = 0 \quad \text{--- (i)}$$

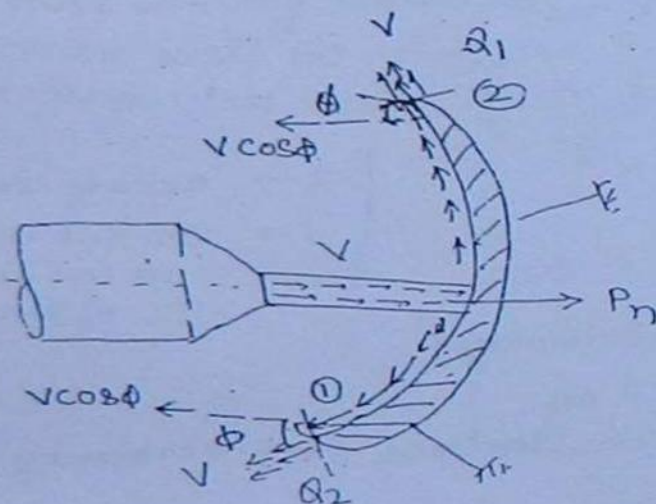
By eqn (i) & (ii)

$$\begin{aligned} Q_1 &= Q/2 [1 + \cos \theta] \\ Q_2 &= Q/2 (1 - \cos \theta) \end{aligned}$$

$$\therefore \frac{Q_1}{Q_2} = \frac{1 + \cos \theta}{1 - \cos \theta}$$

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case IIIrd: Force exerted by Jet on a stationary curve



Force exerted by Jet normal to the plate

$$P_n = \rho Q V - [\rho Q_1 (-V \cos \phi) + \rho Q_2 (-V \cos \phi)]$$

$$= \rho Q V + \rho Q V \cos \phi$$

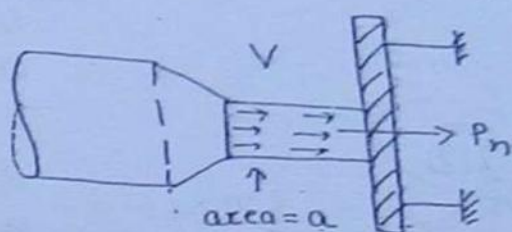
$$P_{n1} = \rho Q V (1 + \cos \phi)$$

$$P_n = \rho a v^2 (1 + \cos \phi)$$

$$\text{work done} = 0$$

1. > Impact of JETS: ↓

Case Ist: when Jet of water strikes normally to a stationary Flat plate



(211)

If Loss of energy due to impact is negligible & surface is smooth so that friction loss is negligible, Force exerted by the jet on the plate is

$$P_n = \rho Q [v - 0]$$

$$P_n = \rho a v^2$$

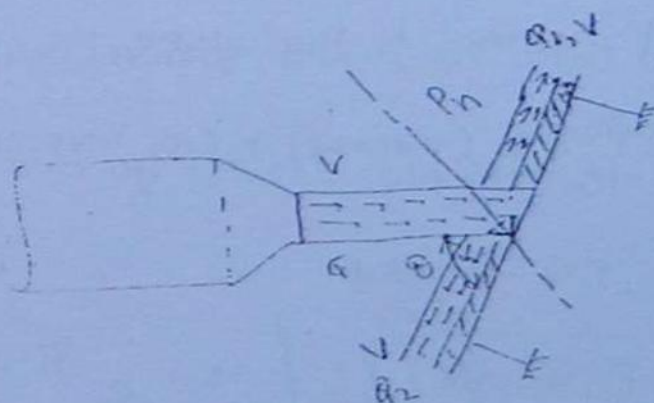
Jet on

work-done by the plate = zero.

[bcz plate remains stationary]

$a \rightarrow$ area of jet
 $v \rightarrow$ initial velocity in the direction of flow

Case IInd: Jet strikes on an Inclined Flat stationary plate



P_n = Force exerted by jet normal to plate

$$= \rho a [v \sin \theta]$$

$$P_n = \rho a v^2 \sin \theta, \text{ [work-done} = 0]$$

$$y_{c2} = \left[\frac{Q^2}{g} \right]^{1/3} = \left[\frac{(6)^2}{9.81} \right]^{1/3} = 1.542 \text{ m} \times \times$$

$$E_{c2} = +5 y_{c2} = 1.5 \times 1.542 = 2.3136 \text{ m}$$

$$E_1' = E_{c2} = 2.3136 \text{ with new u/c depth of } y_1'$$

$$\text{such that } q_1 = y_1' v_1' = 15/3.5$$

$$= 4.2857 \text{ m}^2/\text{s}$$

$$E_1' + \frac{V_1'^2}{2g} = 2.3136$$

$$E_1' + \frac{(4.2857)^2}{2 \times 9.81 \times y_1'} = 2.3136$$

(210)

$$y_1' = 2.122 \text{ m}$$

NOTE:

For the same discharge when $B_2 < B_{2m}$ (under choking conditions) the depth at critical section will be different from y_{cm} and depends upon value of B_2 .

of $15.0 \text{ m}^3/\text{sec}$ at a depth of 2.0 m . It is proposed to reduce the width of the channel at a hydraulic structure. Assuming the transition to be horizontal and flow to be frictionless determine the water surface elevation y_2 and D/S of the constriction when the constricted width is 2.50 m and 2.20 m .

Step 1 check flow (subcritical or supercritical)

$$F_1 = \frac{V_1}{\sqrt{g y_1}} \quad \left[V_1 = \frac{Q}{B_1 y_1} = \frac{15.0}{3.5 \times 2} = 2.143 \text{ m/sec} \right]$$

$$F_1 = \frac{2.143}{\sqrt{9.81 \times 2}} = 0.484 < 1$$

(209)

The U/S flow is subcritical and the transition will cause a drop in the water surface.

Step 2

$$E_1 = y_1 + \frac{V_1^2}{2g} = 2.0 + \frac{(2.143)^2}{2 \times 9.81} = 2.234 \text{ m}$$

Let B_{2m} = minimum width at section (2) which does not cause choking

$$\text{then } E_{2m} = E_1 = 2.234 \text{ m}$$

$$y_{2m} = \frac{2}{3} E_{2m} = \frac{2}{3} \times 2.234 = 1.489 \text{ m}$$

$$\text{But } y_{2m}^3 = \left[\frac{Q^2}{g B_{2m}^3} \right]$$

$$\Rightarrow B_{2m} = \left[\frac{Q^2}{g y_{2m}^3} \right]^{1/3} = 2.636 \text{ m}$$

Step 3

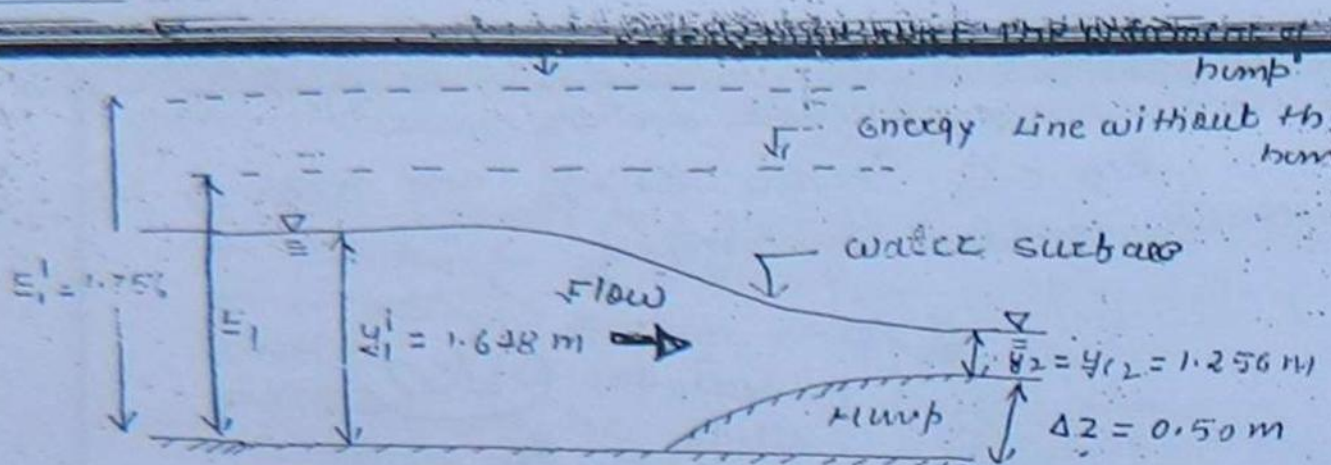
$$\text{Since } B_2 = 2.50 < B_{2m}$$

Hence choking conditions would prevail

$$\therefore \text{The depth at section (2) } y_{c2} = y_2$$

$\therefore U/S$ depth y_1 will increase to y_1'

$$y_2 = \frac{Q}{B_2 V_2} = \frac{15.0}{2.5} = 6.0 \text{ m/sec}$$



Prob 2

A Rect. channel 2.5 m wide carries 6.0 m³/s of flow at a depth of 0.50 m. calculate the height of a flat topped hump required to be placed at a section to cause critical flow. The energy loss due to the obstruction by the hump can be taken as 0.1 times the U/s velocity head.

step 1)

$$q = 6.0 / 2.5 = 2.4 \text{ m}^2/\text{sec}/\text{m}$$

$$V_1 = 2.4 / 0.5 = 4.8 \text{ m/sec}$$

$$Fr_1 = \frac{4.8}{\sqrt{9.81 \times 0.5}} = 2.167 > 1 \Rightarrow \text{U/s Flow is supercritical}$$

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step 2)

$$E_1 = 0.50 + \frac{(4.8)^2}{2 \times 9.81} = 1.674 \text{ m}$$

step 3)

since Flow at section (2) is critical

$$y_c = \left(\frac{q^2}{g} \right)^{1/3} = \left[\frac{(2.4)^2}{9.81} \right]^{1/3} = 0.837 = y_2$$

$$\text{Now } \frac{V_c^2}{2g} = \frac{y_c}{2} = 0.419 = \frac{V_2^2}{2g}$$

∴ Applying Energy Equation b/w section (1) and (2)

$$E_1 - E_L = y_2 + \left(\frac{V_2^2}{2g} \right) + \Delta z$$

↳ height of hump

$$E_L = 0.1 \frac{V_1^2}{2g} = 0.117 \text{ m}$$

$$\Rightarrow 1.674 - 0.117 = 0.837 + 0.419 + \Delta z$$

$$\Rightarrow \boxed{\Delta z = 0.501 \text{ m}}$$

Since $E_{c2} < E_2$ hence $y_2 > y_c$ and depth y_1

available energy at that section

i.e. $E_{c2} < E_2$ hence $y_2 > y_c$ and depth y_1 will be remain unchanged.

step 4) calculation of depth y_2

(207)

$$E_2 = y_2 + \frac{V_2^2}{2g} \quad \left[V_2 = \frac{Q}{y_2} \right]$$

$$E_2 = y_2 + \frac{Q^2}{2gy_2^2}$$

By trial & error

$$1.615 = y_2 + \frac{(2.4)^2}{2 \times 9.81 \times y_2^2} \Rightarrow \boxed{y_2 = 1.481 \text{ m}}$$

(b) At the height of the hump is 0.5 m. Estimate the water surface elevation on the hump and at a section of the hump.

$$V_1 = 1.50 \text{ m/sec}$$

$$F_1 = 0.378 < 1$$

$$E_1 = 1.715 \text{ m}$$

$$y_2 = y_{c2} = 0.837 \text{ m}$$

Available sp. energy at section

$$\Rightarrow E_2 = E_1 - \Delta Z$$

$$E_2 = 1.715 - 0.5 = 1.215 \text{ m}$$

$$E_{c2} = 1.5 y_{c2} = 1.256 \text{ m}$$

→ Since sp. energy at section (2) is greater than E_2 the available sp. energy at that section.

Hence the depth at section (2) will be at the critical depth.

$$\text{Hence } y_2 = y_{c2} = 1.256 \text{ m}$$

The U/s depth y_1 will rise to a depth y_1' so that new sp. energy at the U/s section 1 is

$$E_1' = E_{c2} + \Delta Z$$

$$E_1' = y_1' + \frac{V_1'^2}{2g} = E_{c2} + \Delta Z$$

$$\Rightarrow y_1' + \frac{Q^2}{2gy_1'^2} = 1.256 + 0.5$$

$$\Rightarrow y_1' + \frac{(2.4)^2}{2 \times 9.81 \times y_1'^2} = 1.756$$

$$\Rightarrow \boxed{y_1' = 1.648} \Rightarrow \boxed{y_1' > y_2}$$

channel with hump: ↓

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Prob 1 (a) A Rect. channel has a width of 2.0 m and carries a discharge of $4.80 \text{ m}^3/\text{sec}$ with a depth of 1.6 m. At a certain section a small, smooth hump is proposed to be built. Calculate the likely change in the water surface. Neglect the Energy Loss.

Step 1

Determine the U/s flow conditions i.e. whether flow is subcritical or supercritical.

$$q = \frac{4.80}{2} = \frac{4.80}{2.0} = 2.40 \text{ m}^2/\text{sec/m}$$

$$v_1 = \frac{2.40}{1.6} = 1.50 \text{ m/sec}$$

$$F_1 = \frac{v_1}{\sqrt{g y_1}} = \frac{1.5}{\sqrt{9.81 \times 1.6}} = 0.378 < 1$$

[$y_1 = y_2$ since channel is rectangular]

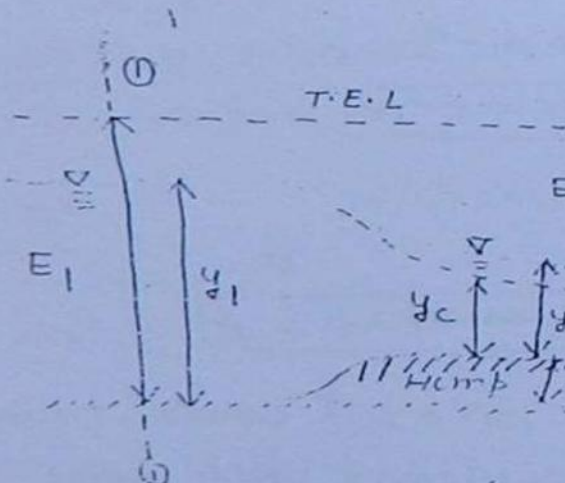
U/s flow is subcritical so the hump will cause a drop in the water surface elevation.

Step 2

$$\begin{aligned} E_1 &= y_1 + \frac{v_1^2}{2g} \\ &= 1.6 + \frac{(1.5)^2}{2 \times 9.81} \\ &= 1.715 \text{ m} \end{aligned}$$

Similarly at section (2)

$$\begin{aligned} E_2 &= E_1 - \Delta z \\ &= 1.715 - 0.10 \\ &= 1.615 \text{ m} \end{aligned}$$



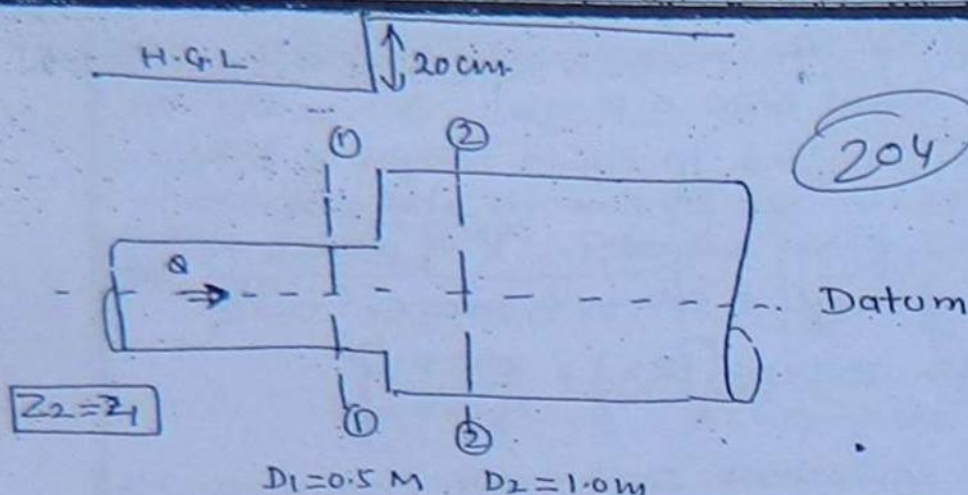
Step 3

check if the flow condition at section (2) is critical.

$$y_c = \left[\frac{q^3}{g} \right]^{1/3} = \left[\frac{(2.4)^3}{9.81} \right]^{1/3} = 0.837 \text{ m}$$

$$E_c = 1.54$$

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$$\text{H.G.L.} \Rightarrow \left(\frac{p_2}{\rho g} + Z_2 \right) - \left(\frac{p_1}{\rho g} + Z_1 \right) = 0.2$$

$$\therefore \frac{p_2}{\rho g} - \frac{p_1}{\rho g} = 0.2 \text{ m} \quad \text{--- (1)}$$

Apply B. Eqn b/w ①-① & ②-②

$$Z_1 + \frac{p_1}{\rho g} + \frac{V_1^2}{2g} = Z_2 + \frac{p_2}{\rho g} + \frac{V_2^2}{2g} + \frac{(V_1 - V_2)^2}{2g}$$

$$\therefore \frac{p_2}{\rho g} - \frac{p_1}{\rho g} = \frac{V_1^2}{2g} - \frac{(V_1 - V_2)^2}{2g} - \frac{V_2^2}{2g}$$

$$0.2 = \frac{V_1^2}{2g} - \frac{V_1^2}{2g} - \frac{V_2^2}{2g} + \frac{2V_1V_2}{2g} - \frac{V_2^2}{2g}$$

$$\Rightarrow 0.2 = \frac{2V_1V_2 - V_2^2}{2g} \Rightarrow 2V_1V_2 - V_2^2 = 0.4 \times 9.81$$

$$\Rightarrow V_1V_2 - V_2^2 = 0.2 \times 9.81$$

$$V_1 = Q/A_1 = \frac{Q}{\pi/4 D_1^2}$$

$$V_2 = Q/A_2 = \frac{Q}{\pi/4 D_2^2} \quad \downarrow \quad [Q = 0.635 \text{ m}^3/\text{sec}]$$

\rightarrow To know direction of flow, know TE_1 & TE_2
 If $TE_1 > TE_2$ then flow from ①-②

Water flows in a 80 mm pipe at Reynolds No. 80,000. The pipe is estimated to have an equivalent sand grain roughness of size 0.16 mm. Determine the head loss expected in 500 m length of pipe if pipe were to act as smooth pipe, how much head loss may be expected. ν of water = 10^{-6}

The following explicit eqn or f may be used

$$\frac{1}{\sqrt{f}} = 1.14 - 2 \log_{10} \left[\left(\frac{k}{d} \right) + \frac{21.25}{Re^{0.9}} \right]$$

Where k = equivalent sand grain roughness =

d = dia. of pipe = 80 mm

Re = Reynolds No. = 80,000

L = 500 m

203

$$Re = \frac{Vd}{\nu} \Rightarrow V = \frac{80,000 \times 10^{-6}}{0.08} = 1 \text{ m/sec.}$$

For Rough pipe \downarrow

$$\frac{1}{\sqrt{f}} = 1.14 - 2 \log_{10} \left[\left(\frac{0.16}{80} \right) + \frac{21.25}{(80,000)^{0.9}} \right]$$

$$\Rightarrow f = 0.0257$$

$$\therefore \text{Head Loss} = \frac{fLV^2}{2gD} = 8.15 \text{ m}$$

For smooth pipe \downarrow

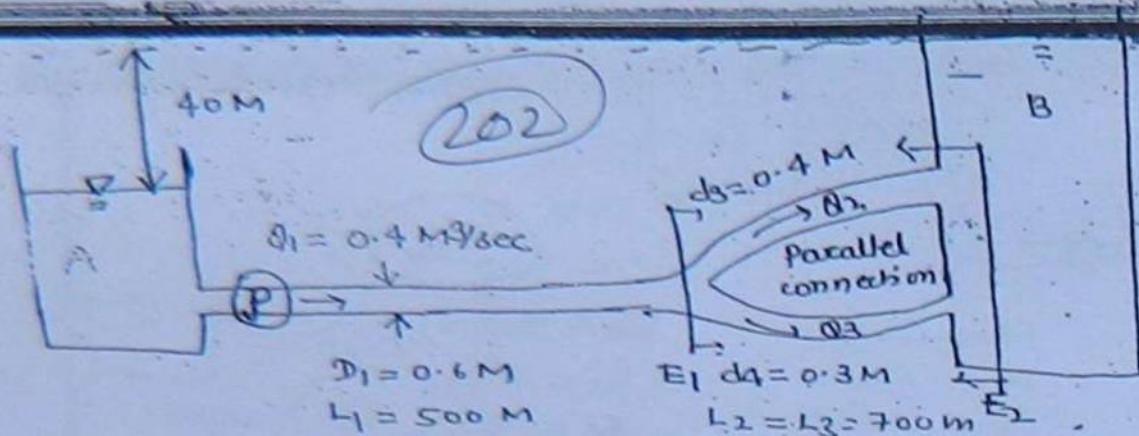
$$k=0$$

$$\frac{1}{\sqrt{f}} = 1.14 - 2 \log_{10} \left[\frac{21.25}{(80,000)^{0.9}} \right]$$

$$\therefore f = 0.0187$$

$$\therefore \text{Head Loss} = \frac{8.15}{0.0257} \times 0.0187 = 5.96 \text{ m.}$$

A horizontal pipe having diameter of 0.5 m expands at a junction to one meter pipe having straight length of hydraulic gradeline at junction rises by 20 cm. Find the flow rate in the pipe.



$$h_{f2} = h_{f3}$$

$$\therefore \frac{f L Q_2^2}{12.1 D_2^5} = \frac{f L Q_3^2}{12.1 D_3^5} \Rightarrow \frac{Q_2^2}{Q_3^2} = \left(\frac{D_2}{D_3}\right)^5 = \left(\frac{4}{3}\right)^5$$

$$\left. \begin{aligned} Q_2 &= 0.268 \\ Q_3 &= 0.13 \end{aligned} \right\}$$

$$\therefore Q_2/Q_3 = \sqrt[5]{4 \cdot 21} \rightarrow$$

$$Q_2 + Q_3 = 0.4 \rightarrow \text{ii}$$

Total head added by the pump
 $= 40 + h_L$

$$[P = 309 \text{ kW}]$$

NOTE: Head added by the pump is $\frac{h_s}{(40)} + \frac{h_{f1} + (h_{f2} + h_{f3})}{(9.97)}$
 However $h_{f2} = h_{f3}$

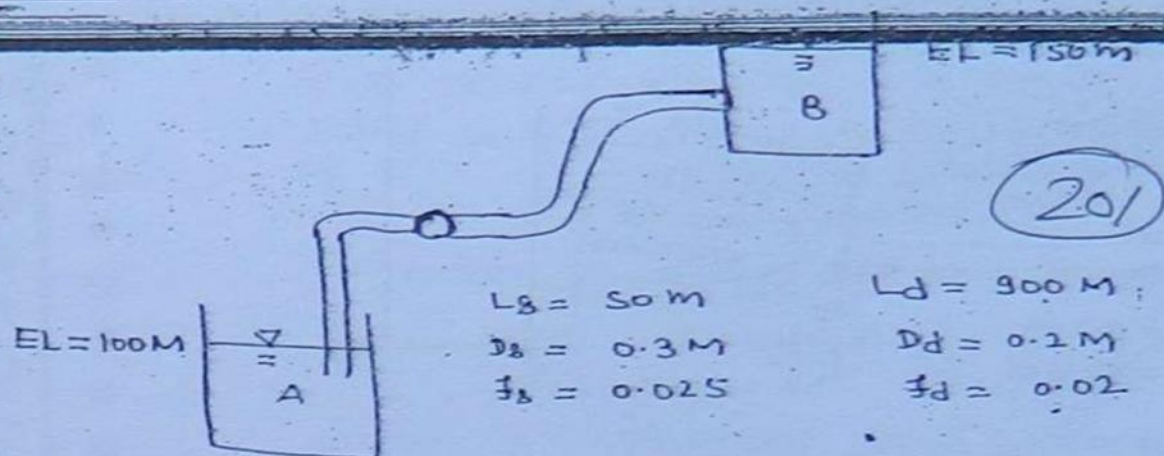
Problem
 30999

A town of 2 Lakh population is to be supplied water from a source, 2500 M away. The lowest water level in the source is 15 M below the water works of the town. The demand of the water is estimated as 150 Lit/capita/day. A pump of 300 H.P. is operated for 15 H. 8% of the max^m demand is 150% of avg. d and velocity of flow through pipe is 1.3 m/s and efficiency of the pump is 70%. Determine the H.G. and friction factor. [Determine

$$\left(\frac{P}{\omega + Z} \right)$$

the total water required]

$$\eta = \frac{\omega Q \text{ Head added by the pump}}{\text{Efficiency of pump}}$$



Head added by the pump.

$$H_p = \text{static head} + h_{fs} + h_{fd}$$

$$= (150 - 100) + \frac{f_s \cdot L_s \cdot Q^2}{12.1 D_s^5} + \frac{f_d \cdot L_d \cdot Q^2}{12.1 D_d^5}$$

$$\therefore h_{fs} = \frac{0.025 \times 50 \times Q^2}{12.1 \times (0.3)^5} = 42.5 Q^2$$

$$h_{fd} = \frac{0.02 \times 900 \times Q^2}{12.1 \times (0.2)^5} = 4648.76 Q^2$$

$$\therefore H_p = 50 + 42.5 Q^2 + 4648.76 Q^2$$

$$= 50 + 4691.26 Q^2 = 80 - 7000 Q^2$$

$$\therefore Q = 0.05 \text{ m}^3/\text{sec}$$

$$\therefore H_p = 80 - 7000 \times (0.05)^2$$

$$= 62.5 \text{ m}$$

$$\therefore \text{Power} = \omega Q H_p$$

Two reservoirs A & B are connected by through system, consisting of ^{conc} 60 cm pipe having ^{500m} 500m length which branches thereafter into two pipes of 40cm dia. & 30 cm dia. each 700 M Long. Pump situated near reservoir A discharges 0.4 m³/sec through the pipe system. The difference in the reservoir level is such that reservoir ^{level} B is 40 M above than Level A. Assuming $f = 0.02$ determine power required for the pump. Assume efficiency = 80%

15. University Question papers of previous year

USN

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10CV45

Fourth Semester B.E. Degree Examination, Dec.2014/Jan.2015 Hydraulics and Hydraulic Machines

Time: 3 hrs.

Max. Marks:100

- Note: 1. Answer any FIVE full questions, selecting atleast TWO questions from each part.
2. Missing data may suitably assumed.

PART - A

1. a. The efficiency η of a fan depends on density ρ , dynamic viscosity μ of the fluid, angular velocity ω , diameter D of the rotor and discharge Q . Express η as

$$\eta = f\left[\frac{Q}{\omega D^3}, \frac{\mu}{\rho \omega D^2}\right]$$

(08 Marks)
- b. Derive different scale ratio's as per Reynold's model law. (06 Marks)
- c. A flow meter tested in the laboratory, gave a pressure drop of 200 kN/m^2 for a discharge of $0.2 \text{ m}^3/\text{s}$ in 200 mm diameter pipe. If a geometrically similar model is tested in 1000 mm diameter pipe at identical conditions of fluid, determine the corresponding discharge and pressure drop in the model. (06 Marks)
2. a. Distinguish between: Pipe flow and open channel flow. (06 Marks)
- b. Derive the Chezy's equation for uniform flow in open channel with usual notations. (07 Marks)
- c. A canal is to have a trapezoidal section with one side vertical and the other sloping at 60° to the horizontal. It has to carry water at $30 \text{ m}^3/\text{s}$ with mean velocity 2 m/s . Compute the dimensions of the section which will require minimum lining. (07 Marks)
3. a. Define specific energy. Explain specific energy curve (sketch). (06 Marks)
- b. A horizontal rectangular channel 4 m wide carries a discharge of $16 \text{ m}^3/\text{s}$. If the initial depth of flow is 0.5 m , determine is there a possibility of formation of hydraulic jump? If the jump forms, determine the sequent depth, Froude number after jump and energy loss. (06 Marks)
- c. Give the classification of surface profiles in case of GVF. (08 Marks)
4. a. Show that the efficiency of a jet striking a series of flat vanes mounted on the periphery of a circular wheel is maximum when the jet velocity is double of vane velocity and maximum efficiency is 50% . (10 Marks)
- b. A jet of 30 mm radius strikes normally on a fixed plate, with a velocity of 35 m/s . Calculate the force exerted by the jet on the plate. If the plate is moving with 15 m/s in the direction of the plate, find the efficiency of the jet. (10 Marks)

Important Note : 1. On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages.
2. Any revealing of identification, appeal to evaluator and/or equations written eg. 42+8 = 50, will be treated as malpractice.

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PART-B

- 5 a. Derive an equation of force exerted by a jet on an unsymmetrical curved vane tangentially, when vane K moving in the x-direction. Draw the velocity triangles and explain. Also find the workdone and efficiency. (10 Marks)
- b. A jet of water with velocity 40m/s strikes a curved vane, which is moving with velocity 20m/s. The jet makes an angle of 30° with the direction of motion of vane at inlet and leaves at an angle of 90° to the direction of motion of vane at outlet. Draw the velocity triangles at inlet and outlet and determine the vane angles at inlet and outlet so that the water enters and leaves the vane without shock. (10 Marks)
- 6 a. Give the classification of turbine with examples. (10 Marks)
- b. Design a Pelton wheel turbine required to develop 1471.5 kW power under a head of 160m at 420 rpm. The overall efficiency may be taken as 85%. Assume $c_v = 0.98$, $c_u = 0.46$, jet ratio = 12. (10 Marks)
- 7 a. Define draft tube. What are its functions? (06 Marks)
- b. What is cavitation? How to eliminate it? (06 Marks)
- c. A Kaplan turbine runner is to be designed to develop brake power of 7350kW, under a head of 5.5m. Diameter of boss is $1/3^{\text{rd}}$ of diameter of runner. Assuming speed ratio = 2.09, flow ratio = 0.68, calculate: i) diameter of runner and boss; ii) speed of runner. Take $\eta_o = 85\%$. (08 Marks)
- 8 a. Define:
- Manometric efficiency
 - Mechanical efficiency
 - Overall efficiency. (06 Marks)
- b. Derive an expression for minimum starting speed of a centrifugal pump. (06 Marks)
- c. The internal and external diameters of the impeller of a centrifugal pump are respectively 200mm and 40mm. The pump is running at 1200rpm. The vane angles of the impeller at inlet and outlet are 20° and 30° . Water enters radially and velocity of flow is constant. Determine the workdone by the impeller per unit weight of water. (08 Marks)

16. Question Bank

Unit-1

1. What is open channel flow?.
2. Differentiate between open channel and pipe flow.
3. Derive various economic sections.
4. Dynamic equation for G.V.F, mild, critical, steep slope.
5. Derive critical depth for various sections?

Unit-2

1. Derive rayleighs, buckinghams pi theorem methods?
2. Derive dimensionless numbers for various formulaes.
3. Model and prototype relations

Unit-3

1. Derive the formulaes for impact on jets for inclined, straight etc.
2. Forces on inclined moving plate, flat stationary plate.
3. Draw the velocity diagrams for jet striking centrally curved symmetrical plate.
4. Work done and Efficiency of flow over radial plate.
5. What is a surge tank? What is the purpose of it? Describe various types of surge tanks
6. Define i) Firm power and secondary power ii) Load factor, Utilization factor and Capacity factor.
7. Describe how hydro power plants are classified into different types based on various criteria.
8. What is intake? Explain different types of intakes with neat sketches
9. Enumerate different elements of hydro electric power station and draw its layout.

Unit-4

1. Describe briefly about the classification of water turbines.
2. Explain the terms unit speed, unit discharge and unit power of a turbine and explain their importance.
3. A pelton wheel operates with a free jet of 150mm diameter under the head of 500m. Its mean runner diameter is 2.25m and it rotates with a speed of 375 rpm. The angle of bucket tip at outlet is 15° coefficient of velocity is 0.98, mechanical losses equal to 3% of power supplied and the reduction in relative velocity of water while passing through bucket is 15%. Find
 - i) The force of jet on the bucket
 - ii) the power developed
 - iii) bucket efficiency and
 - iv) the overall efficiency.

4. Draw a neat sketch of Pelton wheel installation and briefly explain the functions of each component.
5. Describe the points of distinction between impulse turbine and reaction turbines.

Unit-5

1. Obtain the expression for the specific speed of a turbine
2. What do you mean by characteristic curves of a turbine? Discuss about different operating characteristics of a turbine with neat sketches.
3. What is cavitations'? How to detect the cavitation? Explain how to avoid the cavitation.
4. A Kaplan turbine operates under a head of 15.2m, has a speed of 75rpm and develops 50MW of power. The overall efficiency of the turbine is 82%. Calculate the specific speed, unit speed, unit discharge and unit power.
5. Explain the governing of turbines with a neat sketch.
6. The impeller of a centrifugal pump having external diameter and internal diameter 500mm and 250mm respectively, width at outlet 50mm and running at 1200 rpm works against a head of 48m. The velocity of flow through the impeller is constant and is equal to 3m/s. The vanes are back at an angle of 40° at outlet. Determine inlet vane angle, work done by the impeller on water per second and manometric efficiency.
7. Derive an expression for minimum outside diameter of an impeller to enable the pump to start at its normal speed.
8. Explain what are the different efficiencies of a centrifugal pump.
9. Describe multistage pump with (i) impellers in series and (ii) impellers in parallel with the aid of neat sketches.
10. Explain with neat sketches the volute and the diffuser pumps. What is the role of volute chamber of a centrifugal pump

17. Assignment topics

Unit-1: Problems on chezy's Mannings and bazin formulae. Most economical sections. Concept of critical depth, sub critical and super critical

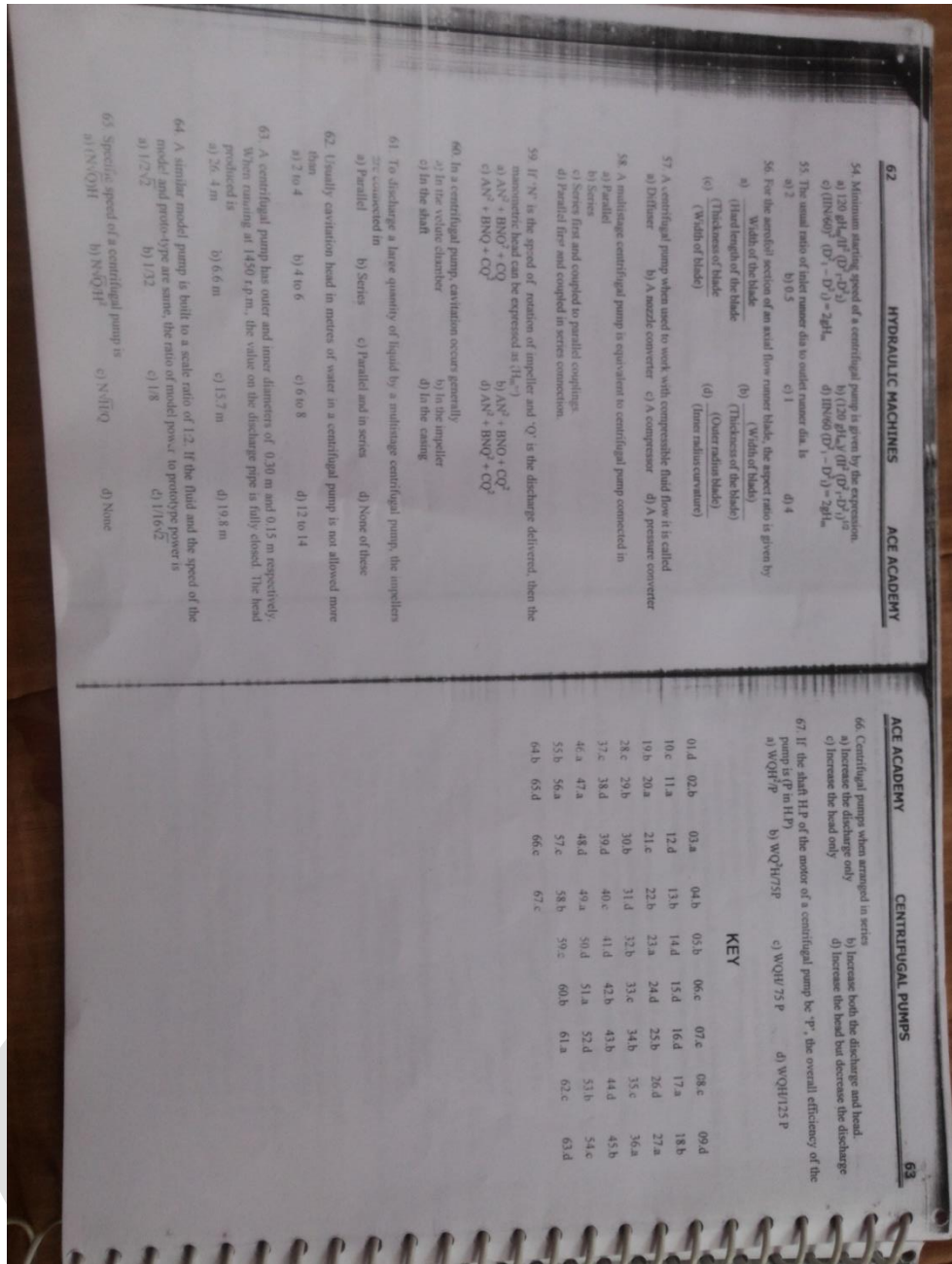
Unit-2: Derivation of Rayleighs , buckinghams pi theorm, problems on dimensional similitude.

Unit-3: Derivation of Impact on jets and its problems. Layout of Hydro Power plant.

Unit-4: Classification of turbines, differentiate between francis, pelton, Kaplan turbine, specific speed concept, cavitation and use of surge tank.

Unit-5: classification of pumps, efficiencies, specific speed of pump, NPSH cavitaion, load factor, capacity factor

18. Unit wise Quiz Questions



32. In a centrifugal pump the inlet angle will be designed to have
 a) Relative velocity vector in the radial direction
 b) Absolute velocity vector in the radial direction
 c) Velocity of flow to be zero
 d) Peripheral velocity to be zero.
33. A fast centrifugal impeller will have
 a) Forward facing blades
 b) Radial blades
 c) Backward facing blades
 d) Propeller type blades
34. In a centrifugal pump, the work done by the impeller is
 a) The manometric head
 b) Greater than manometric head
 c) Less than manometric head
 d) The manometric head loss or losses.
35. Which of the following defines the manometric head of a centrifugal pump
 a) Total height to which the liquid is lifted
 b) Static head min V_f losses in the pump
 c) Difference of energy heads between the outlet of pump casing and the inlet of pump casing
 d) None of the above
36. The number of blades in a centrifugal pump impeller are usually
 a) 8 to 16
 b) 16 to 24
 c) 4
 d) 6
37. Which of the following velocity components is zero at the inlet of a centrifugal pump impeller
 a) Velocity of flow
 b) Absolute velocity
 c) Whirl velocity
 d) Relative velocity
38. The Hydraulic machines subjected to cavitation are
 a) Impulse turbines
 b) Reaction turbines and impulse turbines
 c) Centrifugal pumps and impulse turbines
 d) Centrifugal pumps and reaction turbines.
39. Cavitation is caused by
 a) Low barometric pressure
 b) High pressure
 c) Low velocity
 d) Low pressure
40. In centrifugal pumps, for the vapour pressure consideration, select the incorrect statement.
 a) The diameter of suction side is kept usually more than the discharge side
 b) The suction head is limited to 6 to 8 metres.
 c) The minimum pressure generally occurs along the concave side of vanes near the low pressure side of the impeller
 d) Onset of cavitation is due to higher rotational speeds, restricted suction and higher temperature of liquid.

41. A centrifugal pump when used to work with compressible fluid is called
 a) A fan
 b) A blower
 c) A compressor
 d) All of the above.
42. The partial vacuum created near the impeller centre of a centrifugal pump is called
 a) Diffusion zone
 b) Eye of the impeller
 c) Vortex chamber
 d) Whirl pool chamber
43. In a centrifugal pump impeller, straight radial blades are used to handle
 a) Non-Viscous liquids
 b) Dusty liquids
 c) Oils
 d) Water
44. Manometric head of a centrifugal pump is given by
 a) $h_s + h_d + h_{fr} + h_{gr} + V_f^2/2g$
 b) Work done per kg of water losses with in the impeller
 c) $V_{w2} U_2/g$ for no losses
 d) All of the above.
45. Manometric efficiency of a centrifugal pump is given by
 a) $2gH_m/V_{w2} U_2$
 b) $gH_m/V_{w2} U_2$
 c) $V_{w2} U_2/gH_m$
 d)
46. Mechanical efficiency of a centrifugal pump is given by
 a) $(V_{w2} U_2/gH_m) / S.H.P$
 b) $S.H.P / (V_{w2} U_2/gH_m)$
 c) $S.H.P / H_m$
 d) $H_m / S.H.P$
47. Horse power required to drive a centrifugal pump shaft is given
 a) $(W \cdot Q \cdot H_m / 75) \eta_s$
 b) $75 \eta_s / W \cdot Q \cdot H_m$
 c) $\eta_s / W \cdot Q \cdot H_m$
 d) $\frac{\eta_s}{W \cdot Q \cdot H_m}$
48. Euler's head for a centrifugal pump is
 a) $V_{w2} U_2/g$
 b) The head imparted to the liquid by the impeller
 c) The workdone per second per kg of liquid
 d) All of the above.
49. If H_s is the head imparted to the liquid by the pump impeller with finite number of blades and H_f be the Euler's head then vane efficiency is given by
 a) H_s / H_f
 b) H_f / H_s
 c) $H_s \cdot H_f$
 d) H_s / H_f
50. For maximum efficiency the lease practicable value for outlet vane angle of a centrifugal pump vane is
 a) 60°
 b) 90°
 c) 45°
 d) 20°
51. For 100% vane efficiency, manometric efficiency of a centrifugal pump is
 a) Equal to the hydraulic efficiency
 b) Less than the hydraulic efficiency
 c) Greater than the hydraulic efficiency
 d) None of the above
52. Loss of head due to shock (vortex discharge) at entrance of the impeller of a centrifugal pump
 a) $(U + V \cot \theta)^2 / g$
 b) $(U + V \cot \theta) / g$
 c) $(U - V \cot \theta)^2 / g$
 d) $(U - V \cot \theta) / g$
53. The angle of divergence for the enlarging section of a vortice casing
 a) Shall not be $> 20^\circ$
 b) Shall not be $> 10^\circ$
 c) Shall be 0°
 d) Shall be 45°

10. The pressure in the suction pipe is
 a) above atmospheric b) atmospheric
 c) Below atmospheric d) none of the above.
11. The impeller that are more efficient and suitable for pumping relatively pure liquids are
 a) Shrouded impellers b) Semi open impellers
 c) Open impellers d) None of the above.
12. The foot valve of a centrifugal pump
 a) Is a non return valve b) Is a lower end of suction pipe
 c) Is lower end of delivery pipe d) Both a and b are correct.
13. The first operation of a centrifugal pump is
 a) Governing b) Priming c) Fuelling d) None of the above.
14. Pickup the correct statement. The centrifugal pump should be installed above the water level in the sump such that
 a) Its height is not allowed to exceed 6.7 m
 b) The negative pressure do not reach as low a value as the vapour pressure
 c) Its height is not more than 10.3 m at ordinary temperature of liquid.
 d) Both a and b are correct.
15. Pickup the correct statement Pertaining to centrifugal pump installation.
 a) The suction pipe has larger dia as compared to the discharge pipe.
 b) The suction pipe is provided with a foot valve and a strainer.
 c) The discharge control valve is fitted in the delivery pipe.
 d) All are correct.
16. The purpose of priming is to
 a) Drive out water from sump
 b) Fill the suction pipe with water
 c) Fill the delivery pipe with water
 d) Drive out air or gas or vapour from suction pipe, casing and part of delivery valve.
17. The pressure at the 'eye' of the impeller is
 a) Partial vacuum b) Atmospheric
 c) Above atmospheric d) none of the above.
18. In a centrifugal pump water enters the impeller
 a) Axially b) Radially c) Tangentially d) None
19. When guide vanes are installed in a centrifugal pump, it is called
 a) Volute casing pump b) Turbine pump
 c) Multi stage pump d) Double inlet pump
20. A "turbine pump" is
 a) A diffuser vane pump b) A single volute pump
 c) A double volute pump d) A variable speed pump
21. "Spiral shaped casing" is provided for
 a) Turbine pump b) Diffuser pump c) Volute pump d) none

22. A multi stage pump consists of
 a) Two or more pumps b) Two or more impellers
 c) Two or more suction pipes d) Two or more delivery pipes.
23. A centrifugal pump is basically a
 a) Radial flow pump b) Axial flow pump
 c) Mixed flow pump d) Tangential flow pump
24. Pickup the correct statements:
 a) Mixed flow pumps are generally used where a large quantity of liquid is to be discharged to low heights.
 b) Axial flow pumps are usually designed to deliver very large quantities of liquid at relatively low heads.
 c) A double suction pump neutralize the axial thrust on the impeller.
 d) All are correct.
25. For deep wells and mines, it is better to provide the shaft
 a) Horizontally b) Vertically c) Inclined d) Curvilinearly
26. Pickup the correct pair of the following:
- | Type of pump | Total Head |
|--------------------|------------|
| a) Low head | 0 to 15 m |
| b) Medium head | 15 to 40 m |
| c) High head | 40 m |
| d) All are correct | |
27. 'Static head' is
 a) $h_3 + h_4$ b) $h_1 + h_4 + h_5$ c) $h_1 + h_4 + h_5 + h_6$ d) $h_3 - h_4$
28. The 'Mano metric head' is the
 a) Difference in elevation between the water surface in the high level reservoir and the water level in the pump
 b) The height to which water is lifted the pump
 c) The difference in piezometric heads between the points on the delivery and suction pipes as close to the pump as possible.
 d) Head developed by the pump.
29. The work done by the impeller of a centrifugal pump on water per second per unit weight of water is given by ('2' represents outlet '1' represents inlet).
 (a) $\frac{V_{w2} U_2}{g}$ (b) $\frac{V_{w1} U_1}{g}$ (c) $\frac{V_{w2} U_2 - V_{w1} U_1}{g}$ (d) None
30. The inlet velocity triangle for a centrifugal pump is generally
 a) An acute angled triangle b) Right angle triangle
 c) Obtuse angled triangle d) Straight line
31. The work done by the impeller of a centrifugal pump is
 a) Independent of inlet radius b) Dependent on inlet radius
 c) Dependent on outlet radius d) both a and C are correct.

$$\therefore \text{NPSH} = P_s/\rho g - V_s^2/2g - h_s - h_{fs} - P_s/\rho g + V_s^2/2g$$

$$\text{NPSH} = \frac{P_s - P_v}{\rho g} - h_s - h_{fs}$$

But R.H.S is the total suction head

$$H_{sv} = \frac{P_s - P_v}{\rho g} - h_s - h_{fs}$$

$\therefore \text{NPSH} = H_{sv}$ i.e. Total suction head

$\therefore \text{NPSH}$ is the head required to make the liquid flow through the suction pipe into the impeller.

20. Cavitation in Centrifugal Pumps:

If the pressure at the suction side of the pump drops below the vapour pressure of the liquid then cavitation may occur.
Cavitation in a centrifugal pump results in sudden drop of head and efficiency.

$$\text{Thomas's cavitation factor} = \frac{(H_{\text{atm}} - H_v) - (h_s + h_{fs})}{H_m}$$

$$= \frac{\text{NPSH}}{H_m} = \frac{H_{sv}}{H_m}$$

H_{atm} = atm. pr. head,

H_v = vapour pr. head,

h_s = suction head

h_{fs} = head loss in suction,

H_m = manometric head,

H_{sv} = total suction head.

N_s = sp. speed.

Critical $\tau = 0.103 (N_s/1000)^{0.5}$

"When hot liquids are to be pumped the pumps have to be installed at liquid surface or even below the liquid surface"

In first case $h_s = 0$, in second case $h_s \leq 0$ indicating that there is +ve pr. at pump inlet.

Suction specific speed:

$$S = N/\sqrt{Q} = (N/\sqrt{g})^{0.5}$$

$$H_m^{0.75}/4$$

Range of 'S' for cavitation free operation of C.P. and propeller pumps.

$$S = 4700 \text{ to } 6700$$

$$S = \frac{1.4 \rho m (P_s)^{1/2}}{m^{1/2}}$$

Cavitation due to min. pr. occurs along the convex side of vane near the impeller.

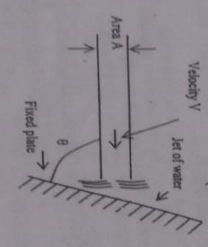
CENTRIFUGAL PUMPS OBJECTIVES

1. A centrifugal pump is a
 - a) Turbo machine
 - b) Rotodynamic machine
 - c) Positive displacement pump
 - d) both a and b
2. In case of centrifugal pumps the energy imparted to the liquid (increases of pressure) is due to
 - a) Centrifugal action alone
 - b) Change of angular momentum also
 - c) Positive displacement
 - d) None of the above.
3. The action of centrifugal pump is
 - a) Reverse of radially in ward flow reaction turbine
 - b) Reverse of propeller turbine
 - c) Reverse of mixed flow turbine
 - d) None
4. A centrifugal pump is a hydraulic machine which converts.
 - a) Hydraulic energy to mechanical
 - b) Mechanical energy to pressure energy
 - c) Mechanical energy to electrical energy
 - d) none
5. A centrifugal pump converts
 - a) Electrical energy to mechanical energy
 - b) Kinetic energy to pressure energy
 - c) Pressure energy to kinetic energy
 - d) mechanical energy to electrical energy.
6. The motion that helps in the working of a centrifugal pump is
 - a) Radial flow
 - b) Free vortex flow
 - c) Forced vortex flow
 - d) Axial flow
7. The pipe which connects the sump and the inlet of the pump is called
 - a) Delivery pipe
 - b) Foot valve
 - c) Suction pipe
 - d) none of the above.
8. Pickup the correct statement of the following:

The hydraulic function of

 - a) An impeller is to convert mechanical energy into hydraulic energy.
 - b) A casing is to transform most of the kinetic energy to into pressure energy.
 - c) Both a and b are correct
 - d) None
9. Pickup the incorrect statement
 - a) The discharging capacity of a centrifugal pump is much greater than that of a reciprocating pump.
 - b) Centrifugal pump can be used for lifting highly viscous liquids.
 - c) Centrifugal pump can be operated at high speeds without any danger of separation and cavitation.
 - d) Maintenance cost of centrifugal pump is very high.

PREVIOUS APPSC QUESTIONS

01. A jet of water of cross-sectional area 1200 mm^2 impinges on a fixed plate in the direction perpendicular to the plate with a velocity of 5 m/sec . Taking unit weight of water as 10 kN/m^3 , find the force exerted on the plate
(A) 3 kN (B) 30 kN (C) 500 N (D) 12 kN (APPSC 2006 LR)
02. A water jet of cross-sectional area 2000 mm^2 hits a symmetrical curved fixed plate with a velocity of 10 m/s . If the deflection angle of the plate is 120° , what is the force exerted on the plate
(A) 100 N (B) 1866 N (C) 1500 N (D) 866 N (APPSC 2006 LR)
03. A water jet impinges on a series of vanes with a velocity of 25 m/s causing the vanes to move with a velocity of 10 m/s . The efficiency of the wheel is _____ (APPSC 2006 LR)
(A) 40% (B) 60% (C) 36% (D) 48%
04. A jet boat moves with a velocity of 36 km/hr when a jet of water issues with a velocity of 30 m/sec through an orifice of cross-sectional area 2000 mm^2 . What is the propulsion force?
(A) 24 kN (B) 60 kN (C) 36 N (D) 48 kN (APPSC 2006 LR)
05. Jet boat moves with a velocity of 10 m/s when the velocity of jet is 30 m/s . What is the efficiency of propulsion?
(A) 33.33% (B) 37.5% (C) 30% (D) 45% (APPSC 2006 LR)
06. The force exerted by the jet on fixed plate as shown in the figure, is equal to (APPSC 2006 LR)
(A) $\frac{AV^2}{g} \sin \theta$
(B) $\frac{\rho AV^2}{g} \sin \theta$
(C) $\frac{\rho AV^2}{g} \cos \theta$
(D) $\frac{AV^2}{g} \cos \theta$
- 
07. The efficiency of a jet of water having a velocity 'V' striking a series of vertical plates moving with a velocity 'U' maximum when: (APPSC 2008)
(A) $U = 2V$ (B) $U = \frac{V}{2}$ (C) $U = \frac{3V}{2}$ (D) $U = \frac{4V}{3}$
08. A water jet of area 0.002 m^2 and velocity 15.0 m/s impinges on a plate moving along the jet at 5.0 m/s . The force exerted by the jet on the plate is (APPSC 2008 LR)
(A) 0.02 kN (B) 450.0 N (C) 100.0 N (D) 0.20 kN

09. A change of angular momentum is a fluid flow along a curved path produces a
(A) Change in momentum
(B) Change in energy
(C) Change in pressure
(D) Torque (APPSC - 2011)
10. For a fluid flowing in a curved path, a change in its angular momentum develops a
(A) torque
(B) velocity
(C) change in pressure
(D) slope (APPSC - 2011)
11. The moment correction factor (β) for a laminar flow is given by
(A) One (B) (1.15) (C) (1.25) (D) (1.35) (APPSC - 2011)
12. The force exerted by a jet of water on a stationary vertical plate in the direction of jet is given by
(A) $F = \rho A V^2 \sin^2 \theta$
(B) $F = \rho A V^2 (1 + \cos \theta)$
(C) $F = \rho$
(D) None of the above (APPSC - 2011)
13. A jet of fluid, of area A and velocity V, impinges on a stationary flat plate which is held perpendicular to the flow direction. Then the normal force exerted by the jet on the plate is
(A) ρAV (B) ρAV^2 (C) $\rho A^2 V$ (D) $\rho A^2 V^2$ (DVS - 2011)
Where ρ = mass of the jet fluid
14. A jet of water impinges on a stationary plate inclined at angle of $\alpha = 30^\circ$ to the direction of the jet. If the normal force action on the jet is 600 N , the component of force parallel to the direction of jet is
(A) 540 N (B) 460 N (C) 300 N (D) 225 N (DVS - 2011)
15. A jet of water strikes a flat vertical plate moving in the direction of jet at one-fourth the velocity of the jet. Then the force exerted by the jet on the plate is given by $K \times \rho A V^2$ where $K =$ _____
(A) One (B) $2/3$ (C) $2/9$ (D) $4/9$ (DVS - 2011)
16. Efficiency in the case of a jet striking plate is defined as the ratio of
(A) work done on the plate to the kinetic energy of the jet
(B) work done on the plate to the potential energy of the jet
(C) kinetic energy to potential energy of the jet
(D) potential energy to kinetic energy of the jet (DVS - 2011)
- KEY:
01. 02. 03. (D) 04. 05. (D) 06. (B) 07. (B) 08. (D) 09. 10. 11. 12. (C) 13. 14. 15. 16.

20. Known gaps ,if any

--NONE--

21. References, Journals, websites and E-links

Text Books

1. Fluid mechanics and Hydraulic machines by Modi & Seth
2. Fluid mechanics and Hydraulic machines by Raj put

Reference Text Books

1. Fluid mechanics and fluid power engineering by D.S. Kumar

2. Fluid mechanics and machinery by D.Rama durgaiiah.
3. Hydraulic machines by Banga & Sharma
4. Instrumentation for engineering Measurements by James W. Dally, William E. Riley,

Journals

1. International Journal of fluid mechanics
2. International Journal of numerical methods in fluids.
3. Annual Review of Fluid Mechanics
4. Journal of Fluid Mechanics
5. Physics of Fluids
6. European Journal of Mechanics B/Fluids
7. Journal of Turbulence

Websites

1. www.ieeefmh.org/

<http://www.efluids.com/>

http://www.yahoo.com/Science/Engineering/Mechanical_Engineering/Fluid_Dynamics/

<http://www.cfd-online.com/>

22. Quality Control Sheets

EVALUATION SCHEME:

PARTICULAR	WEIGHTAGE	MARKS
End Examinations	75%	75
Two Sessionals	20%	20
Assignment	5%	5
TEACHER'S ASSESSMENT(TA)*	WEIGHTAGE	MARKS

*TA will be based on the Assignments given, Unit test Performances and Attendance in the class for a particular student.

23. Student List

II-A Section

S.No	Roll No	Student Name
1	14R11A0102	ATHIREK SINGH JADHAV
2	14R11A0103	BODAPATI ARVIND RAJ
3	14R11A0104	BODHASU MADHU
4	14R11A0105	BOLAGANTI YASHWANTH TEJA
5	14R11A0106	CHADA SHIVASAI REDDY
6	14R11A0107	D SATISH KUMAR
7	14R11A0108	E TEJASRI
8	14R11A0109	G DARSHAN
9	14R11A0110	GALIPELLI SRIKANTH
10	14R11A0111	GATTU MANASA
11	14R11A0112	GEEDI SRINIVAS
12	14R11A0113	GUNTUPALLY MANOJ KUMAR
13	14R11A0114	K ANJALI
14	14R11A0115	KASULA HIMA BINDU
15	14R11A0116	KASTHURI VINAY KUMAR
16	14R11A0117	KOPPULA KEERTHIKA
17	14R11A0118	KRISHNA VAMSHI TIPPARAJU
18	14R11A0119	MADDULA MANORAMA REDDY
19	14R11A0120	MALINENI VENKATA DILIP
20	14R11A0121	MANDA KUMIDINI
21	14R11A0122	MINNIKANTI NAGASAI GANESH BABU
22	14R11A0123	MOHD ABDUL WALI KHAN
23	14R11A0124	MOTUPALLI VENTAKA KIRAN
24	14R11A0125	MUDDETI HARI

25	14R11A0126	MUSHKE VAMSHIDAR REDDY
26	14R11A0127	NAGUNOORI PRANAY KUMAR
27	14R11A0128	NALLA UDHAY KUMAR REDDY
28	14R11A0129	P GAYATHRI
29	14R11A0130	PADALA SRIKANTH
30	14R11A0131	PASUPULATI SWETHA
31	14R11A0132	POLISETTY VINEEL BHARGAV
32	14R11A0133	PUNYAPU VENKATA SHRAVANI
33	14R11A0134	R DIVYA
34	14R11A0136	RAVULA VAMSHI
35	14R11A0138	S BARATH KUMAR
36	14R11A0139	S PRASHANTH REDDY
37	14R11A0140	S SAI RAGHAV
38	14R11A0141	SHAIK SHAMEERA
39	14R11A0142	SREEGAADHI SAICHARAN
40	14R11A0143	SRIRAM SURYA
41	14R11A0144	SUNKARI SHIVA
42	14R11A0145	VANAMALA SURENDER NIKITHA
43	14R11A0146	YADAVALLI PAVAN KUMAR

II-B-section

S. No	Roll No	Student Name
1	14R11A0149	A. SRAVAN KUMAR
2	14R11A0150	B MAHENDRA VARDHAN
3	14R11A0151	B. VIJAY
4	14R11A0152	B. KIRAN KUMAR
5	14R11A0153	B. SUNIL NAIK
6	14R11A0154	D. VENU CHARY
7	14R11A0155	D. VASANTHA KUMAR

8	14R11A0157	G. NIKHIL
9	14R11A0158	G. SANDEEP KUMAR
10	14R11A0159	G. CHARAN KUMAR
11	14R11A0160	J. HARISH KUMAR
12	14R11A0161	K.J. NANDEESHWAR
13	14R11A0162	K. SANTHOSH KUMAR
14	14R11A0163	K BHARATH KUMAR
15	14R11A0164	K ABHILASH
16	14R11A0165	K SAI KRISHNA
17	14R11A0168	MOHD. ABBAS
18	14R11A0169	M SRINIVAS
19	14R11A0170	N SANTHOSH
20	14R11A0172	OSA NITHISH
21	14R11A0173	P INDRA TEJA
22	14R11A0174	P NAVEEN KUMAR
23	14R11A0175	P BHARATH NARSIMHA REDDY
24	14R11A0176	P SURENDER
25	14R11A0177	R VIHARI PRAKASH
26	14R11A0178	S BHANU KISHORE
27	14R11A0179	SHAILESH KUMAR SINGH
28	14R11A0180	SYED OMER ASHRAF
29	14R11A0181	V SAI SHARATH
30	14R11A0182	Y VENKATA MOHAN REDDY

24. Group-Wise students list for discussion topics

II-A Section

S. No	Group No	Roll No	Student Name
1	1	14R11A0102	ATHIREK SINGH JADHAV

2	1	14R11A0103	BODAPATI ARVIND RAJ
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